

NOVEMBER 1982

MDC H0145

**SPACE STATION NEEDS,
ATTRIBUTES AND ARCHITECTURAL OPTIONS**

**Midterm Main Briefing
16 November 1982**

COPY NO 9

MCDONNELL DOUGLAS ASTRONAUTICS COMPANY



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**SPACE STATION NEEDS,
ATTRIBUTES AND ARCHITECTURAL OPTIONS**

**Midterm Main Briefing
16 November 1982**

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MDC H0145

MCDONNELL DOUGLAS ASTRONAUTICS COMPANY-HUNTINGTON BEACH

5301 Bolsa Avenue, Huntington Beach, California 92647 (714) 896-3311

**SPACE STATION
NEEDS, ATTRIBUTES, AND
ARCHITECTURAL OPTIONS STUDY
NASA HEADQUARTERS**

Midterm Review

16 November 1982

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MDAC SPACE STATION MIDTERM BRIEFING

AGENDA

- Summary — Dave Wensley
- Mission Requirements (Task 1)
 - Methodology - Dave Riel
 - User Interaction - Dr. Harry Wolbers
 - Science and Applications Missions - Dr. Harry Wolbers
 - Commercial Missions
 - Mission Candidates - Dr. Harry Wolbers
 - Electrophoresis - Jim Rose - MDAC - St. Louis
 - Selected Missions - Dr. Myron Weinberg - Booz, Allen & Hamilton
 - Technology and Operational Missions
 - National Security Missions (Summary)
 - Missions Requirements Summary

A2

Dave Riel

AGENDA (CONT)

- **Programmatics (Task 3) — Bob Cowls**

- Funding Model
- Element Costs
- Program Costs

- **Mission Implementation (Task 2) — Bill Nelson**

- Methodology
- Architectural Options
- Strawman Program

- **National Security Missions (DoD Task 4) — Dave Riel** **(Classified Session)**

- **Discussion**

■ Summary — Dave Wensley

- MDAC Team Organization
- Study Approach
- Progress Versus Plan
- Results to Date
- Midterm Conclusions (Trends)

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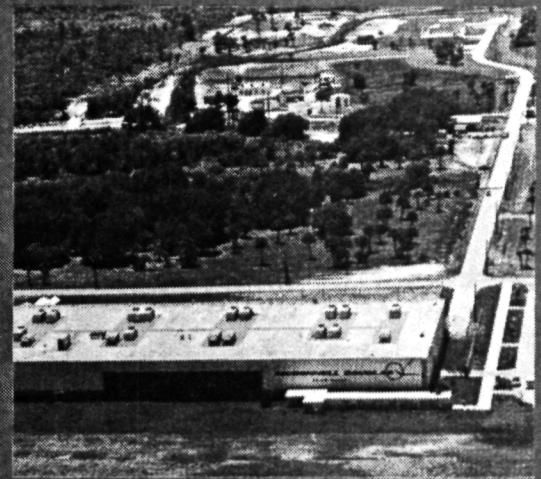
MCDONNELL DOUGLAS ASTRONAUTICS COMPANY
President – John F. Yardley

**MCDONNELL DOUGLAS
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General Manager
R. K. Jacobson
■ Florida ■ Huntsville ■ Houston

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Vice President
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E. F. Branahl

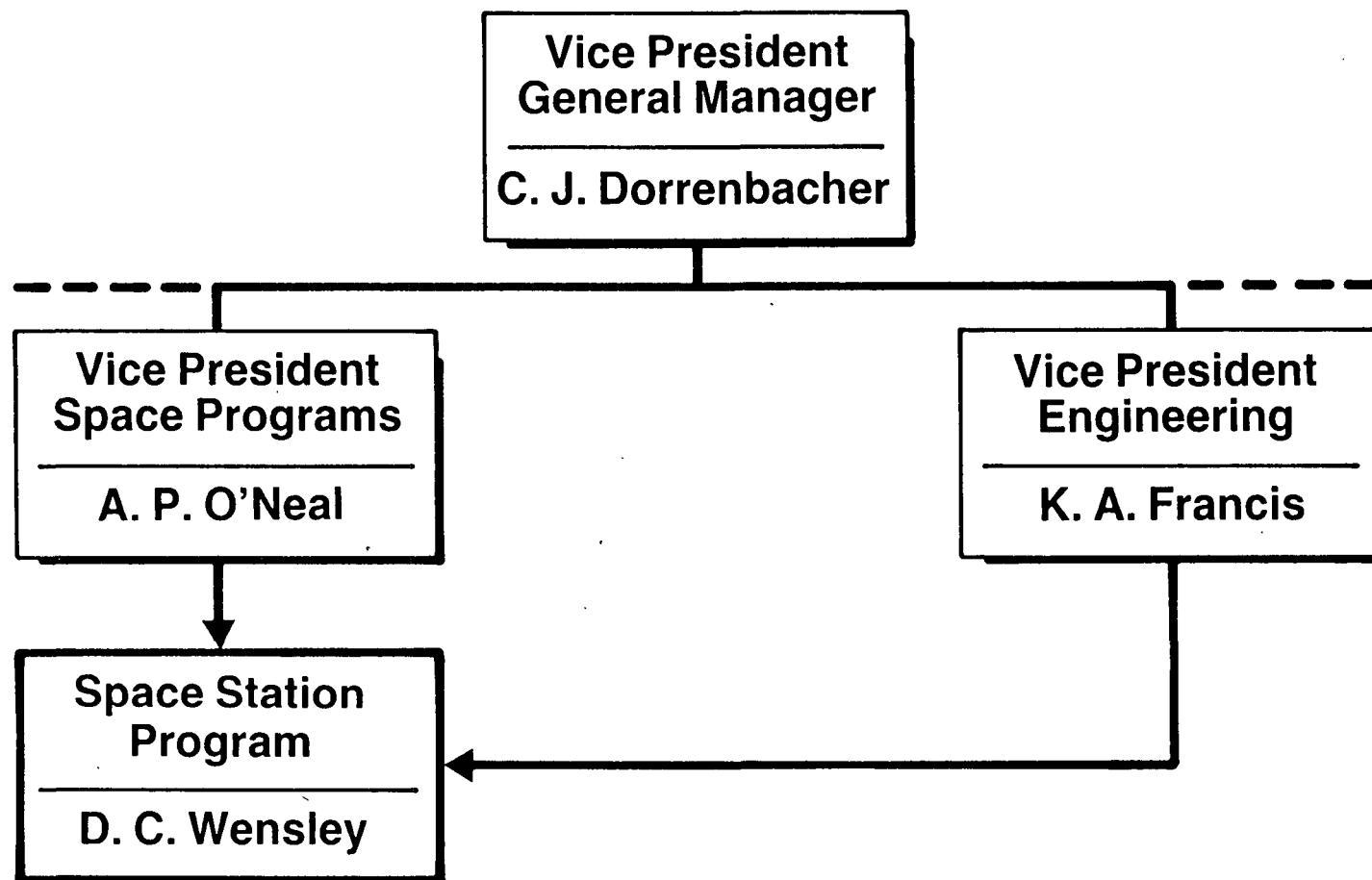
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Vice President
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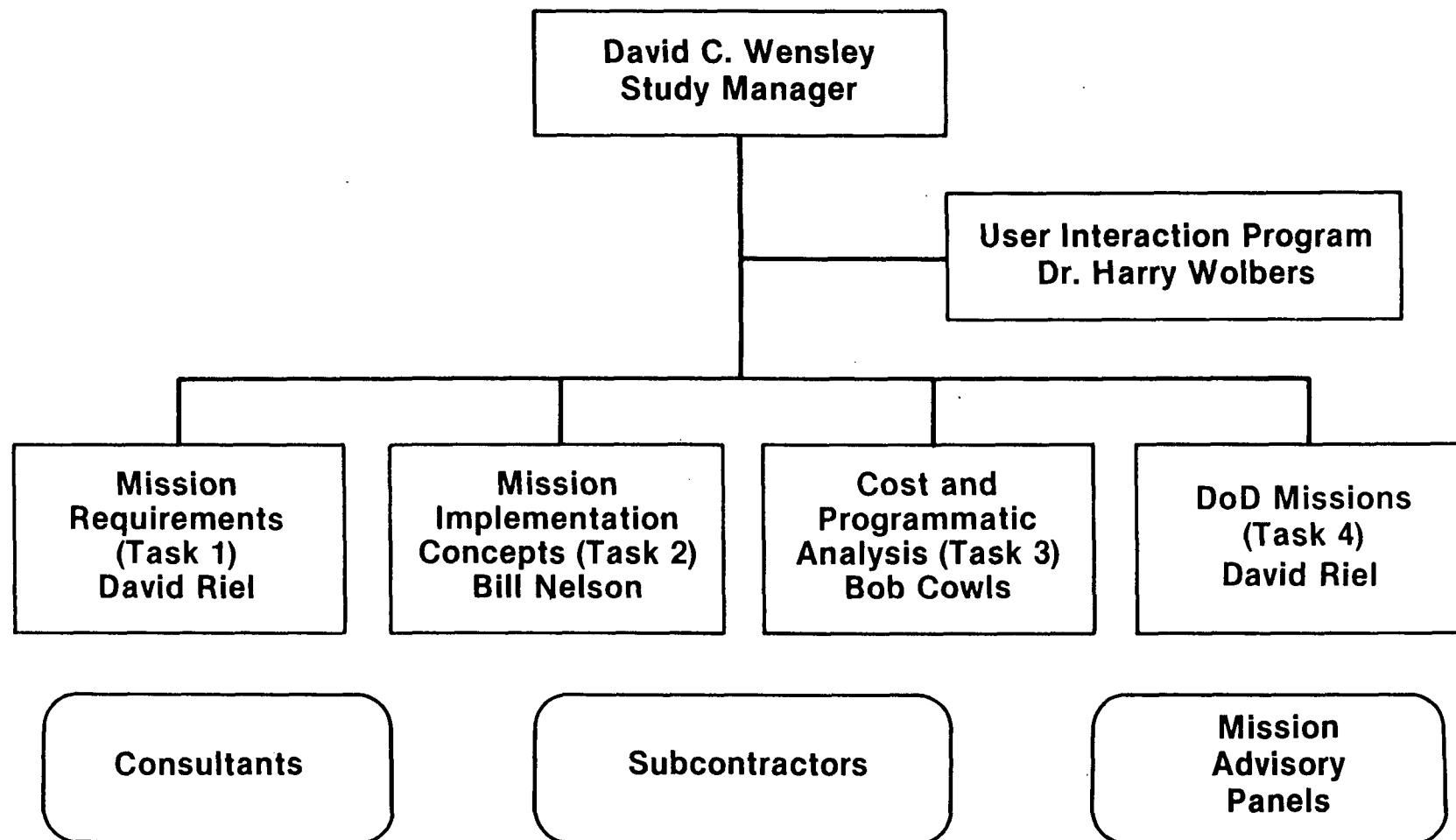
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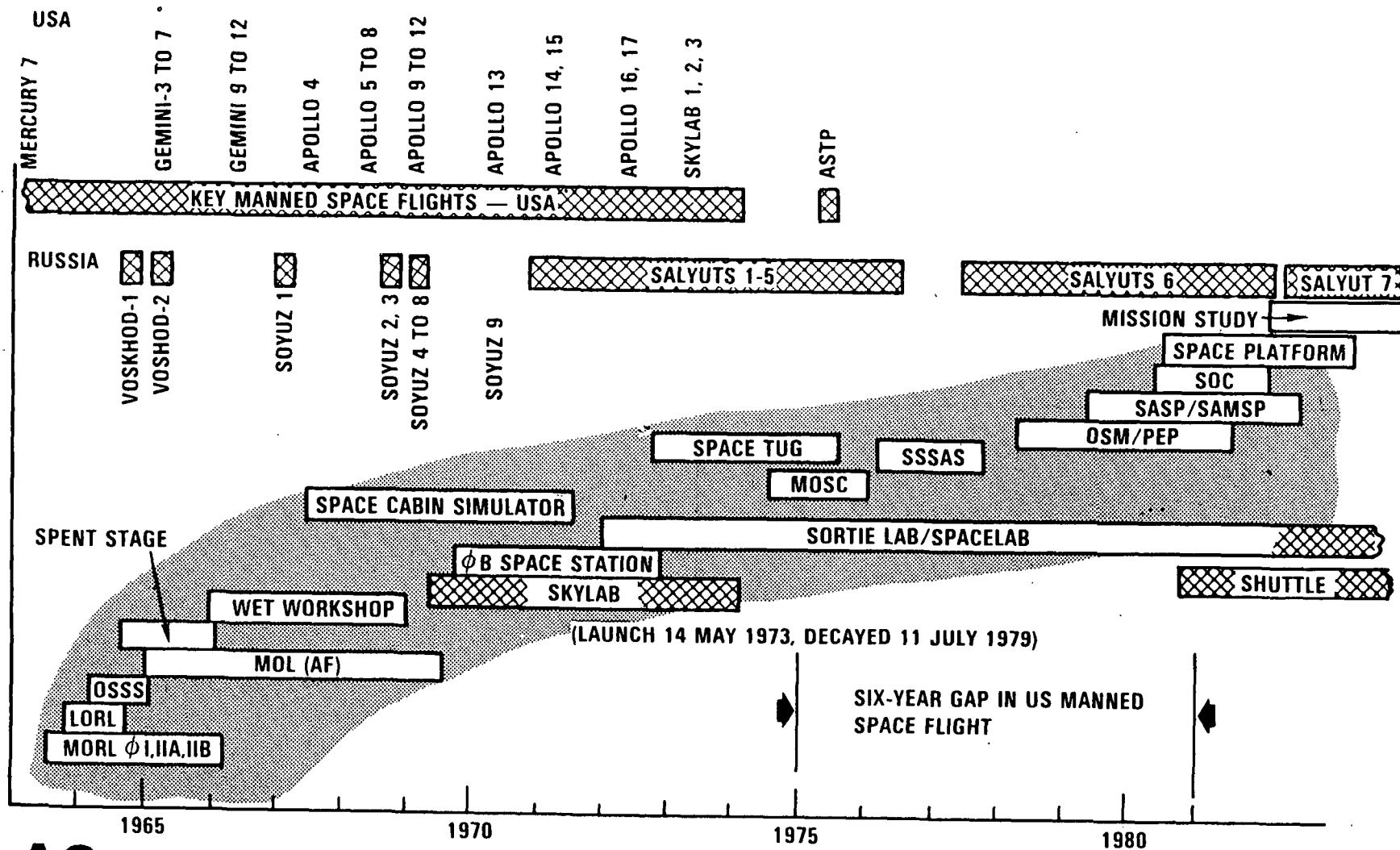
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MDAC-HB STUDY ORGANIZATION (U)



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SPACE STATION HERITAGE



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MDAC SUPPORT TEAMS

Consultants

- Booz, Allen & Hamilton
 - Commercial Missions
 - Benefits Analysis
- MDAC — St. Louis
 - Commercial Missions
- Stanford Research Institute
 - National Defense Missions
- Dr. John Logsdon
 - Program Planning

Mission Advisory Panels

- Science and Applications
- Commercial Missions
- National Security Missions

Subcontractors

- Ford Aerospace
 - Communications Missions
 - Ground Data System
- Hamilton Standard
 - Environmental Control and Life Support Systems
- Bendix
 - Navigation and Control
- Vought/LTV
 - Teleoperators

- Operational Missions
- Technology Missions

MDAC TEAM IS PROMOTING SPACE STATION

- 6 TV Interviews
- 5 Press Conferences and News Releases
- 2 Papers and Publications
- 12 Presentations and Briefings
- 8 Meetings With Private Sector
- 15 Meetings With Government Sector — Political
- 11 Meetings With Foreign Sector

Total: 59 Events to Midterm

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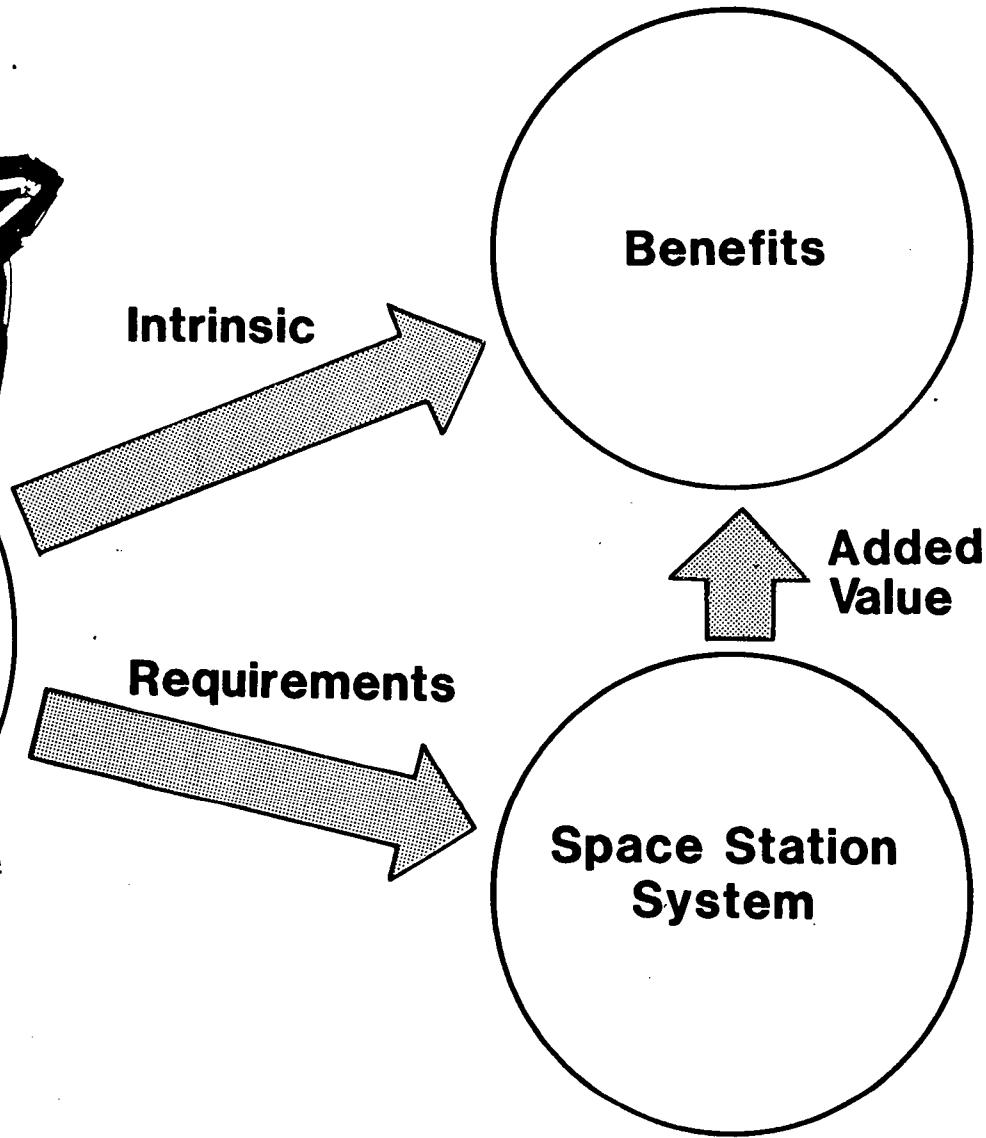
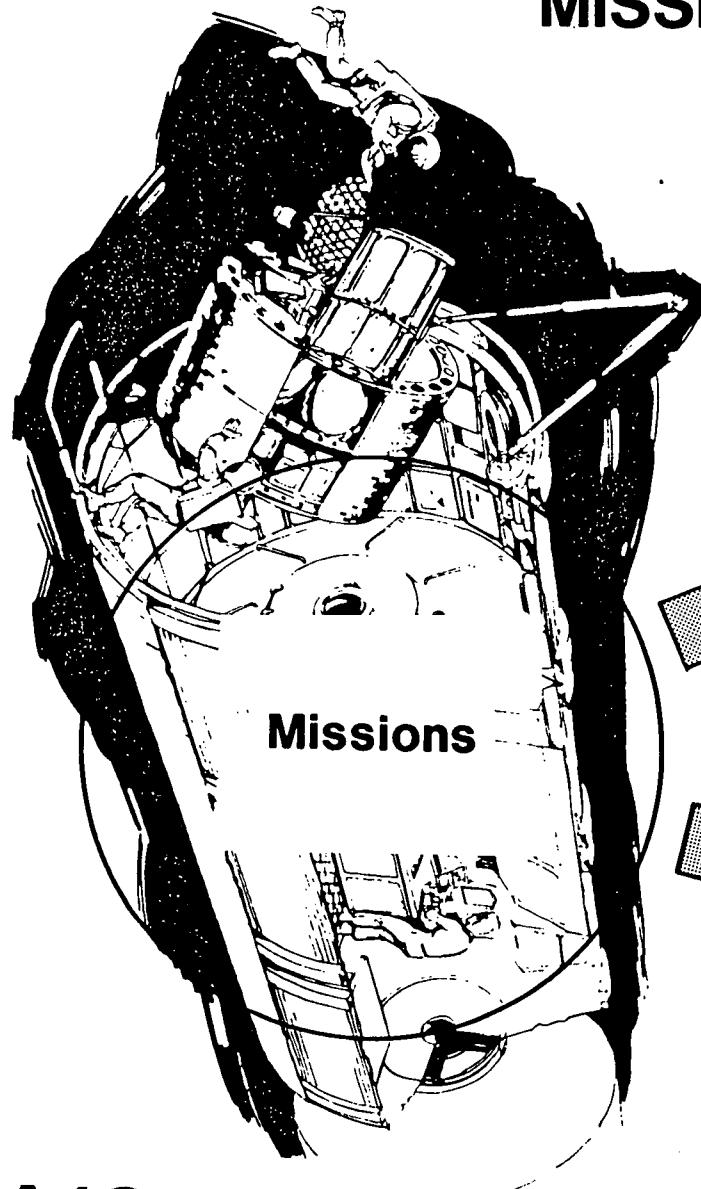
STUDY OBJECTIVES

DEFINE

- The Missions
- Requirements They Impose:
 - For Manned Space Station
 - For Supporting Orbital Facilities
 - For Transportation
- Architectural Solutions:
 - To Implement Above Requirements
- Program Concepts:
 - Content
 - Costs
 - Schedules

A11

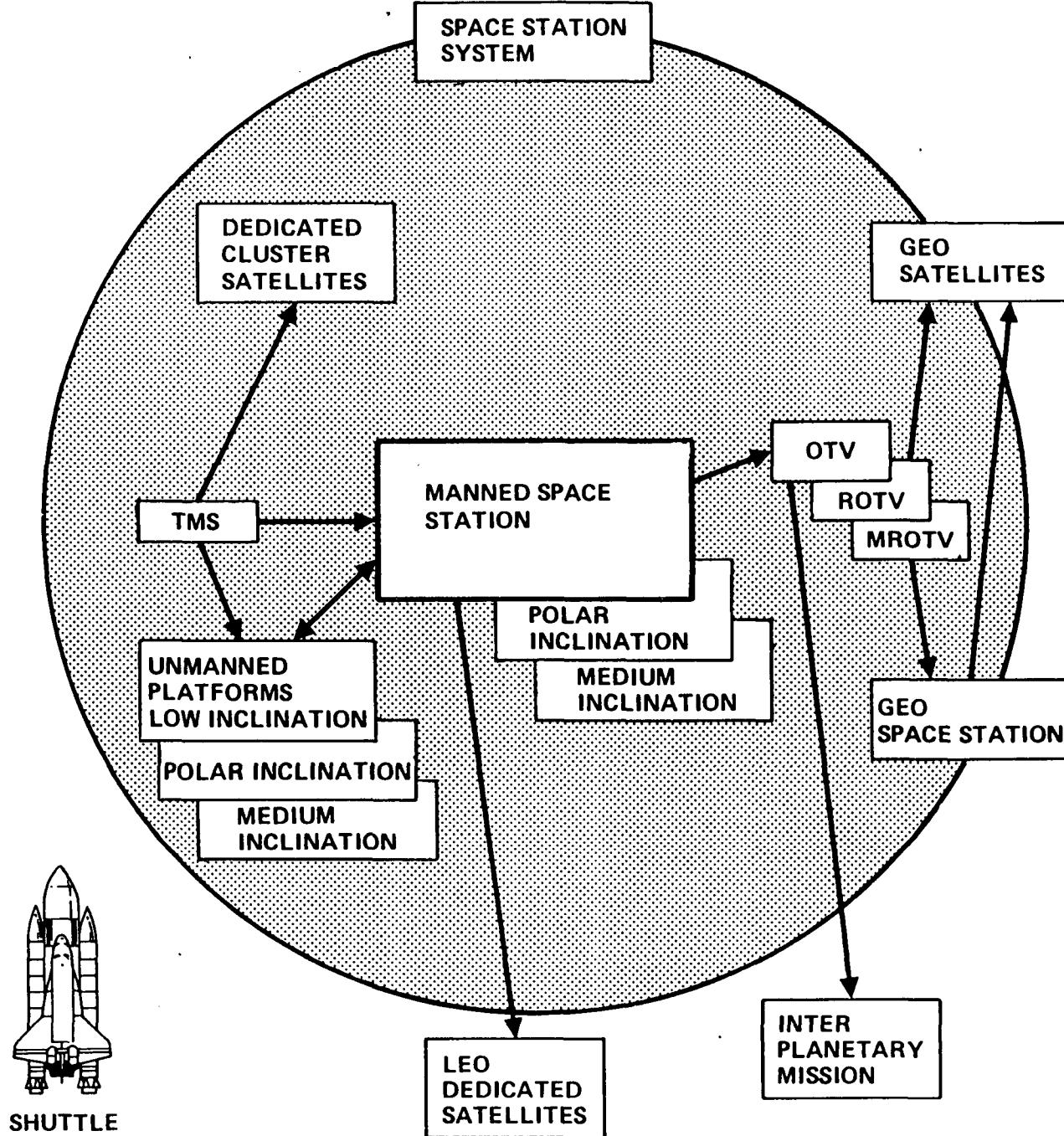
MISSION ANALYSIS



A12

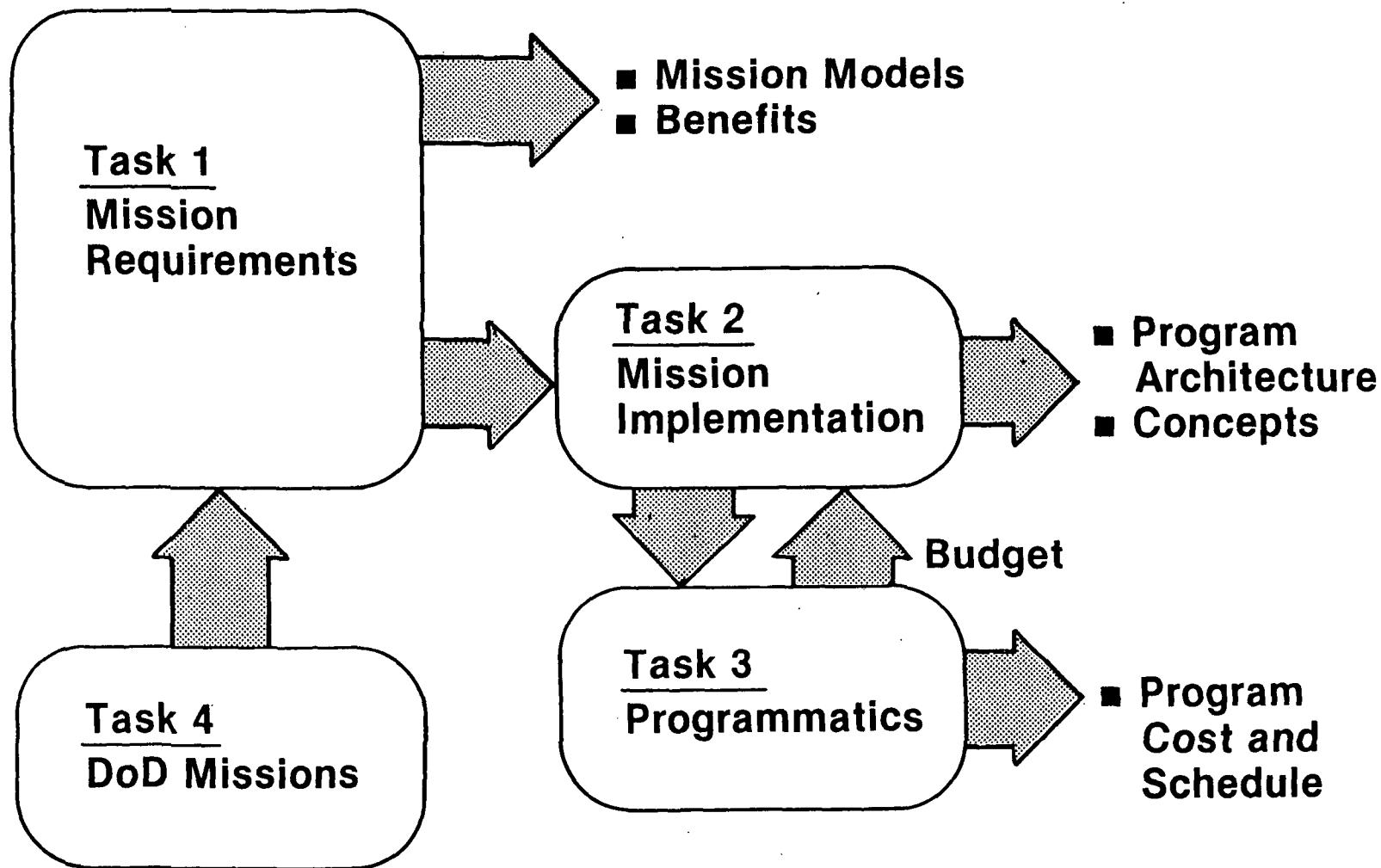
MANNED SPACE STATION — CORE ELEMENT OF THE SPACE STATION SYSTEM

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MDAC STUDY APPROACH



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FEATURES OF MDAC STUDY APPROACH

- Complete First Study Cycle by Midterm
- Extensive Use of Background Data Base
- Use of Mission Advisory Panels
- Emphasis on Commercial Missions
- Buffered Access to Key Commercial Users
- Use of “Seed Ideas” to Stimulate New Missions
- Primary Focus on Initial Capability Needs
- Budget Constrained Optimization: “Build-To-Budget”
 - Missions
 - Concepts
 - Programs

MDAC PROGRESS VERSUS PLAN — MIDTERM

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- User Interaction Plan Implemented**
- User Orientation Briefing Package Completed**
- 63 User Contacts Completed**
- 365 Missions Defined; 95 Selected**
- Computerized Mission Data Base Is Operating**
- Budget Alternatives Are Defined**
- Cost Models Are Defined**
- System Costing Model Is Operating**
- Architectural Options Are Defined**
- Strawman Program Is Selected**

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First Study Cycle Complete As Planned

RESULTS SUMMARY — MISSIONS

| Mission Categories | Identified Missions | Mid-Term Selections | Architecture and Concept Drivers |
|-----------------------------|---------------------|---------------------|-----------------------------------------------------------------------|
| 1. Science/ Applications | 137 | 40 | Orbit Location, Stability, Field of View, Contamination Control |
| 2. Commercial | 61 | 12 | High Power, Proprietary Control, Man |
| 3. National Security | 65 | 6 | Secure Operations, Endurance/Survivability, Hardening |
| 4. Operational Support | 25 | 25 | Teleoperators, Manipulators, Depot Services, Man |
| 5. Technology Development | 77 | 12 | Exterior OPS (EVA), Man, Hazard Control |
| Totals | 365 | 95 | |

BENEFITS OF MAN IN ORBIT

Functions



Scientist/Observer

- Real-Time Data Analysis
- Multiple Sensor Use
- Sensor Mode/Parameter Selection
- Cooperation With Principal Investigator
- Target Selection

Typical
Tasks

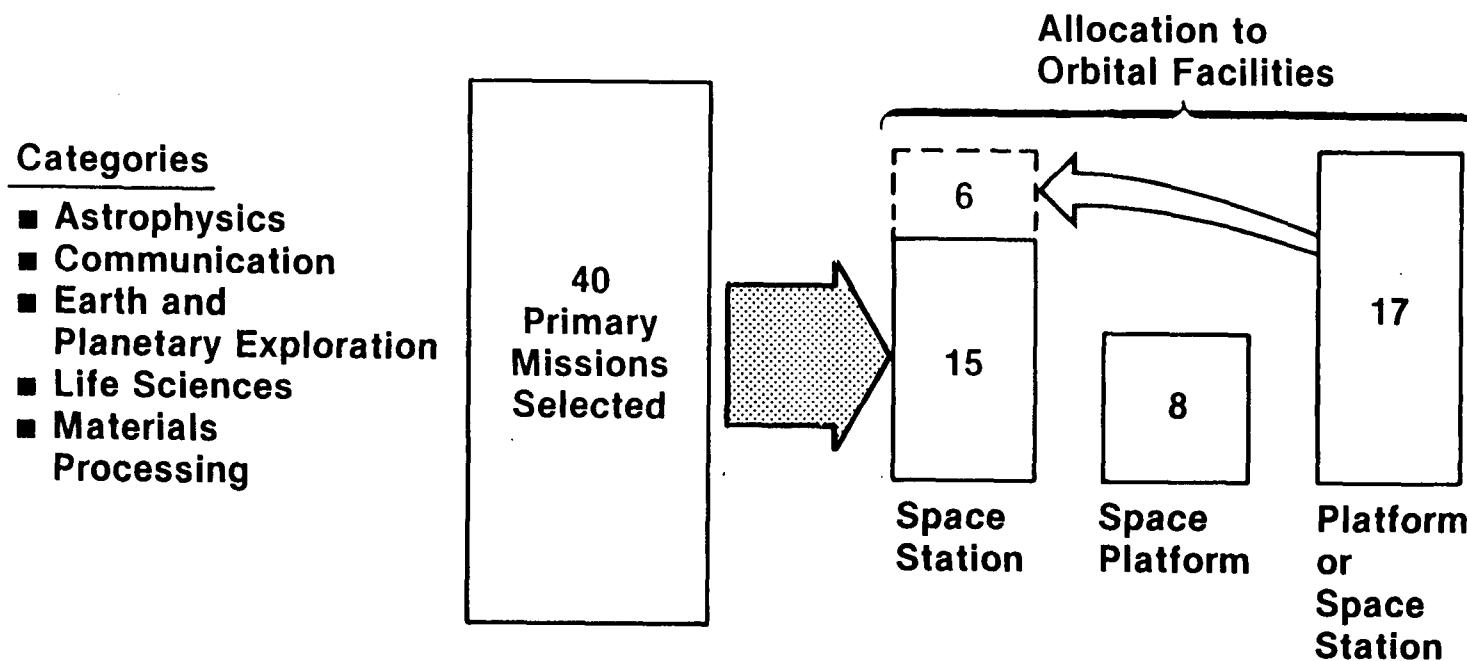
Development Engineer

- Sensor Operation
- Sensor Evaluation
- Component Testing

Technical Operations Specialist

- Equipment Setup, Checkout, Maintenance, Calibration
- Servicing of Sensor and Equipment Consumables

SCIENCE AND APPLICATION MISSIONS



Results:

- 15 Missions Require Manned Station
- 6 Others Will Benefit Significantly

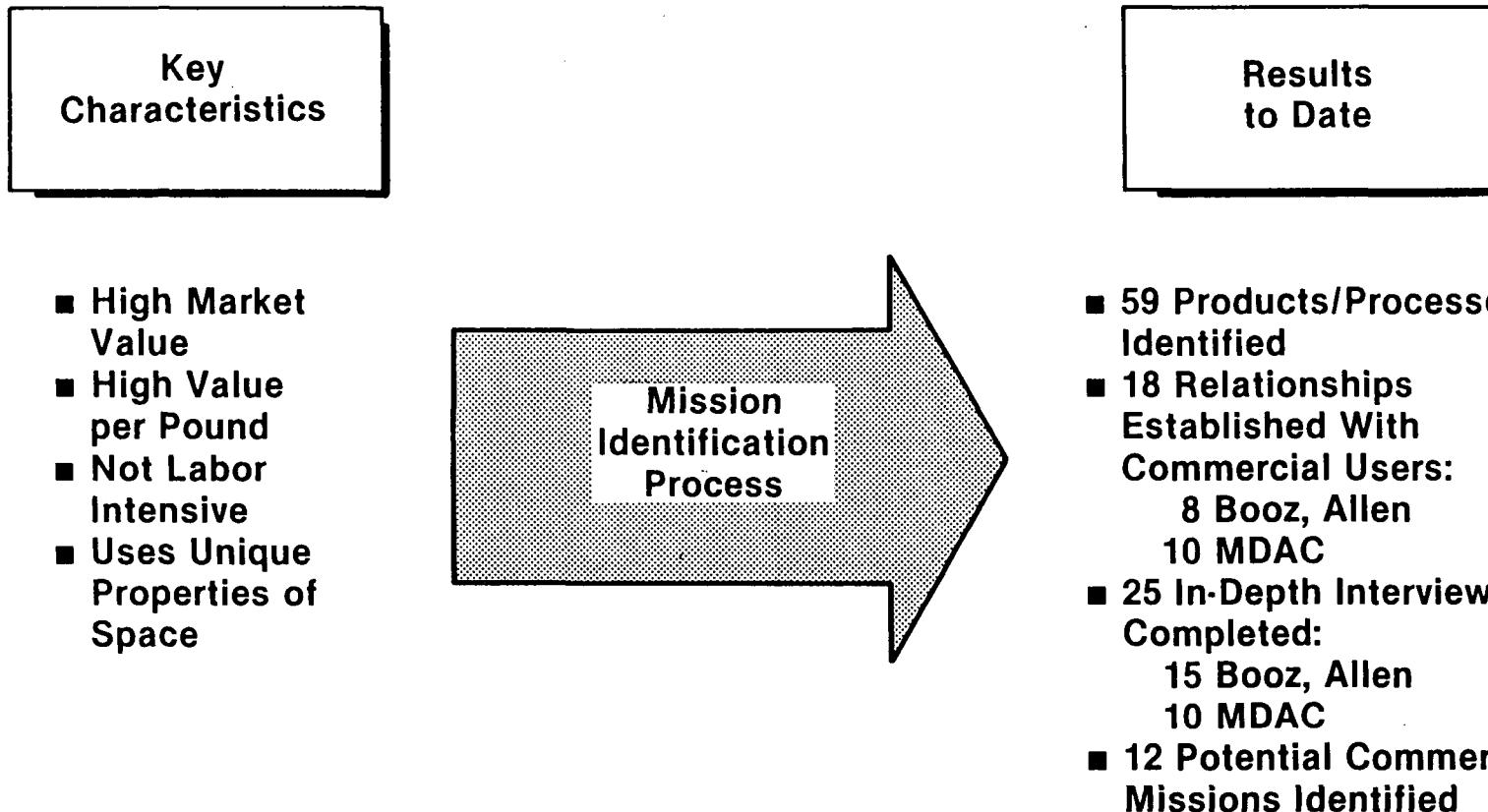
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REQUIREMENTS DEFINITION – ASTROPHYSICS (TYPICAL)

| | MASS (KG) | ALTITUDE (KM) | INCLINATION (DEG) | POWER (KW) | HEAT REJECTION (KW) | FIELD OF VIEW (DEG) | POINTING (ARCMIN) | STABILITY ARCSEC/ TIME | DATA RATE (MBPS) |
|-------------------------------|-----------------|------------------|----------------------|---------------|---------------------------|---------------------------|----------------------|------------------------------|------------------------|
| SOT | 8,200 | 400 | 57 | 6.8 | 0.9 | 0.025 | 0.017/90 | 0.1/15 | 50 |
| SIRTF | Mass (kg) | 100 | Inclination (deg) | 1.3 | Heat Rejection (kW) | 0.125 | Pointing (arcmin) | 2/20 | |
| STARLAB | 100 | 100 | | 2.2 | 0.8 | 0.8 | 10/30 | | |
| SCRN | 100 | 100 | | 0.8 | 70 | | N/A | | |
| SOLAR SOFT X-RAY TELESCOPE | 1,300 | 430 | 57 | 0.2 | 0.2 | | | 0.1 | |
| STO | 16,600 | Altitude (km) | 57 | Power (kW) | | Field of View (deg) | 8-12 | Stability sec/Time | |
| PINHOLE X-RAY CAMERA | 10,000 | | 97 | | | | | | |
| X-RAY OBSERVATORY | 3,600 | 400 | 28.5 | 0.9 | 0.9 | | 1.0 | | |
| HRS | 1,800 | 400 | < 45 | 0.5 | 0.5 | 10 | 6/90 | 36/0.02 | 0.03 |
| XTE | 1,000 | 400 | 28.5 | 0.6 | 0.6 | | | | |
| AXAF | 10 TO 12,000 | 500 | 28.5 | 2.0 | 2.0 | | 30 | 1.0 | |
| LAMAR | 9,500 | 400 | 28 | 3.4 | 0.4 | 1 | 3/67 | 10/0.02 | 0.1 |
| VLBI | 1,400 | 400 | 57 | 0.9 | 0.9 | 0.1 | 2.5/45 | 150/60 | 12 |
| ASO | 12,500 | 400 | 57 | 4.1 | | 0.025 | 0.17/90 | 0.1/15 | 42 |

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COMMERCIAL MISSIONS



COMMERCIAL MISSIONS

(12 Identified to Date)

| MDAC Data Bank Identifier | Areas of Responsibility | |
|---------------------------|-----------------------------|-----|
| | MDAC | BAH |
| CIR001 | Materials Research Facility | ● |
| CMP001 | Electrophoretic Processes | ● |
| CMP002 | Silicon Ribbon Manufacture | ● |
| CMP003 | Crystals/Diffractors | ● |
| CMP004 | Melting/Refreezing | |
| CMP005 | Homogeneous Mixtures | ● |
| CMP006 | Directional Crystal Growth | ● |
| CMP007 | Hot/Cold Processes | ● |
| CMP008 | Unidirectional Processes | ● |
| CMP009 | Earth Observations | ● |
| CMP010 | Materials Production | ● |
| CMP011 | Misc Operations | ● |

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BENEFITS ANALYSIS ELECTROPHORETIC PROCESSES

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| Human Needs | Product Objective | Current Status |
|--------------------------------------------|----------------------------------------------------------------|---------------------------------------|
| Growth Hormone (850,000)* | Stimulates Juvenile Bone Growth, Promotes Healing of Ulcers | Research Quantities, Low Purity |
| Beta Cells (3,200,000) | Single Injection Cure for Diabetes | Clinical Quantities, Not Separable |
| α - Antitrypsin (500,000)* | Limit Emphysema Disease State, Enhance Cancer Chemotherapy | Research Quantities, Low Purity |
| Epidermal Growth Factor (1,100,000)* | Skin Burn and Wound Healing | Research Quantities Low Purity |
| Interferon (20,000,000)* | Viral Infection Immunity | Low Yield and Purity |
| Antihemophilic Factor (15,000)* | Eliminate Immunological Reactions for Hemophilia | Low Purity and Loss of By-Products |

*Annual Patient Load — U.S. Market

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MANNED SPACE STATION OPERATIONS

Enhances Rate of New Product Additions

- 15 Products in 10 Years With Space Station vs 3 Products With Unmanned Free-flyer
- Product Characterization Time Is Reduced From 1 or 2 Years to a Few Months
- Production Time for Clinical Materials is Reduced From 1 or 2 Years to a Few Months
- Dedicated Facilities and Manned Operation Allows:
 - Multiple Product Evaluation
 - Parallel Operations
 - Quick Turnaround

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COMMERCIAL MISSIONS MID-TERM INDICATIONS

- Electrophoresis Is Highest Confidence COMMERCIAL PRODUCTION Mission Identified to Date
 - Major Obstacles to Space Exploitation
 - Proprietary Issue
 - Cost
 - Time Delay
 - Risk
 - Unknowns
 - Attractive Alternatives
 - Large Scale Production Will Ultimately Require Independent Facilities, Privately Funded
- 
- An Available R&D Space Facility is Best Incentive

CANDIDATE NATIONAL SECURITY MISSIONS

| Source \ Mission Area | R&D | Data Fusion Center | Space Command Post | Service and Logistics | Surveillance and Reconnaissance | Weapon Platform |
|----------------------------------------|-----|--------------------|--------------------|-----------------------|---------------------------------|-----------------|
| Source | | | | | | |
| Military Space System Technology Model | | | 1 | 24 | | 1 |
| Space Policy and Requirements | | | | 16 | | |
| Space Policy and Advanced Concepts | | | | 7 | 1 | |
| Military Space Station Study | 3 | 1 | 1 | 1 | 1 | 2 |
| Legacy Missions | 3 | | | | | |
| New Ideas | | | | 2 | 1 | |
| Total 65 | 6 | 1 | 2 | 50 | 3 | 3 |

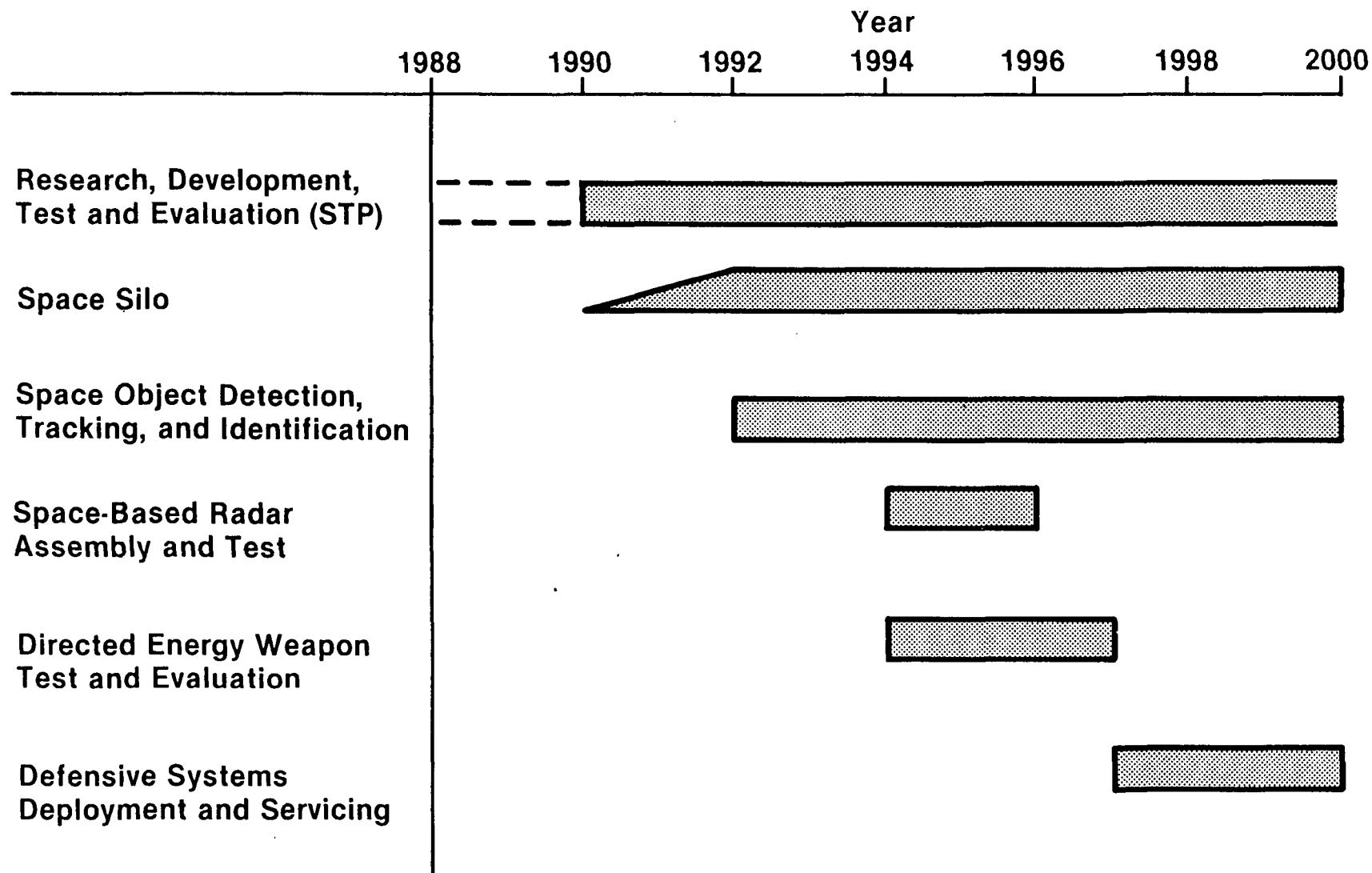
Results
to Date

- 65 Candidate Missions Defined
- Most Require Dedicated Satellites
- 6 Are Space Station Candidates

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NATIONAL SECURITY MISSIONS MIDTERM

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SPACE OPERATIONS MISSIONS

- Transportation (OTV, TMS, Other)
 - Deploy/Retrieve
 - Debris Collection
- Assembly, Integration, Checkout
 - Large Structures
 - Stage/Payload Mating
- Service
 - Maintain/Repair/Replenish
 - Instrument Reconfiguration
- Storage
 - Propellants (Cryo, Storables)
 - Spares
 - Payloads
- Space Utilization
 - Quarantine
 - Rescue

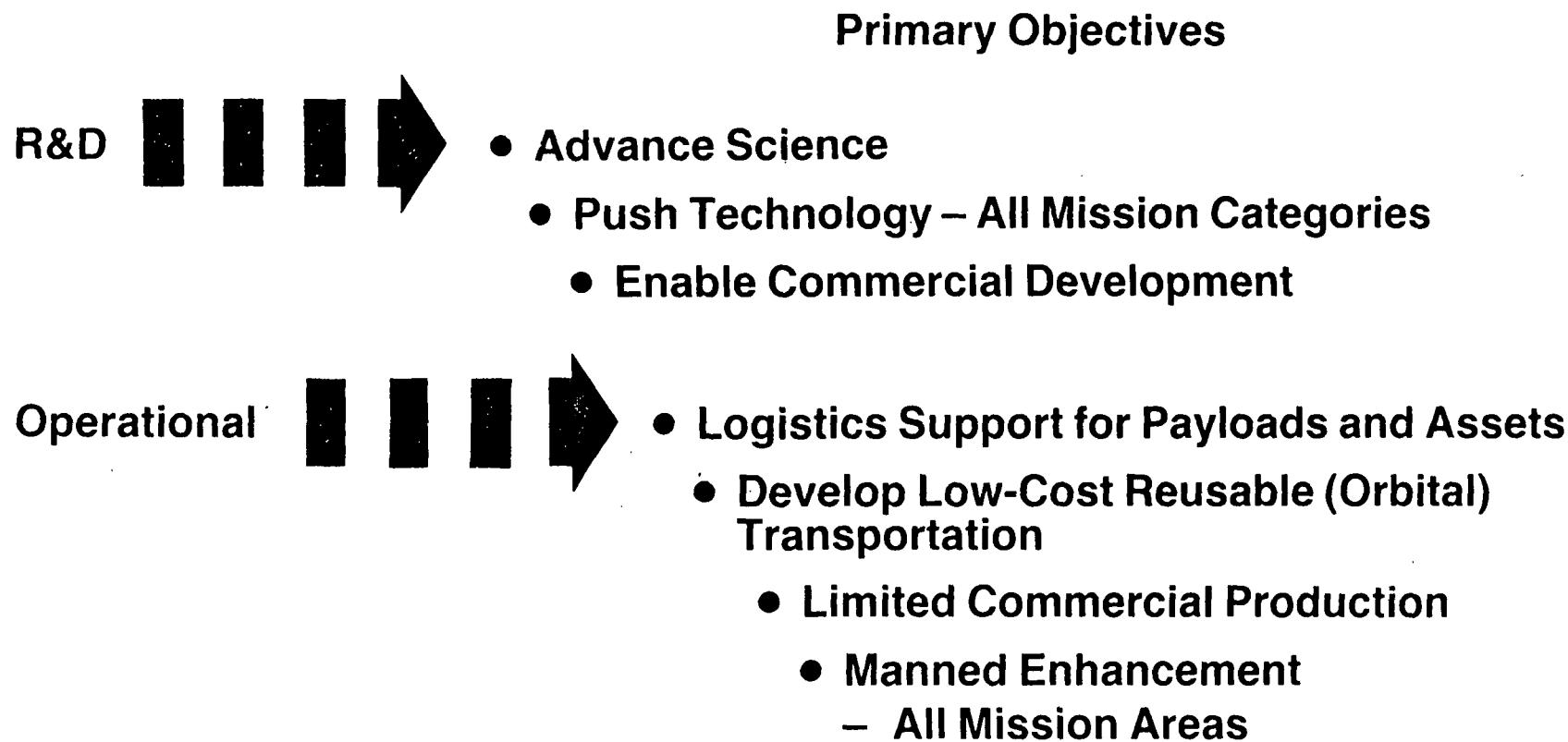
Results to Date

- Teleoperator
 - Is Required
- Satellite Servicing
 - Low Cost
 - High Payoff
- Cryogen Depot
 - Offers Major Economic Benefit
- Man Participation is Essential

TECHNOLOGY DEVELOPMENT MISSIONS

- Utilize the Unique Space Station Environment
 - Enable:
 - Advanced Mission Technology
 - Increased Space Station Capability
 - Provide Benefits to All Categories of Users
 - Majority Require Manned Participation
 - Are Relatively Short Term and Orbit Independent
- 

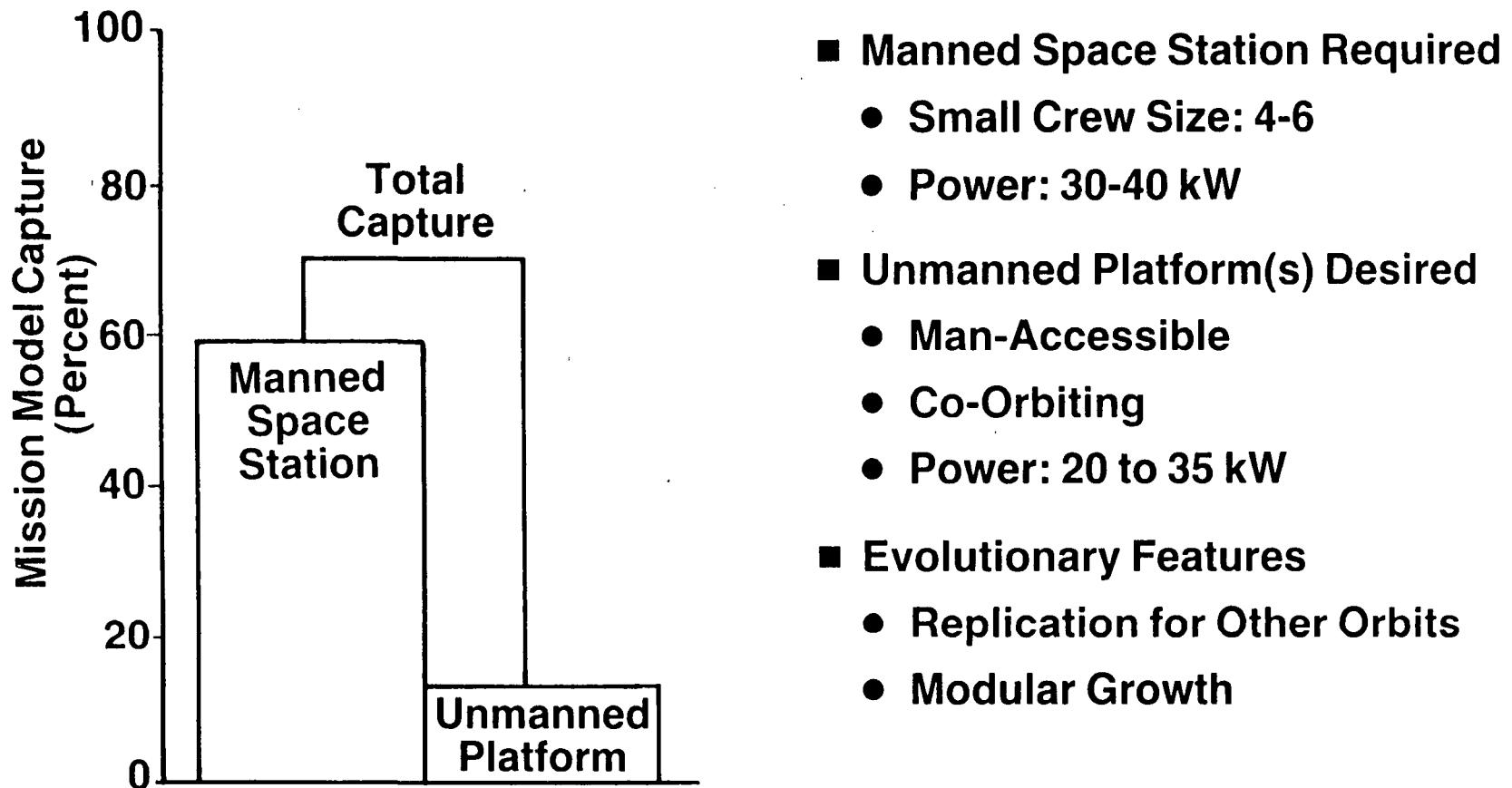
MISSION FOCUS – INITIAL CAPABILITY



REQUIREMENTS ACCOMMODATION

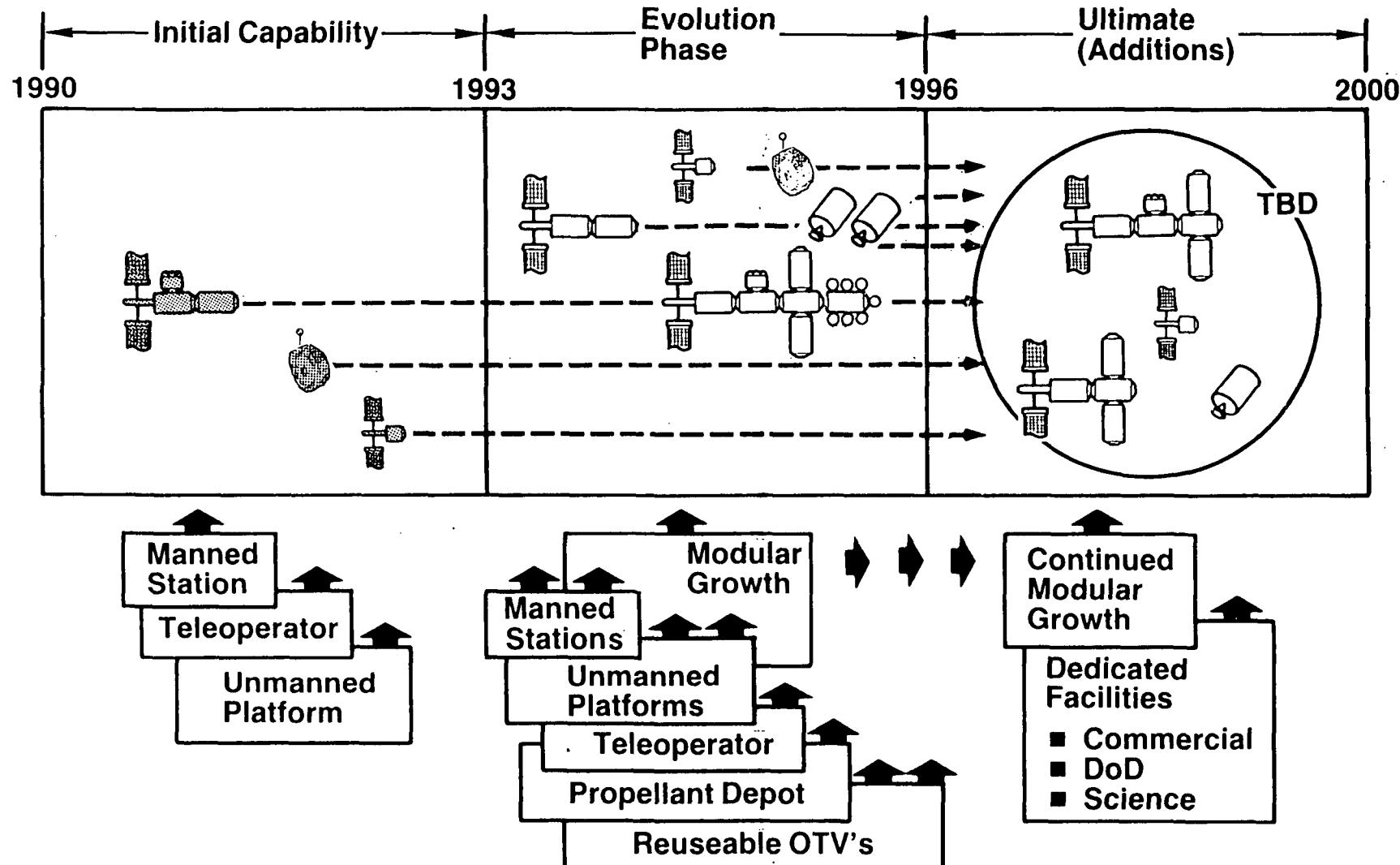
INITIAL CAPABILITY

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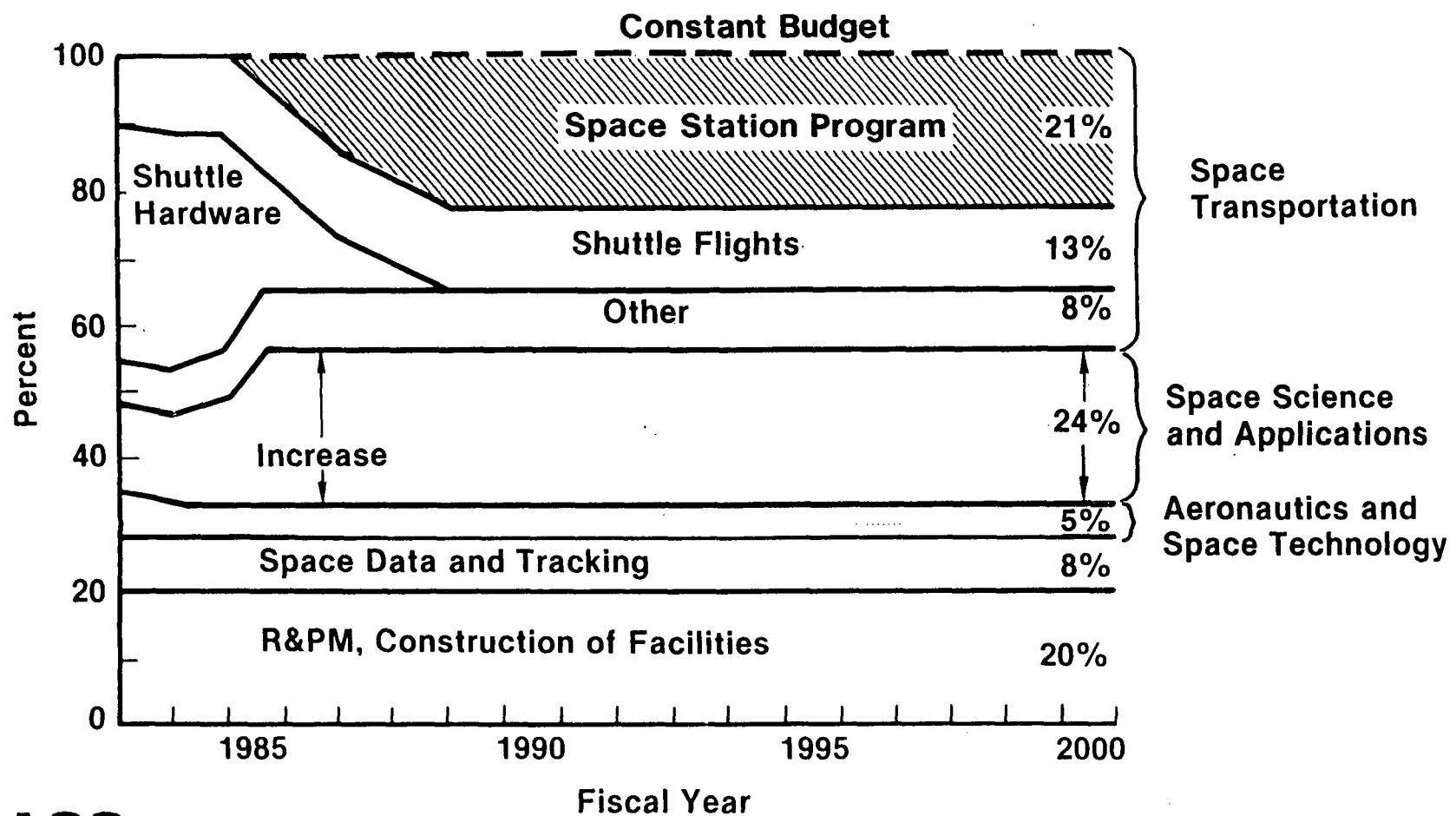
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CAPABILITY GROWTH OPTIONS



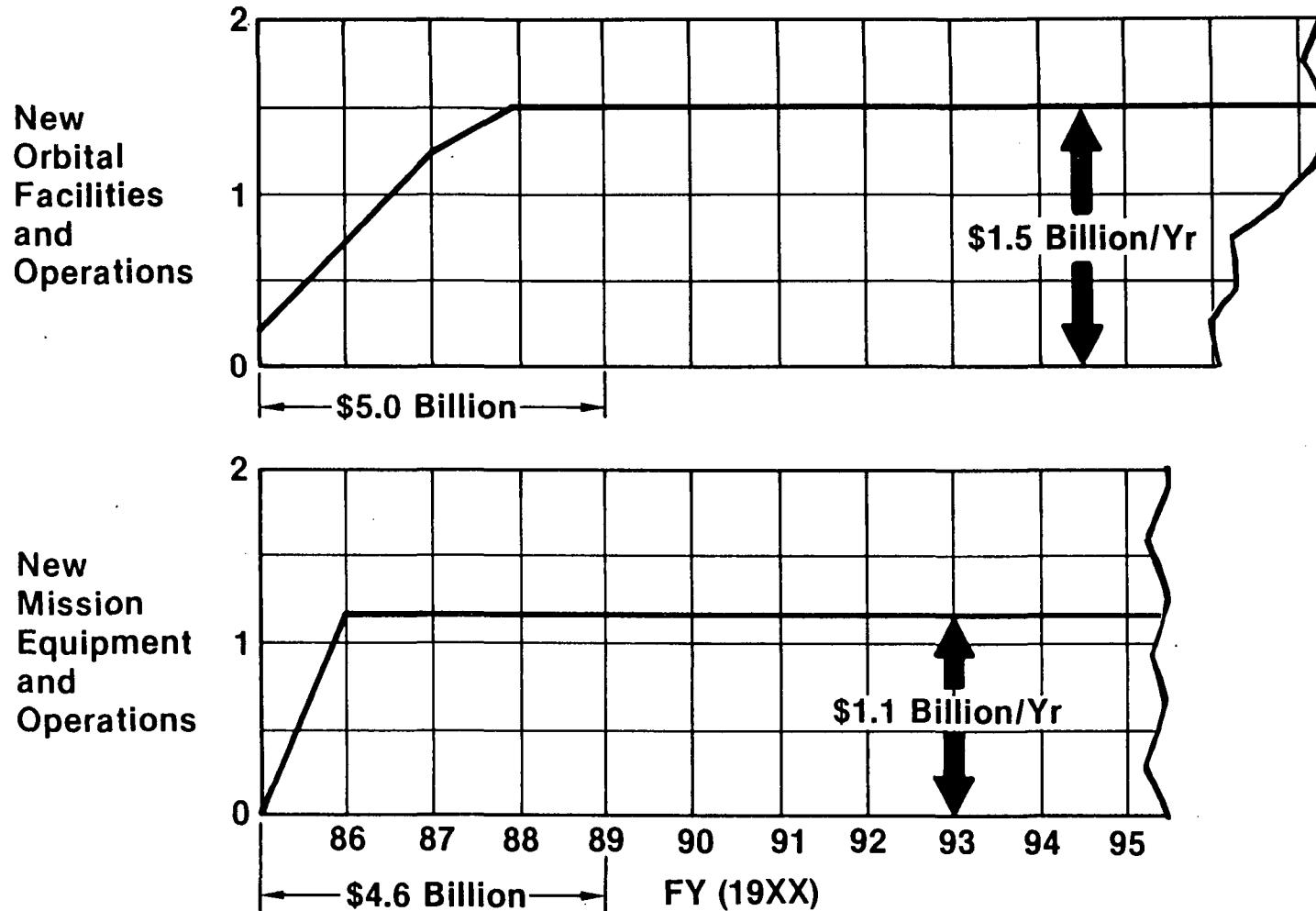
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NASA BUDGET ALLOCATION ASSUMPTIONS



BUDGET MODEL NOMINAL CASE

(Billion Dollars, 1984)



Notes: (1) Science and Applications Budget Increased 60% Above 1983

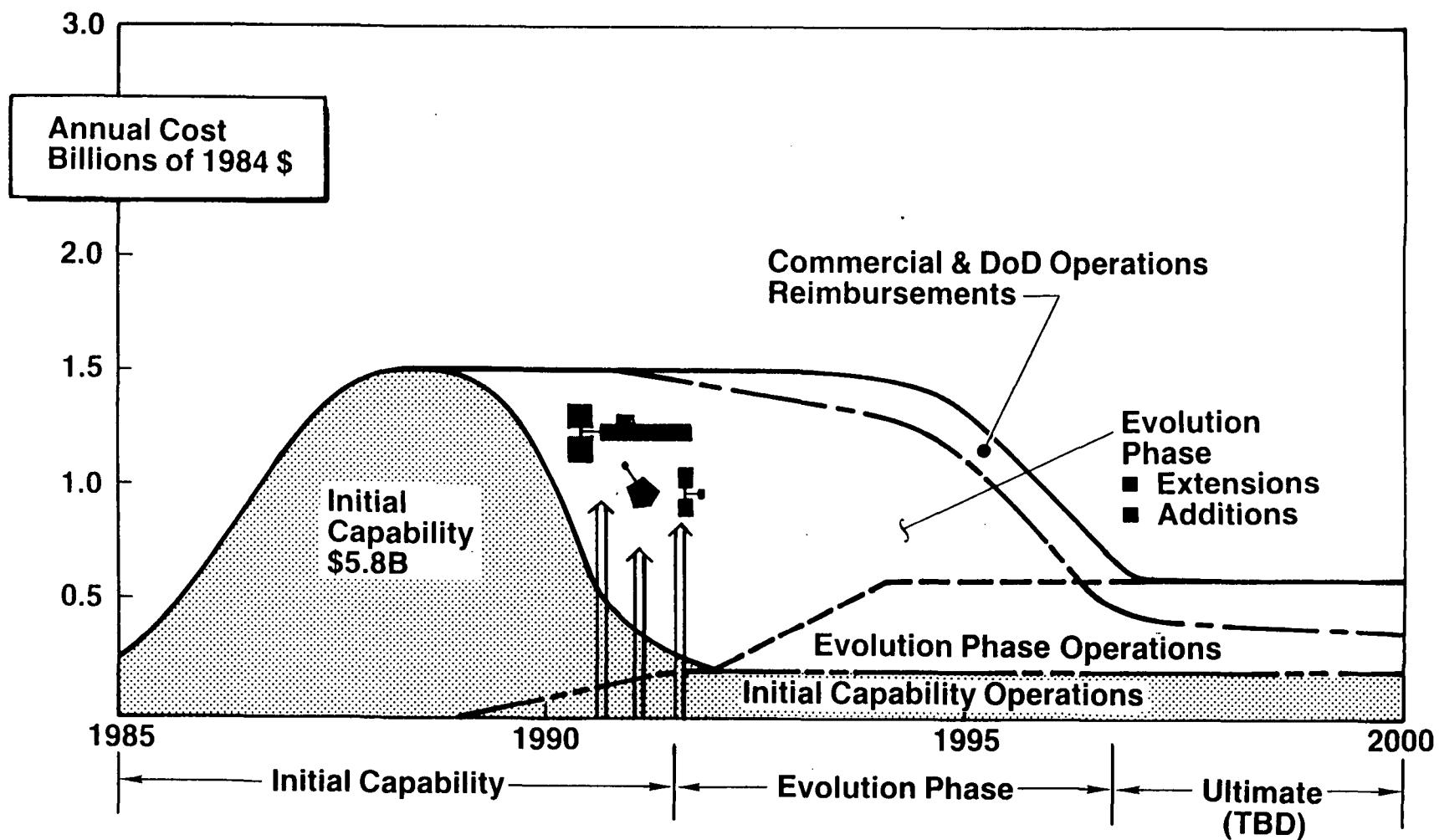
(2) Shuttle Flights Budgeted at \$0.9 Billion/Yr, Are Excluded

A34 (3) All NASA Funds; No Commercial, DoD or Foreign Funds

SAMPLE PROGRAM COSTS

100% MISSION CAPTURE

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MID-TERM SUMMARY

Mission
Needs



- Mission Opportunities and Benefits Sufficient to Justify Space Station
- Early Needs are R&D and Operations Oriented
- Space Station Availability Will Stimulate Commercial Interest

Space
Facilities



- Manned Facility has Highest Mission Capture
- 4-6 Man Crew Indicated
- Unmanned Platforms Highly Desired
- Multiple Orbit Locations Needed

Costs



- Affordable Within Projected NASA Budget

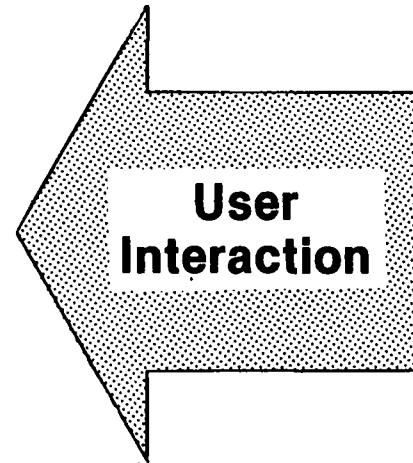
MISSION REQUIREMENTS (TASK 1) METHODOLOGY

Dave Riel

MISSION REQUIREMENTS

- Define Missions in Each Category
- Assess Benefits
- Validate Requirements and Benefits
- Prioritize Missions
- Derive Space Station System Sizing Requirements

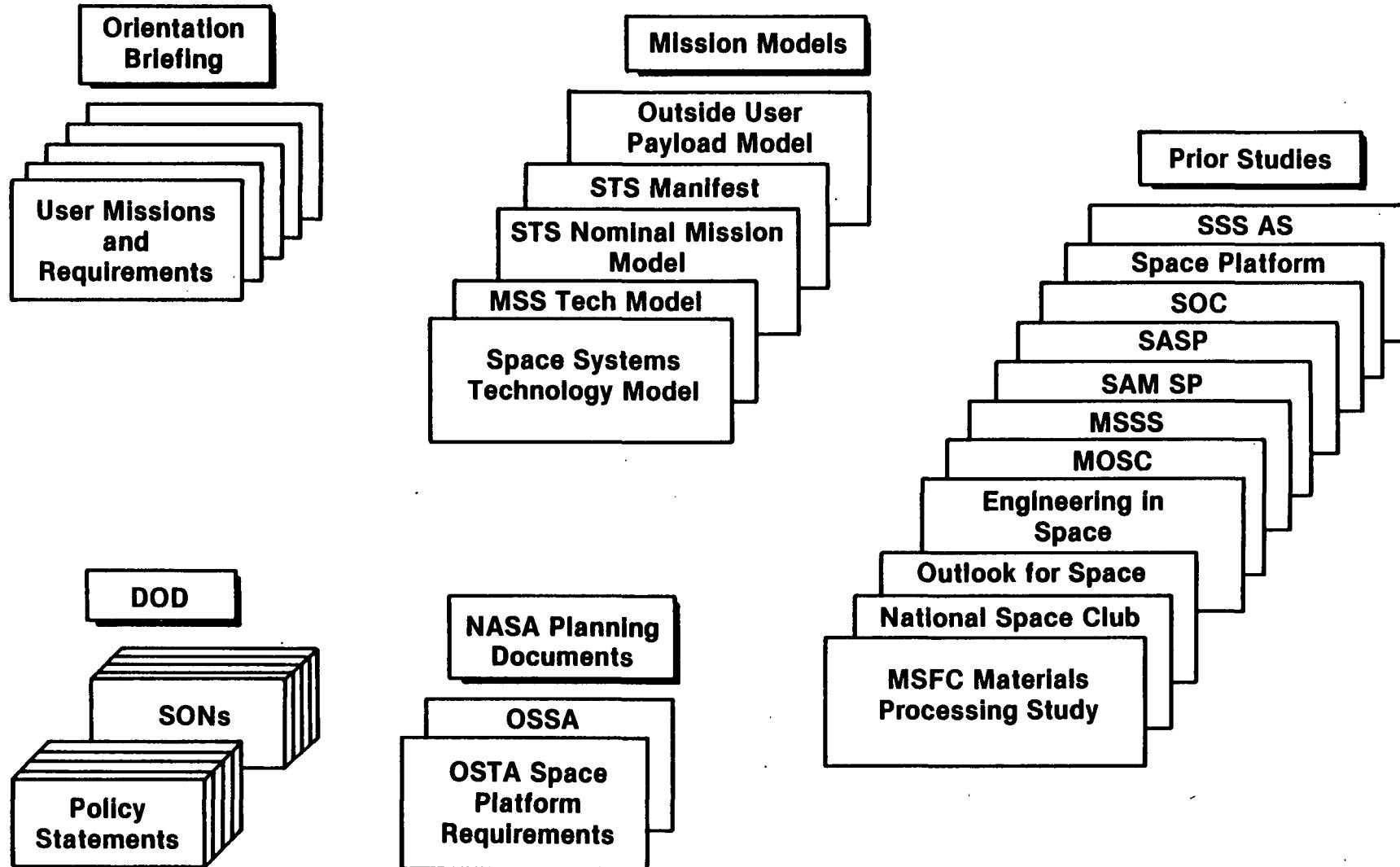
- Manned Space Station
- Platform
- Dedicated Satellites



- Orbit Location
- Volume
- Power
- Crew

B1

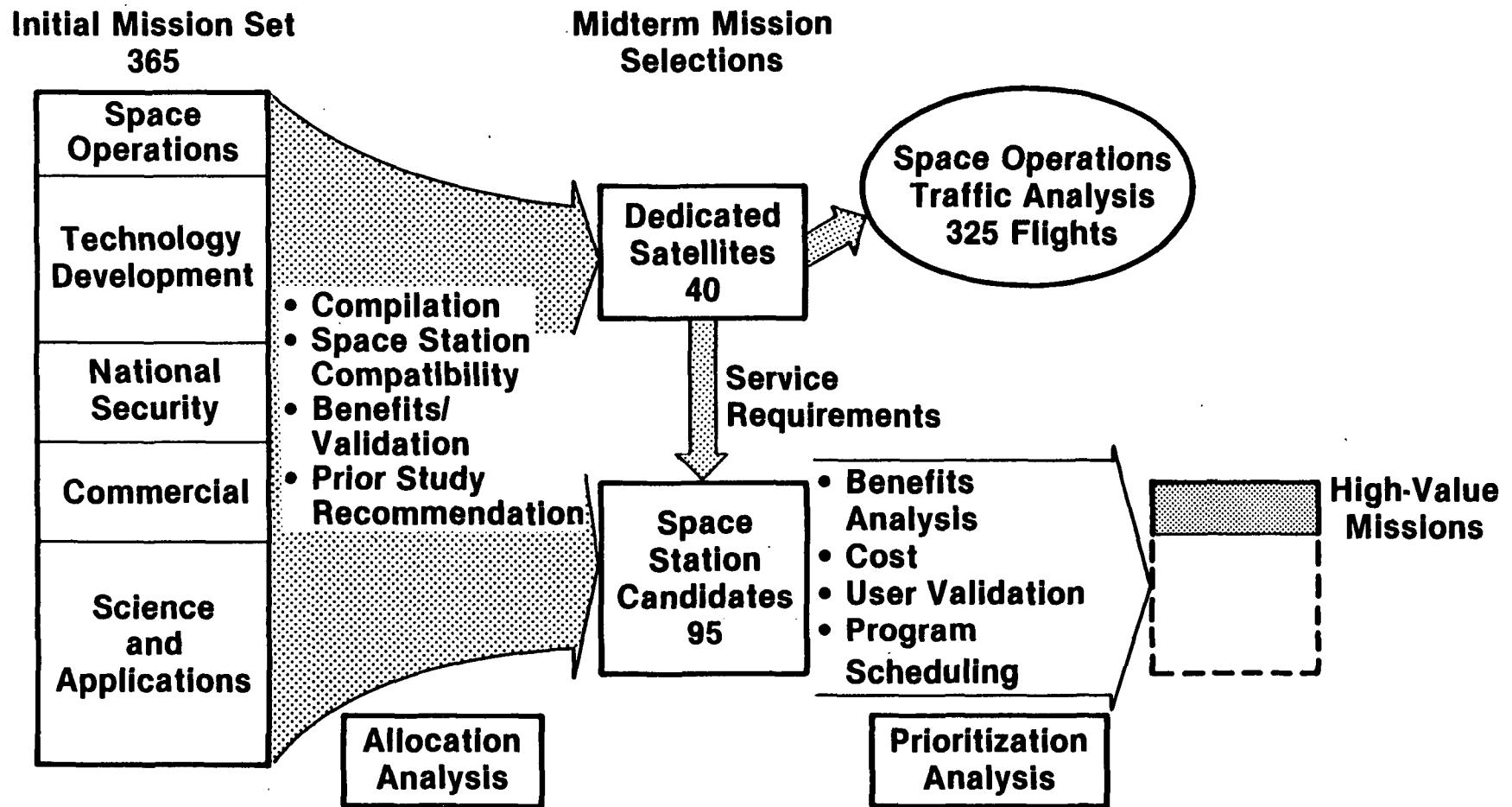
PRIMARY MISSION DATA SOURCES



BENEFIT ANALYSIS TECHNIQUES USED

| Mission Category | Technique | Data/Value |
|------------------------|------------------------------------|-----------------------------------------------------------------------------------------------------------------------------------------------------------------|
| Science/Applications | Peer Judgment Performance | <ul style="list-style-type: none"> ■ Technology Mission Models ■ NASA Planning Documents ■ Increased Capability |
| Commercial | Economic Indicators Performance | <ul style="list-style-type: none"> ■ Market Potential ■ Return on Investment ■ Value Added ■ Increased Throughput |
| National Security | Performance Constituency | <ul style="list-style-type: none"> ■ New Capabilities ■ Augmented Capacity ■ SONs |
| Technology Development | Performance Peer Judgment | <ul style="list-style-type: none"> ■ Enabling Capability ■ Technology Mission Model ■ OAST Plans ■ Subsystem Test Bed |
| Space Operations | Performance Cost | <ul style="list-style-type: none"> ■ Increased Delivery ■ Reduced Flights |

MISSION SELECTION PROCESS



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HIGH-VALUE MISSIONS — MIDTERM

| <u>Category</u> | <u>Mission Capability</u> | <u>Benefit</u> |
|--------------------------|--------------------------------------------------|-----------------------------------------------------------|
| ■ Science/Applications | ■ Solar, Stellar, Earth Orientation Capabilities | ■ Increased Performance |
| ■ Commercial | ■ Electrophoresis | ■ Measured Throughput Increase ■ Investment Commitment |
| ■ National Security | ■ RDT&E Mission Capability | ■ Allows Development of Needed Systems |
| ■ Technology Development | ■ Cryogen Transfer/Storage | ■ Enables High-Performance and Cost Saving OTV |
| ■ Space Operations | ■ Space Telescope Servicing | ■ Reduced Launch Costs |

MIDTERM MISSION DEFINITION

95 Mission Defined

| | | | | |
|---------------------------------------|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|--------------------------------|----------|
| MISSION NAME | CODE | TYPE | | |
| | | Science and Applications <input type="checkbox"/> Astrophysics <input type="checkbox"/> Communications <input type="checkbox"/> Earth and Planetary Exp <input type="checkbox"/> Environmental Observations <input type="checkbox"/> Life Sciences <input type="checkbox"/> Materials Commercial <input type="checkbox"/> Earth and Ocean Operations <input type="checkbox"/> Communications <input type="checkbox"/> Materials Processing <input type="checkbox"/> Industrial Research National Security <input type="checkbox"/> Research and Development | | |
| CONTACT (Name, address, phone) | | | | |
| STATUS | <input type="checkbox"/> Planned <input type="checkbox"/> Operational <input type="checkbox"/> Approved <input type="checkbox"/> Candidate <input type="checkbox"/> Opportunity | | | |
| Year of first flight | | | | |
| Number of missions | | | | |
| ORBIT CHARACTERISTICS | | | | |
| Apogee, km | | Perigee | Tolerance ± | |
| Inclination, deg | | Tolerance ± | | |
| Argument of perigee, deg | | Ephemeris accuracy | | |
| Synchronization | | <input type="checkbox"/> None | <input type="checkbox"/> Earth | |
| | | <input type="checkbox"/> Sun | <input type="checkbox"/> Other | |
| POINTING (Real Time) | | | | |
| View direction | | <input type="checkbox"/> Inertial | <input type="checkbox"/> Solar | |
| | | <input type="checkbox"/> Earth | <input type="checkbox"/> Other | |
| Pointing accuracy | | Field of view | | |
| Specific targets | | Stability angle | | |
| DATA/COMMUNICATIONS | | | | |
| Monitoring requirements | | | | |
| Data rate | | <input type="checkbox"/> On-board data proc <input type="checkbox"/> Encryption/Decryption | | |
| THERMAL | | | | |
| Type of concept | | | | |
| Temperature, deg C | | Operational min | max | Peak |
| Cryogenic Load | | Temperature Duration | | |
| Heat Rejection W | | Operational Peak | | |
| POWER | | | | |
| Operating | | | | |
| Standby | | | | |
| Peak | | | | |
| Voltage, V | | | | |
| CREW REQUIREMENTS | | | | |
| Estimated crew size | | Permanent | Service | EVA |
| Manhours/mission | | Average time between visits, days | | |
| Skills required | | | | |
| PHYSICAL CHARACTERISTICS | | | | |
| Launch mass, kg | | Deployed mass | Expendables | |
| Length, m | | Launch w/OTV | Undeployed | Deployed |
| Diameter, m | | Launch | Undeployed | Deployed |
| Center of gravity location, m | | X | Y | Z |
| SPECIAL CONSIDERATIONS/CLARIFICATIONS | | SKETCH | | |

Name
Code
Type
Contact
Status
Flight Date
Number
Objective
Description
Altitude
Inclination
Pointing Direction
Accuracy
Data Rate
Power
Crew Number
Crew Hours
Mass
Length

MISSION REQUIREMENTS (TASK 1)

USER INTERACTION

Dr. Harry Wolbers

USER INTERACTION

- **User Requirements Define the Market for Space Systems**
- **Our Goal**
 - Understand Needs of Potential Users and Encourage Their Utilization of Future Space Systems Where Appropriate
- **Our Approach**
 - Review of the Literature
 - Review by MDAC Mission Advisory Panels
 - Direct Contact With Representatives of Each Interest Area
- **Our Emphasis in Approach Varies With the Maturity/Heritage of the Area**

USER INTERACTION PLAN

| AREA | STATUS | PURPOSE OF USER CONTACT |
|---------------------------------|-----------------------------------------------------------------------------------------------------------------------------------------------------------------------------|------------------------------------------------------------------------------------------------------------------------|
| Science and Applications | <ul style="list-style-type: none"> ■ Requirements Well Documented ■ Benefit of Prior Studies and Continuing Peer Reviews | <ul style="list-style-type: none"> ■ Validate Our Understanding of Current Plans |
| Commercial | <ul style="list-style-type: none"> ■ Emerging Area ■ Little Hard Documentation | <ul style="list-style-type: none"> ■ Continuing Contacts Needed to Stimulate New Insights |
| National Security | <ul style="list-style-type: none"> ■ Requirements Documented but Facilities Not Defined | <ul style="list-style-type: none"> ■ Validate Our Understanding of Needs and Offer Ideas |
| Operations and Space Technology | <ul style="list-style-type: none"> ■ Previous Studies and Documentation Provide Point of Departure ■ New Requirements Emerge As Systems Are Defined | <ul style="list-style-type: none"> ■ Validate Our Understanding of Current Requirements and Offer Ideas |

C2

USER CONTACTS TO DATE

| MISSION AREAS | INTERVIEWS |
|--------------------------------------------------------|-------------------|
| Science and Applications | 5* |
| — (MDAC Science Advisory Panel) | |
| Commercial | 25 |
| — (15 Booz-Allen, 10 MDAC) | |
| National Security | 16 |
| — (DoD Space Division Headquarters, Los Alamos) | |
| Operations and Space Technology | 17 |
| — (MSFC, JSC, JPL, MDTSCO and 6 Aerospace Contractors) | |
| Total to Date | 63 |

*Contacts Limited to MDAC Advisory Panels Pending Review With NASA Space Station Task Force Science and Application Mission Panel

MISSION REQUIREMENTS (TASK 1)

SCIENCE AND APPLICATIONS

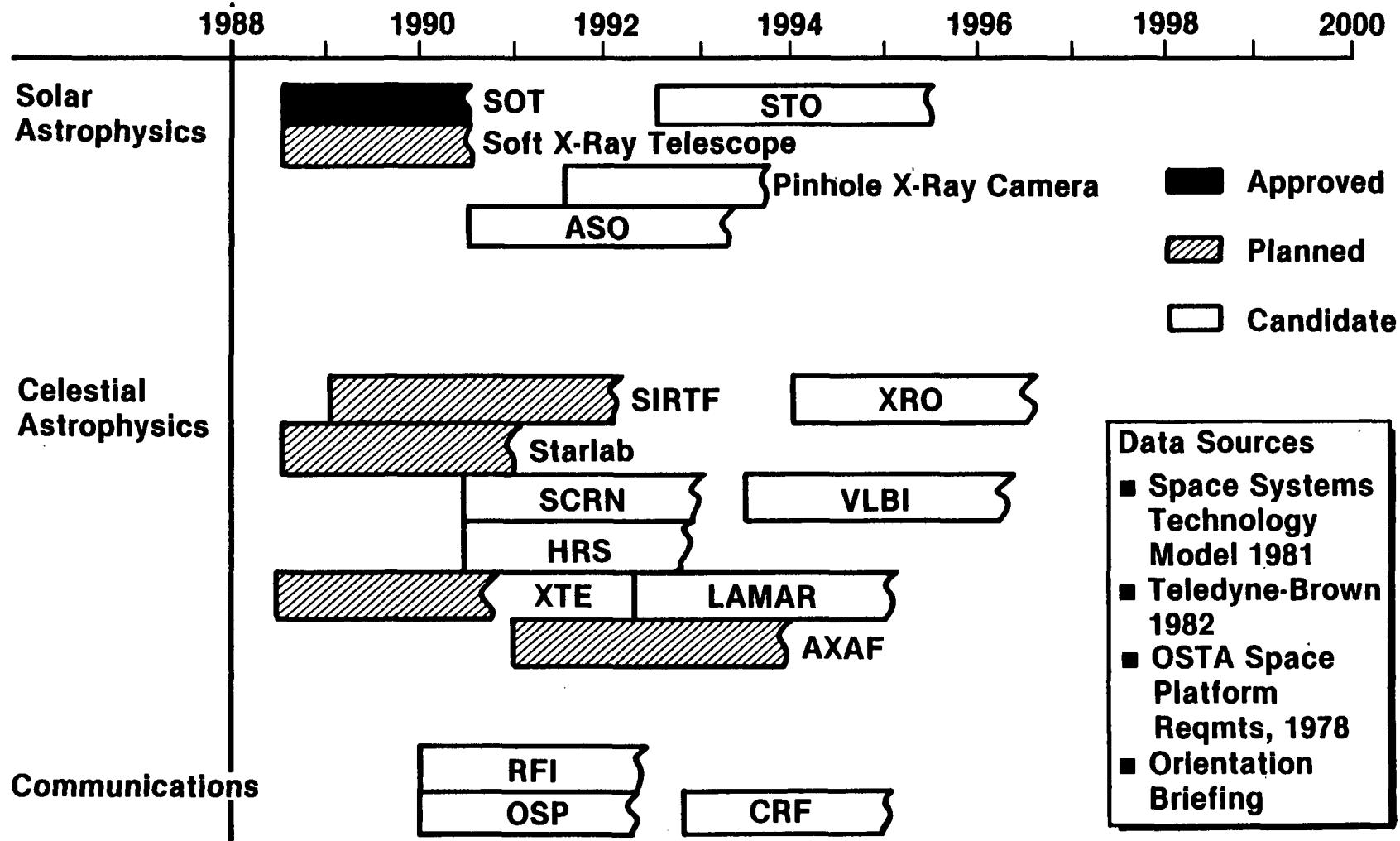
MISSIONS

Dr. Harry Wolbers

SCIENCE AND APPLICATIONS MISSION PLANS

SELECTION BASED ON PEER JUDGMENT

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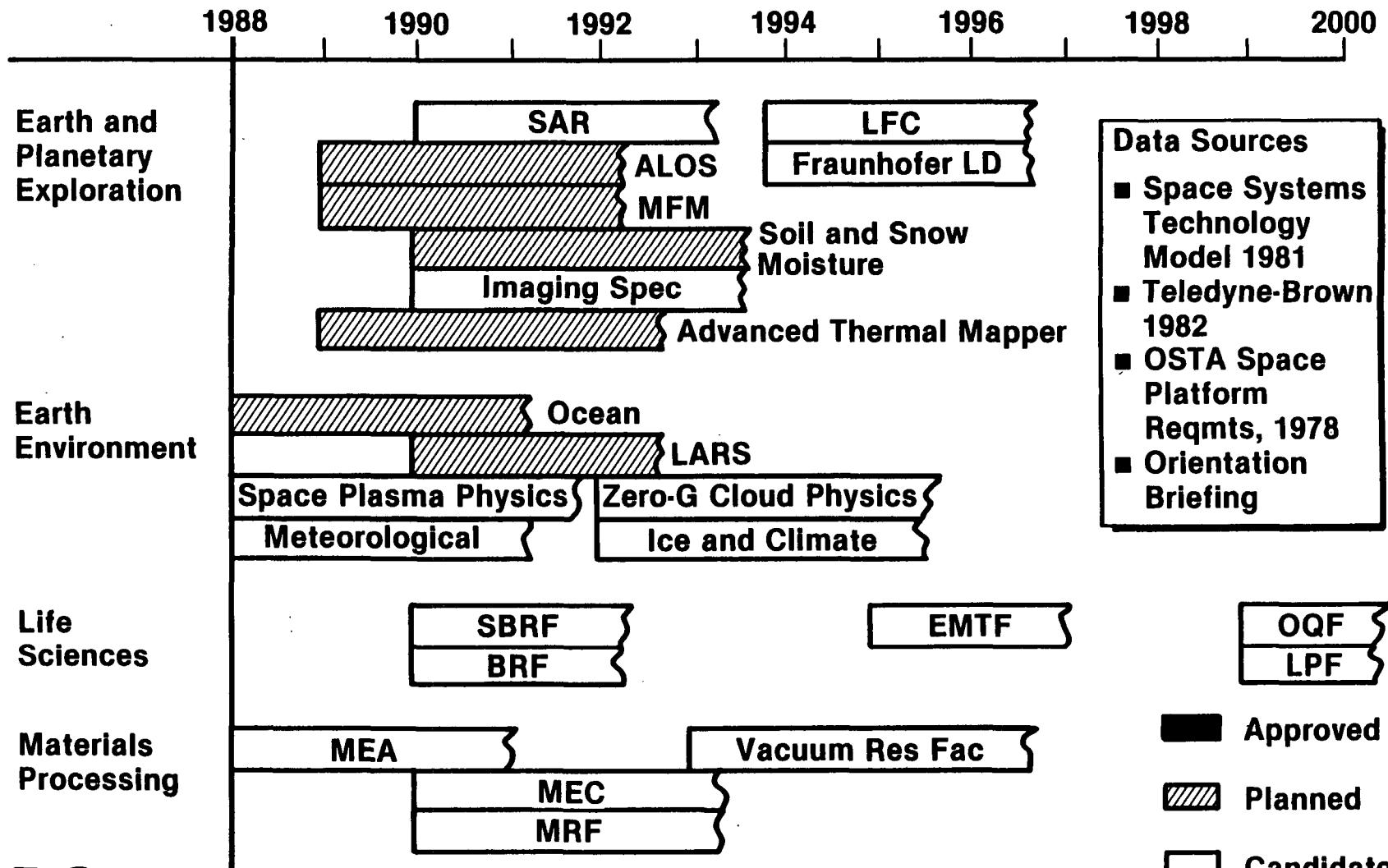


D1

SCIENCE AND APPLICATIONS MISSION PLANS

SELECTION BASED ON PEER JUDGMENT

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D2

SCIENCE AND APPLICATIONS MISSIONS

- **Astrophysics**
- **Communications**
- **Earth and Planetary Exploration**
- **Environmental Observations**
- **Life Sciences**
- **Materials Processing**

ASTROPHYSICS PAYLOADS

OBJECTIVES

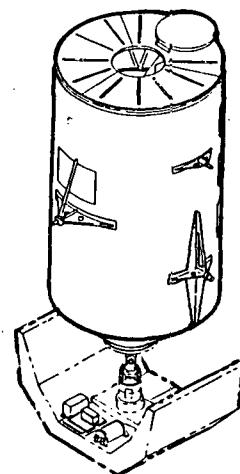
- Investigate Properties of Extragalactic Space, the Milky Way Galaxy, and the Solar System
- Address the Key Questions of Cosmic Evolution

KEY MISSION DRIVERS

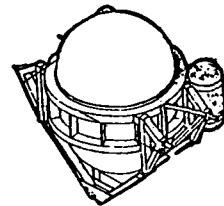
- Precise Pointing and Stability Requirements (SIRTF, STARLAB, SOT, ASO)
- Hot Object Avoidance Zones for Celestial Instruments
 - 90° Zone Around Sun (SIRTF, LAMAR)
 - 60° Zone From LOS (SIRTF)
- High Slew Rates ($> 40^\circ/\text{Min}$) (SIRTF, STARLAB)
- High Data Rates (SOT, ASO, VLBI)
- Focal Plane Instruments Susceptible to Radiation (SIRTF, STARLAB)
- Optical Instruments Sensitive to Contaminants/Condensation
- Some Instruments Vent He, Xe, CH₄ (SCRN, LAMAR)

D3

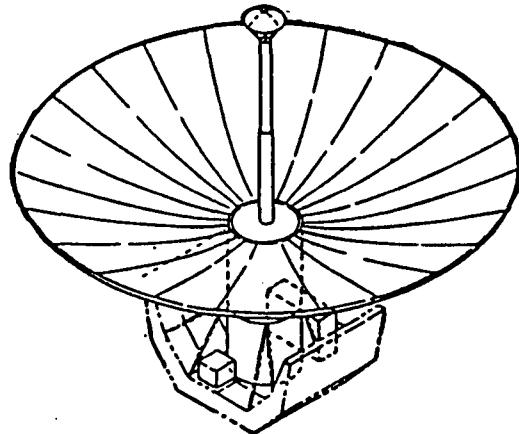
ASTROPHYSICS INSTRUMENTS/FACILITIES



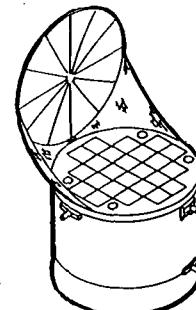
Solar Optical
Telescope (SOT)



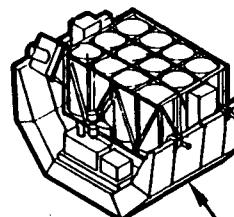
Spectra of
Cosmic Ray
Nuclei (SCRN)



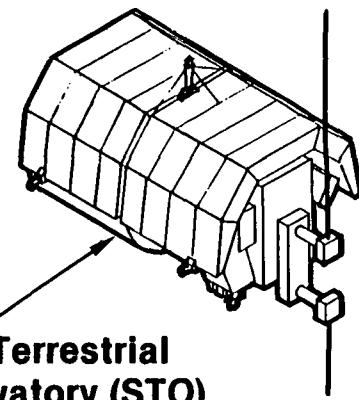
Very Long Baseline
Interferometry (VLBI)



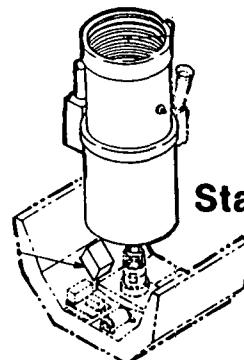
Large Area Modular
Array of Reflectors
(LAMAR)



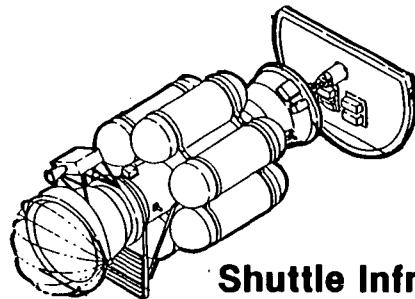
Solar Terrestrial
Observatory (STO)



Advanced Solar
Observatory (ASO)



Starlab



Shuttle Infrared
Telescope Facility
(SIRTF)

CHARACTERISTICS OF ASTROPHYSICS INSTRUMENTS/FACILITIES

| | MASS (KG) | ALTITUDE (KM) | INCLINATION (DEG) | POWER (KW) | HEAT REJECTION (KW) | FIELD OF VIEW (DEG) | POINTING (ARCMIN) | STABILITY ARCSEC/TIME | DATA RATE (MBPS) |
|----------------------------|--------------|---------------|-------------------|------------|---------------------|---------------------|-------------------|-----------------------|------------------|
| SOT | 8,200 | 400 | 57 | 6.8 | 0.9 | 0.025 | 0.017/90 | 0.1/15 | 50 |
| SIRTF | Mass (kg) | 100 | Inclination (deg) | 1.3 | Heat Rejection (kW) | 0.125 | Pointing (arcmin) | 2/20 | |
| STARLAB | | 100 | | 2.2 | | 0.8 | | 10/30 | |
| SCRN | | 100 | | 0.8 | | 70 | | N/A | |
| SOLAR SOFT X-RAY TELESCOPE | 1,300 | 430 | 57 | 0.2 | 0.2 | | | 0.1 | |
| STO | 16,600 | Altitude (km) | 57 | Power (kW) | | Field of View (deg) | 8-12 | Stability sec/Time | |
| PINHOLE X-RAY CAMERA | 10,000 | | 97 | | | | | | |
| X-RAY OBSERVATORY | 3,600 | 400 | 28.5 | 0.9 | 0.9 | 1.0 | | | |
| HRS | 1,800 | 400 | < 45 | 0.5 | 0.5 | 10 | 6/90 | 36/0.02 | 0.03 |
| XTE | 1,000 | 400 | 28.5 | 0.6 | 0.6 | | | | |
| AXAF | 10 TO 12,000 | 500 | 28.5 | 2.0 | 2.0 | | 30 | 1.0 | |
| LAMAR | 9,500 | 400 | 28 | 3.4 | 0.4 | 1 | 3/67 | 10/0.02 | 0.1 |
| VLBI | 1,400 | 400 | 57 | 0.9 | 0.9 | 0.1 | 2.5/45 | 150/60 | 12 |
| ASO | 12,500 | 400 | 57 | 4.1 | | 0.025 | 0.17/90 | 0.1/15 | 42 |

D5

CHARACTERISTICS OF ASTROPHYSICS INSTRUMENTS/FACILITIES

| | MASS (KG) | ALTITUDE (KM) | INCLINATION (DEG) | POWER (KW) | HEAT REJECTION (KW) | FIELD OF VIEW (DEG) | POINTING (ARC MIN) | STABILITY (ARC SEC/ TIME) | DATA RATE (MBPS) |
|-------------------------------|-----------------|------------------|----------------------|---------------|---------------------------|---------------------------|-----------------------|---------------------------------|------------------------|
| SOT | 8,200 | 400 | 57 | 6.8 | 0.9 | 0.025 | 0.017/90 | 0.1/15 | 50 |
| SIRTF | 5,300 | 400 | 28.5 | 1.3 | 0.8 | 0.125 | 1.6/20 | 2/20 | 1 |
| STARLAB | 3,300 | 400 | 28 | 2.2 | 1.0 | 0.8 | 2.5/30 | 10/30 | 7 |
| SCRN | 3,100 | 400 | 57 | 0.8 | 0.8 | 70 | CONT | N/A | 0.1 |
| SOLAR SOFT X-RAY TELESCOPE | 1,300 | 430 | 57 | 0.2 | 0.2 | | | 0.1 | |
| STO | 16,600 | 400 | 57 | 9.2 | | VARIOUS | 0.08-120 | 2-1800 | |
| PINHOLE X-RAY CAMERA | 10,000 | 370 | 97 | | | | | | |
| X-RAY OBSERVATORY | 3,600 | 400 | 28.5 | 0.9 | 0.9 | | 1.0 | | |
| HRS | 1,800 | 400 | < 45 | 0.5 | 0.5 | 10 | 6/90 | 36/0.02 | 0.03 |
| XTE | 1,000 | 400 | 28.5 | 0.6 | 0.6 | | | | |
| AXAF | 10 TO 12,000 | 500 | 28.5 | 2.0 | 2.0 | | 30 | 1.0 | |
| LAMAR | 9,500 | 400 | 28 | 3.4 | 0.4 | 1 | 3/67 | 10/0.02 | 0.1 |
| VLBI | 1,400 | 400 | 57 | 0.9 | 0.9 | 0.1 | 2.5/45 | 150/60 | 12 |
| ASO | 12,500 | 400 | 57 | 4.1 | | 0.025 | 0.17/90 | 0.1/15 | 42 |

COMMUNICATIONS PAYLOADS

VFX887

OBJECTIVES

- Provide Orbital R&D Facilities for Measurement of: Terrestrial Noise; Ionospheric Effects; Tropospheric Dielectric Properties and Transmissibility; Multipath Linkages
- Develop Space Deployment and Calibration Techniques
- Evaluate Millimeter and Optical (Including Laser) Systems
- Develop Potential Application in Nav, Traffic Control, Search and Rescue, etc.

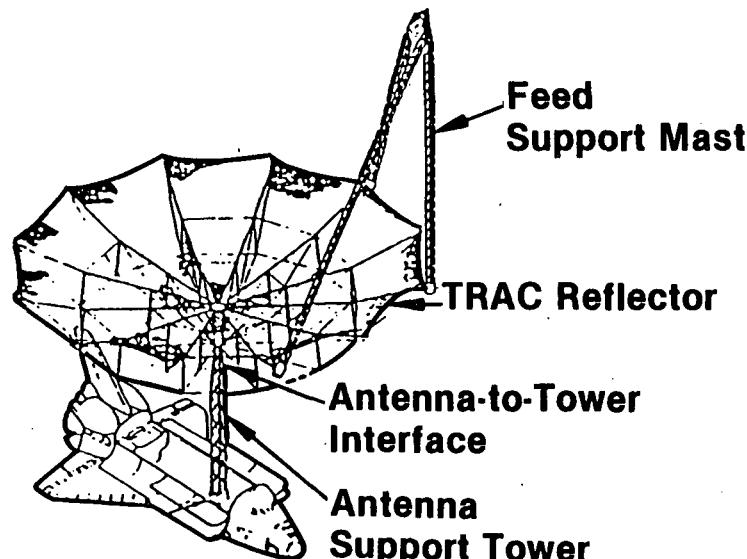
KEY MISSION DRIVERS

- Payloads Sensitive to EMI and RFI
- Require Attitude to Within 1.7 Arc Min
- Require Position to Within 100 Meters
- Continuous Operation of Some Payloads
- Potentially High Data Rates and/or Onboard Storage of Data
- Simultaneous Operation of Multiband Space Transponders and Variety of Antennas
- Multiple Directional Pointing

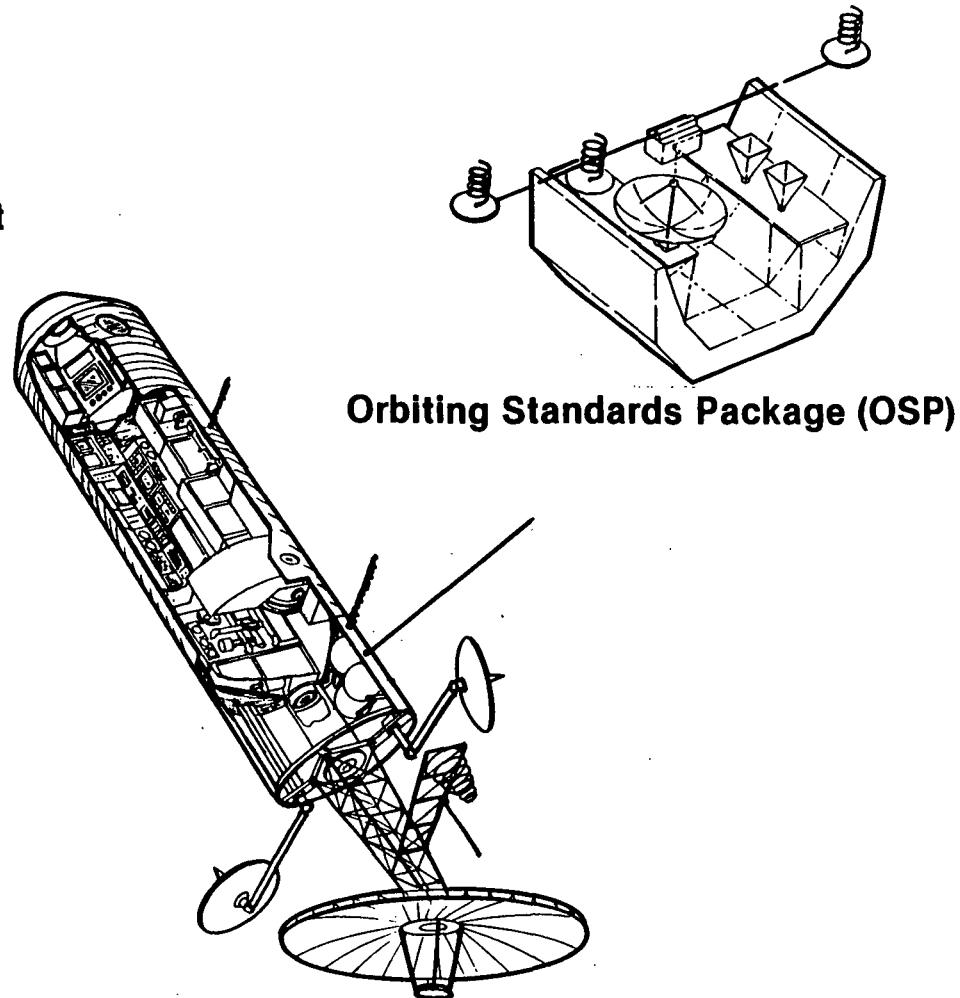
D6

COMMUNICATIONS INSTRUMENTS/FACILITIES

VFX865



**Remote Sensing and
RFI Measurements (RFI)**



Manned Communications Research Facility (CRF)

D7

CHARACTERISTICS OF COMMUNICATIONS INSTRUMENTS/FACILITIES

VFX874

| | MASS (KG) | ALTITUDE (KM) | INCLINATION (DEG) | POWER (KW) | HEAT REJECTION (KW) | FIELD OF VIEW (DEG) | POINTING (ARC MIN) | STABILITY (ARC SEC/ TIME) | DATA RATE (MBPS) |
|----------------------------------------|--------------|------------------|----------------------|---------------|---------------------------|---------------------------|-----------------------|---------------------------------|------------------------|
| REMOTE SENSING/RFI | ~ 2,500 | ANY | > 57 | 0.12 | | | ~ 30 | ~ 150 | 0.005 |
| ORBITING STANDARDS PACKAGE | 50–100 | | 57 | 0.15 | | | 0.5 | | 0.1 |
| COMMUNICATIONS RESEARCH FACILITY | 15,000 | 435 | 57 | 25 | 25 | 90 | 1.7 | 35 | 120 |

EARTH AND PLANETARY EXPLORATION

OBJECTIVES

Development of Remote Sensing Capabilities for

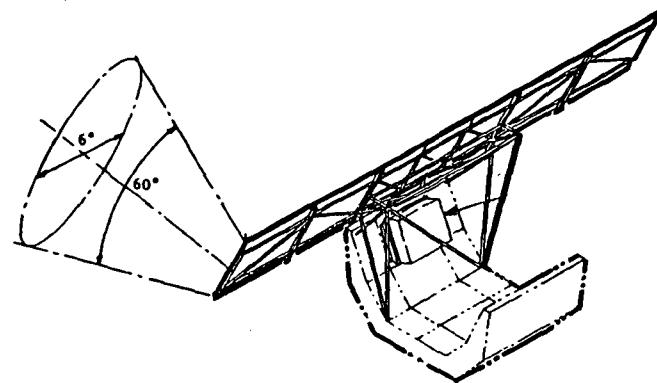
- **Earth Resources Assessment**
- **Crop Monitoring and Forecasts**
- **Cartography**
- **Water Resources and Management**

KEY MISSION DRIVERS

- **High Resolution**
- **Broad Spectral Coverage**
- **Global Coverage**
- **Repeatable Ground Track**
- **High Data Rates (SAR to 120 MBPS)**
- **High Power (to 6 kW)**
- **SAR Susceptible to RFI**
- **Some Instruments May Leak N₂ (IS)**
- **Simultaneous Operation of Instruments**

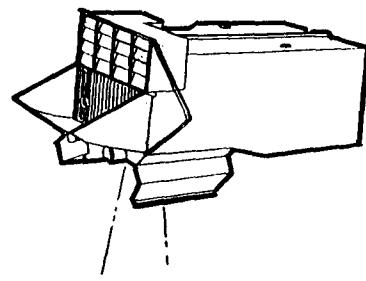
EARTH AND PLANETARY EXPLORATION INSTRUMENTS/FACILITIES

VFX864

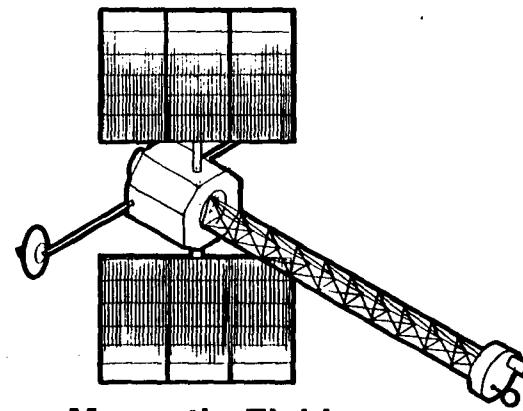


Synthetic Aperture Radar (SAR)

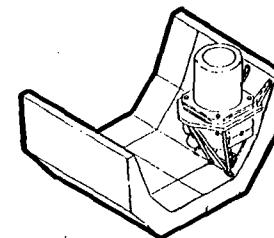
D9



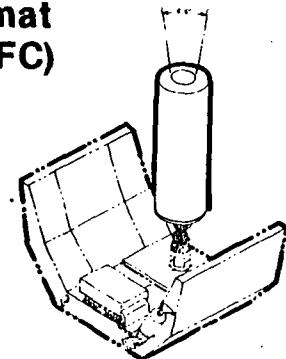
Multispectral Linear Array (Used on Adv Land Obs Sys, ALOS)



Magnetic Field Mapper (MFM)



Large Format Camera (LFC)



Imaging Spectrometer (IS)

CHARACTERISTICS OF EARTH AND PLANETARY EXPLORATION INSTRUMENTS

VFX873

| | MASS (KG) | ALTITUDE (KM) | INCLINATION (DEG) | POWER (KW) | HEAT REJECTION (KW) | FIELD OF VIEW (DEG) | POINTING (ARC MIN) | STABILITY (ARC SEC/ TIME) | DATA RATE (MBPS) |
|----------------------------------|--------------|------------------|----------------------|---------------|---------------------------|---------------------------|-----------------------|---------------------------------|------------------------|
| SYNTHETIC APERTURE RADAR | 1,900 | 400 | 57-90 | 6.5 | | 6 x 60 | 60 | 70 | 120 |
| ADV LAND OBS SYS | 300 | 4-700 | 57-98 | 0.3 | | | | | |
| MAGNETIC FIELD MAPPER | 800 | 300 | 57-97 | 0.1 | | | 30 | 360 | 0.02 |
| SNOW AND MOISTURE ASSESSMENT | 3-500 | 465 | 90 | 1.2 | | 45 | 6 | 100 | 0.2 |
| LARGE FORMAT CAMERA | | 250 | 57 | | | | ~ 0.5 | ~ 1 | N/A |
| IMAGING SPECTROMETER | 1900 | 400 | 57 | 2.8 | | 8.6 | 0.5 | 0.04/0.077 | 100 |
| FRAUNHOFER LINE DISCRIMINATOR | 60 | 2-800 | 28-90 | 0.2 | | 30 x 0.06 | 0.5 | 6 | |
| ADVANCED THERMAL MAPPER | | 400 | 90 | | | | | | |

ENVIRONMENTAL OBSERVATIONS

OBJECTIVES

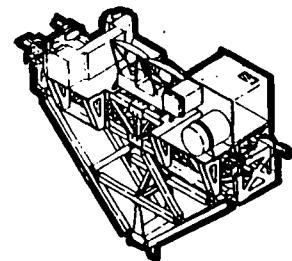
- Atmospheric and Ocean Observations to Further the Fundamental Understanding of the
 - Solar Terrestrial Interactions
 - Effects of Man on Environment
 - Effects of Natural Phenomena on Environment
- Contribute to the Development of Global Environmental Models

KEY MISSION DRIVERS

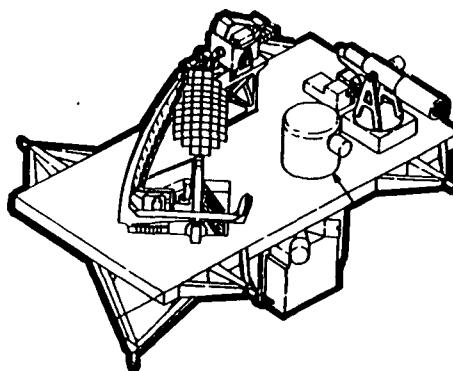
- Global Coverage
- Broad Spectral Coverage
- Long-Term/Coordinated Multisensor, Multidirectional Measurements
- High Data Rates (to 120 MBPS)
- Some Instruments Require Cross-Track Scanning/Viewing
- Continuous Operation
- WISP Antenna Extends to 300 (± 150) Meters and Must Be Aligned with Respect to Magnetic Field
- High Voltages on HF and VLF Transmitters

ENVIRONMENTAL OBSERVATIONS INSTRUMENTS/FACILITIES

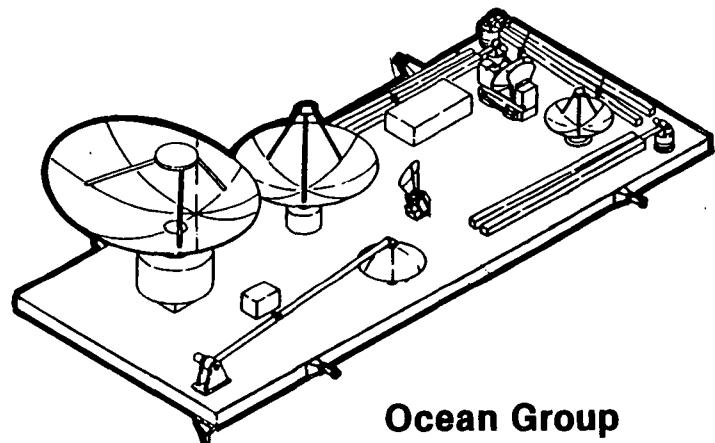
VFX863



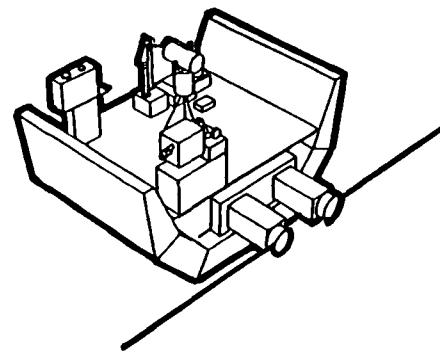
**Meteorological
Payload (MET)**



**Upper Atmosphere
Research Satellite (UARS)**



Ocean Group



Space Plasma Physics (SPP)

D11

CHARACTERISTICS OF ENVIRONMENTAL OBSERVATION PAYLOADS

VFX872

| | MASS (KG) | ALTITUDE (KM) | INCLINATION (DEG) | POWER (KW) | HEAT REJECTION (KW) | FIELD OF VIEW (DEG) | POINTING (ARC MIN) | STABILITY (ARC SEC/ TIME) | DATA RATE (MBPS) |
|-------------------------------|--------------|------------------|----------------------|---------------|---------------------------|---------------------------|-----------------------|---------------------------------|------------------------|
| OCEAN | 10,000 | 400 | 57-90 | 10 | 10 | 1 | 720 | 720 | 120 |
| LARS | 1,200 | 780 | ≥ 60 | 1.7 | | | | | 50 |
| UARS | 2,400 | 400 | 56, 70 | 1.3 | 0.8 | VARIOUS | | | 0.02 |
| SPACE PLASMA PHYSICS | 3,200 | 3-400 | 57-90 | 2.7 | 1.8 | 45 | 60 | 60 | 7.5 |
| ZERO-G CLOUD PHYSICS | 500 | ANY | ANY | 1.4 | | N/A | N/A | N/A | 0.5 |
| METEOROLOGY | 1,200 | 400 | 57 | 1.2 | 0.74 | | 6 | 6 | 0.01 |
| ICE AND CLIMATE EXPERIMENT | 3,500 | 275 | 87 | 2.3 | | | | | 1.4 TO 17.8 |

LIFE SCIENCES FACILITY PAYLOAD

OBJECTIVES

Provide Facilities for

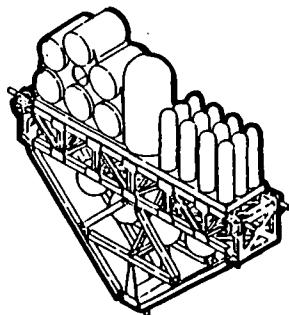
- Understanding the Role of Gravity in Life Sciences
- Addressing the Problems of Long-Duration, Manned Space Missions

KEY MISSION DRIVERS

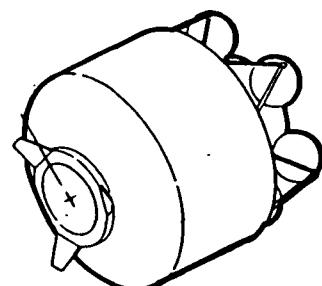
- Low Gravity, $< 10^{-5}$ g's
- Initial Payloads May Operate Unmanned, but "Shirtsleeve" Environment for Active Manned Involvement Desired at Earliest Opportunity
- Living Specimens, Including Man, Serve as Experimental Subjects
- Onboard Centrifugation of Specimens Required
- Continuous Operation Required

D12

LIFE SCIENCES INSTRUMENTS/FACILITIES

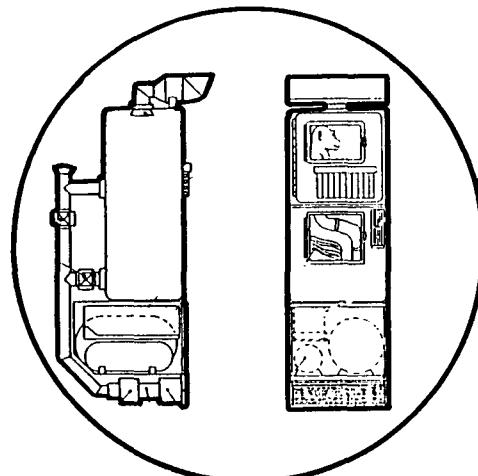
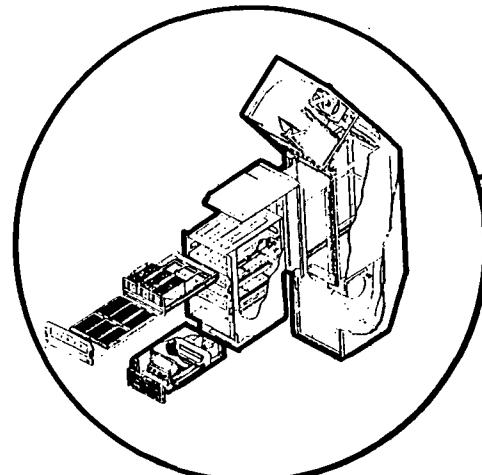


**Early Life
Science Payload**

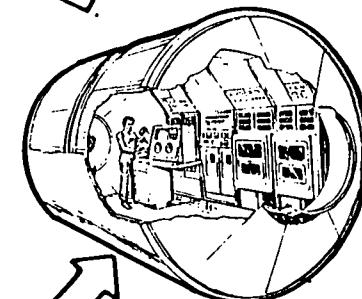


**Pressurized
Life Science
Payload**

**Space Biology
Research Facility (SBRF)**



Large Primate Facility (LPF)



**Biomedical
Research
Facility (BRF)**

D13

CHARACTERISTICS OF LIFE SCIENCES FACILITIES

VFX875

| | MASS (KG) | ALTITUDE (KM) | INCLINATION (DEG) | POWER (KW) | HEAT REJECTION (KW) | FIELD OF VIEW (DEG) | POINTING (ARC MIN) | STABILITY (ARC SEC/ TIME) | DATA RATE (MBPS) |
|-----------------------------------------------|--------------|------------------|----------------------|---------------|---------------------------|---------------------------|-----------------------|---------------------------------|------------------------|
| LARGE PRIMATE FACILITY | ~ 3000 | ~ 400 | ANY | 1.8 | 1.8 | N/A | N/A | N/A | 0.017 |
| SPACE BIOLOGY RESEARCH FACILITY | ~ 3000 | ~ 400 | ANY | 3.5 | 3.5 | N/A | N/A | N/A | 0.05 |
| BIOMEDICAL RESEARCH FACILITY | ~ 3000 | ~ 400 | ANY | 2.4 | 2.4 | N/A | N/A | N/A | 0.016 |
| ORBITING QUARANTINE FACILITY | ~ 3000 | ~ 400 | ANY | 1.7 | 1.7 | N/A | N/A | N/A | TBD |
| EXPERIMENTAL MEDICAL TREATMENT FACILITY | ~ 3000 | ~ 400 | ANY | 1.2 | 1.2 | N/A | N/A | N/A | TBD |

MATERIALS PROCESSING FACILITIES

OBJECTIVES

- Utilize the Unique Features of Space to Process Laboratory Quantities of R&D Materials
- Develop Standards
- Verify Conceptual Approaches to Process/Product Development

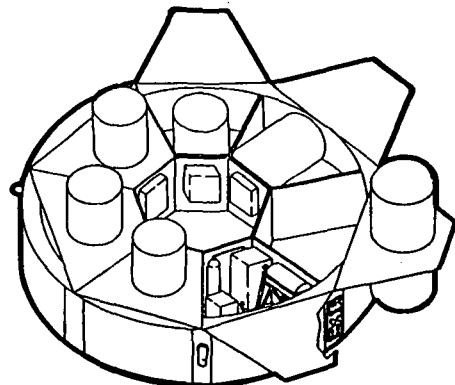
KEY MISSION DRIVERS

- Low Level of Acceleration/Jitter, $< 10^{-5}$ g's
- Vacuum Useful in Some Processes
- Emission of Purge Gases and Process Materials
- High Power Requirements (to 25 kW)
- Long-Duration Missions — Ninety Days or More
- Wide Range of Sample Sizes and Quantities

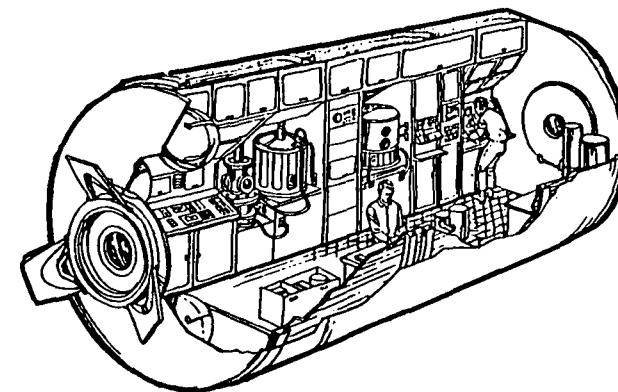
D14

MATERIALS PROCESSING INSTRUMENTS/FACILITIES

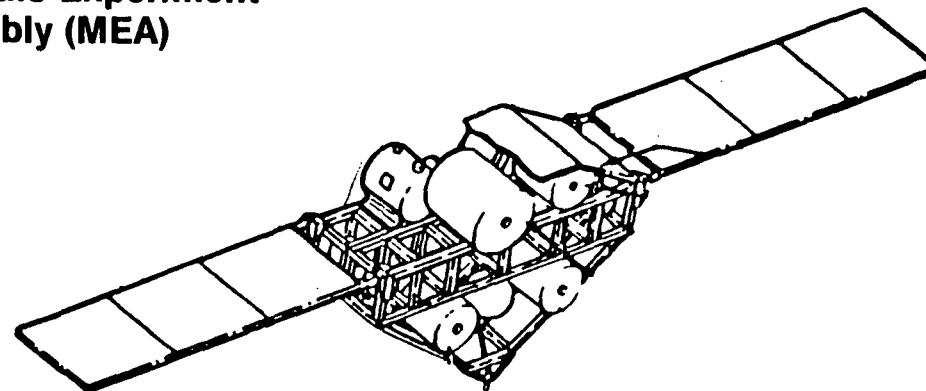
VFX862



**Materials Experiment
Assembly (MEA)**



**Materials Research
Facility (MRF)**



**Materials Experiment
Carrier (MEC)**

D15

CHARACTERISTICS OF MATERIALS PROCESSING FACILITIES

| | MASS (KG) | ALTITUDE (KM) | INCLINATION (DEG) | POWER (KW) | HEAT REJECTION (KW) | FIELD OF VIEW (DEG) | POINTING (ARC MIN) | STABILITY (ARC SEC/ TIME) | DATA RATE (MBPS) |
|------------------------------------|--------------|------------------|----------------------|---------------|---------------------------|---------------------------|-----------------------|---------------------------------|------------------------|
| MATERIAL EXPERIMENT ASSEMBLY | 2,200 | 435 | 28.5-90 | 5 | 5 | N/A | N/A | 10^{-5} G'S | 0.006 |
| MATERIAL EXPERIMENT CARRIER | 7,300 | 435 | 28.5 | 12 | 12 | N/A | N/A | 10^{-5} G'S | 0.5 |
| MATERIAL RESEARCH FACILITY | 7,500 | ~ 400 | ANY | 25 | 25 | N/A | N/A | | 10 |
| SPACE VACUUM RESEARCH FACILITY | 1,000 | 400 | ANY | 4.0 | | N/A | 120 | | |

BENEFITS OF MAN IN ORBIT

Scientist/Observer

- Real-Time Data Analysis
- Multiple Sensor Use
- Sensor Mode/Parameter Selection
- Cooperation With Principal Investigator
- Target Selection

Development Engineer

- Sensor Operation
- Sensor Evaluation
- Component Testing

Technical Operations Specialist

- Equipment Setup, Checkout, Maintenance, Calibration
- Servicing of Sensor and Equipment Consumables

CONCERNS OF MAN IN ORBIT

Safety of Flight

- External Environment
- Physiological Limits
- Psychological Stress
- Onboard Safety

Performance Degradation

- Acceleration Disturbances
- Effluent Release
- Repetitive Duty Cycles

EVALUATION OF MAN IN-ORBIT INFLUENCES

| | | | | ENVIRONMENTAL | | | | | LIFE SCIENCES | | | MPS | | | | | | |
|--------------------|---------------------------|----------------------------------------------------------|--|---------------|-----|-------|-----|--------------|---------------|-----|------|-----|-----|------|-----|-----|-----|--------|
| | | | | ICE | MET | OCEAN | SPP | ZERO-g CLOUD | LARS | LPF | SBRF | BRF | OAF | EMTF | MEA | MEC | MRF | VACUUM |
| BENEFICIAL | SCIENTIST OBSERVER | REAL-TIME DATA ANALYSIS | | ○ | ● | ○ | ○ | ○ | ○ | ● | ● | ● | ● | ● | ○ | ○ | ● | |
| | DEVEL ENGR | MULTIPLE SENSOR USE | | ● | ● | ○ | ○ | ● | ○ | ● | ● | ● | ● | ● | ● | ● | ● | |
| | TECH OPS | SENSOR MODE/PARAMETER SELECTION | | ○ | ● | ○ | ○ | ● | ○ | ● | ● | ● | ● | ● | ● | ● | ● | |
| | SAFETY OF FLIGHT | COOPERATION WITH PRINCIPAL INVESTIGATOR | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | |
| | PERF DEGRAD | TARGET SELECTION | | ● | ● | ● | ○ | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | |
| | FLIGHT | SENSOR OPERATION | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | |
| | OPS | SENSOR EVALUATION | | ● | ○ | ○ | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | |
| | DEGRAD | COMPONENT TESTING | | ○ | ○ | ○ | ○ | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | |
| | OF FLIGHT | EQUIPMENT SETUP, CHECKOUT, MAINTENANCE, CALIBRATION, ETC | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | |
| | OPS | SERVICING OF SENSOR AND EQUIPMENT CONSUMABLES | | ○ | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | |
| DETERRIMENTAL | EXTERNAL ENVIRONMENT | | | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | |
| | PHYSIOLOGICAL LIMITS | | | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | |
| | PSYCHOLOGICAL STRESS | | | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | |
| | ONBOARD SAFETY | | | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ? | ? | ○ | ○ | |
| | ACCELERATION DISTURBANCES | | | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | |
| PERF DEGRAD | EFFLUENT RELEASE | | | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ? | ? | ○ | ○ | |
| | REPETITIVE DUTY CYCLES | | | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | |
| | SPACE STATION CANDIDATE | | | ○ | ○ | ○ | ● | ● | ● | ● | ● | ● | ● | ● | ○ | ● | ● | |
| PLATFORM CANDIDATE | | | | ○ | ● | ○ | ● | ○ | ○ | ● | ● | ● | ● | ● | ○ | ● | | |

Required
Desirable
Acceptable
Intolerable
Effects Unknown

Space Station Candidate

Platform Candidate

D18

EVALUATION OF MAN IN-ORBIT INFLUENCES

VFX878

| | | | ASTROPHYSICS | | | | | | | | | | | | | | |
|------------|-----------------------|----------------------------------------------------------|---------------------------|-------|----------|------|------------|-----|---------------|-----|-----|-----|------|-------|------|-----|---|
| | | | SOT | SIRTF | STAR LAB | SCRN | SOFT X-RAY | STO | PINHOLE X-RAY | XRO | HRS | XTE | AXAF | LAMAR | VLBI | ASO | |
| BENEFICIAL | SCIENTIST OBSERVER | REAL-TIME DATA ANALYSIS | ● | ○ | ○ | ○ | ○ | ● | ○ | ● | ○ | ○ | ○ | ○ | ○ | ● | |
| | DEVEL ENGR | MULTIPLE SENSOR USE | ○ | ○ | ● | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | |
| | TECHNICIAN | SENSOR MODE/PARAMETER SELECTION | ● | ● | ● | ○ | ● | ● | ● | ○ | ○ | ○ | ○ | ○ | ○ | ○ | |
| | PERF DEGRAD OF FLIGHT | COOPERATION WITH PRINCIPAL INVESTIGATOR | ● | ● | ● | ○ | ● | ● | ● | ○ | ● | ○ | ○ | ● | ○ | ● | |
| | | TARGET SELECTION | ● | ● | ● | ○ | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | |
| | | SENSOR OPERATION | ● | ● | ● | ○ | ○ | ○ | ○ | ○ | ○ | ● | ● | ○ | ○ | ○ | |
| | | SENSOR EVALUATION | ● | ● | ● | ○ | ● | ● | ● | ○ | ○ | ○ | ○ | ○ | ○ | ○ | |
| | | COMPONENT TESTING | ● | ● | ● | ○ | ● | ● | ● | ○ | ○ | ○ | ○ | ○ | ○ | ○ | |
| | | EQUIPMENT SETUP, CHECKOUT, MAINTENANCE, CALIBRATION, ETC | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | |
| | | SERVICING OF SENSOR AND EQUIPMENT CONSUMABLES | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | |
| | | | EXTERNAL ENVIRONMENT | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ |
| | | | PHYSIOLOGICAL LIMITS | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ |
| | | | PSYCHOLOGICAL STRESS | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ |
| | | | ONBOARD SAFETY | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ |
| | | | ACCELERATION DISTURBANCES | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ? | ○ | ○ | ○ |
| | | | EFFLUENT RELEASE | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ |
| | | | REPETITIVE DUTY CYCLES | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ |
| | | | SPACE STATION CANDIDATE | ● | ● | ● | ● | ● | ● | ● | ○ | ○ | | | | | |
| | | | PLATFORM CANDIDATE | ○ | | | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ | ○ |

EVALUATION OF MAN IN-ORBIT INFLUENCES

VFX879

| | | | COMM | | | EARTH/PLANETARY EXPERIMENTS | | | | | | | | | |
|-----------------------------|---------------------------|---------------|-------------------------------------------------------------|-----|-----|-----------------------------|------|-----|-----|-----|----|-----|-----|--|--|
| | | | RFI | OSP | CRF | SAR | ALOS | MFM | SMA | LFC | IS | FLD | ATM | | |
| BENEFICIAL | SCIENTIST OBSERVER | DEVEL ENGR | REAL-TIME DATA ANALYSIS | | | | | | | | | | | | |
| | | | O | ● | ● | O | O | O | O | O | O | O | O | | |
| | | | O | ● | ● | O | O | O | O | O | O | ● | ● | | |
| | | | ● | ● | ● | ● | ● | O | O | ● | O | ● | ● | | |
| | | | ● | ● | ● | O | O | ● | ● | ● | ● | ● | ● | | |
| | | | O | O | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
| | | | ● | ● | ● | O | O | ● | ● | ● | ● | ● | ● | | |
| | | | O | O | ● | O | O | O | O | O | O | O | O | | |
| | | | O | O | ● | O | O | O | O | O | O | O | O | | |
| | | | ● | O | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
| DETERRIMENTAL | TECH- NICKIAN | DEVEL ENGR | SENSOR OPERATION | | | | | | | | | | | | |
| | | | ● | ● | ● | O | O | ● | ● | ● | ● | ● | ● | | |
| | | | O | O | ● | O | O | ● | ○ | O | O | ● | ● | | |
| | | | O | O | ● | O | O | O | O | O | O | O | O | | |
| | | | O | O | ● | O | O | O | O | O | O | O | O | | |
| | | | ● | O | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
| | | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| PERF DEGRAD OF FLIGHT | SAFETY OF FLIGHT | DEVEL ENGR | SENSOR EVALUATION | | | | | | | | | | | | |
| | | | O | O | ● | O | O | ● | ○ | O | O | ● | ● | | |
| | | | O | O | ● | O | O | ● | ○ | O | O | ● | ● | | |
| | | | O | O | ● | O | O | ○ | O | O | O | O | O | | |
| | | | O | O | ● | O | O | O | O | O | O | O | O | | |
| | | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
| | | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
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| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| PERF DEGRAD OF FLIGHT | EXTERNAL ENVIRONMENT | DEVEL ENGR | COMPONENT TESTING | | | | | | | | | | | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
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| | | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
| | | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| PERF DEGRAD OF FLIGHT | PSYCHOLOGICAL STRESS | DEVEL ENGR | EQUIPMENT SETUP, CHECKOUT, MAINTENANCE, CALIBRATION, ETC | | | | | | | | | | | | |
| | | | ● | O | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
| | | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
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| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| PERF DEGRAD OF FLIGHT | ONBOARD SAFETY | DEVEL ENGR | SERVICING OF SENSOR AND EQUIPMENT CONSUMABLES | | | | | | | | | | | | |
| | | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
| | | | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
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| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| PERF DEGRAD OF FLIGHT | ACCELERATION DISTURBANCES | DEVEL ENGR | EXTERNAL ENVIRONMENT | | | | | | | | | | | | |
| | | | O | O | O | O | O | ? | ? | O | O | O | O | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
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| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| PERF DEGRAD OF FLIGHT | EFFLUENT RELEASE | DEVEL ENGR | PHYSIOLOGICAL LIMITS | | | | | | | | | | | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
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| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| PERF DEGRAD OF FLIGHT | REPETITIVE DUTY CYCLES | DEVEL ENGR | PSYCHOLOGICAL STRESS | | | | | | | | | | | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
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| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| PERF DEGRAD OF FLIGHT | SPACE STATION CANDIDATE | DEVEL ENGR | ONBOARD SAFETY | | | | | | | | | | | | |
| | | | O | ● | ● | ● | ● | ● | ● | ● | ● | ● | ● | | |
| | | | O | O | O | O | O | O | O | O | O | O | O | | |
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| | | | O | O | O | O | O | O | O | O | O | O | O | | |
| PERF DEGRAD OF FLIGHT | PLATFORM CANDIDATE | DEVEL ENGR | ACCELERATION DISTURBANCES | | | | | | | | | | | | |
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EVALUATION OF MAN IN-ORBIT INFLUENCES

VFX880

| | | | | ENVIRONMENTAL | | | | | LIFE SCIENCES | | | MPS | | | | | | |
|-------------------------|-------------------------|----------------------------------------------------------|---|---------------|-----|-------|-----|--------------|---------------|-----|------|-----|-----|------|-----|-----|-----|--------|
| | | | | ICE | MET | OCEAN | SPP | ZERO G CLOUD | LARS | LPF | SBRF | BRF | OQF | EMTF | MEA | MEC | MRF | VACUUM |
| BENEFICIAL | SCIENTIST OBSERVER | REAL-TIME DATA ANALYSIS | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| BENEFICIAL | DEVEL ENGR | MULTIPLE SENSOR USE | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| BENEFICIAL | TECH-NICIAN | SENSOR MODE/PARAMETER SELECTION | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| BENEFICIAL | DEVEL ENGR | COOPERATION WITH PRINCIPAL INVESTIGATOR | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| BENEFICIAL | TECH-NICIAN | TARGET SELECTION | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| DETRIMENTAL | SAFETY DEGRAD OF FLIGHT | SENSOR OPERATION | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| DETRIMENTAL | SAFETY DEGRAD OF FLIGHT | SENSOR EVALUATION | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| DETRIMENTAL | SAFETY DEGRAD OF FLIGHT | COMPONENT TESTING | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| DETRIMENTAL | PERF DEGRAD | EQUIPMENT SETUP, CHECKOUT, MAINTENANCE, CALIBRATION, ETC | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| DETRIMENTAL | PERF DEGRAD | SERVICING OF SENSOR AND EQUIPMENT CONSUMABLES | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| DETMINAL | SAFETY DEGRAD OF FLIGHT | EXTERNAL ENVIRONMENT | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| DETMINAL | SAFETY DEGRAD OF FLIGHT | PHYSIOLOGICAL LIMITS | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| DETMINAL | SAFETY DEGRAD OF FLIGHT | PSYCHOLOGICAL STRESS | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| DETMINAL | SAFETY DEGRAD OF FLIGHT | ONBOARD SAFETY | O | O | O | O | O | O | O | O | O | O | ? | ? | O | O | O | |
| DETMINAL | PERF DEGRAD | ACCELERATION DISTURBANCES | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| DETMINAL | PERF DEGRAD | EFFLUENT RELEASE | O | ? | O | O | O | O | O | O | O | O | ? | ? | O | O | O | |
| DETMINAL | PERF DEGRAD | REPETITIVE DUTY CYCLES | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| SPACE STATION CANDIDATE | | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |
| PLATFORM CANDIDATE | | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | O | |

IMPACT OF MANNED PRESENCE ON SCIENCE AND APPLICATIONS

VFX868

Pros

- **Responds Creatively As Unanticipated Events or Problems Arise**
- **Contributes to Assembly, Maintenance, Repair**
- **Allows Schedule Compression — Reduced Cost and Risk**
- **Has Unique Perceptual Abilities**
- **Contribution is Historic Fact**

Cons

- **Creates Disturbances for Fine Pointing**
- **Has Physiological and Psychological Performance Limits**
- **Manned Life Support Systems Can Reduce Viewing Sensitivities**

SERVICE MISSIONS IDENTIFIED SCIENCE AND APPLICATIONS

| Missions | Mass (kg) | Altitude (km) | Inclination (deg) | Service Requirements |
|-----------------------|-----------|---------------|-------------------|---------------------------------------------|
| Space Telescope | 11,000 | 600 | 28.8 | Repair, 5-year Refurbishment |
| SIDM | 2,600 | 575 | 28/98 | Propellant Resupply, Refurbishment |
| LDR | 20,500 | 700 | 28 | Cryogen Resupply, Refurbishment, Reboost |
| Gamma Ray Observatory | 11,000 | 400 | 28.5 | Repair, Propellant Resupply, Refurbishment |
| LANDSAT D-D' | 1,600 | 705 | 97 | Repair, Refurbishment |
| GRAVSAT-B | | | | Cryogen Resupply, Refurbishment, Reboost |
| UARS | 2,400 | 500 | 56, 70 | Cryogen Resupply, Repair, Refurbishment |
| TIROS-N | 740 | 830 | 90 | Repair, Refurbishment |

SCIENCE AND APPLICATIONS MISSION ALLOCATION STATUS

VFX869

ALLOCATED TO SPACE STATION — 15

- | | | | |
|--------|-----------|----------------|-----------------|
| ■ SOT | ■ STARLAB | ■ SCRN | ■ Pinhole X-Ray |
| ■ OSP | ■ LFC | ■ Zero g Cloud | ■ LPF |
| ■ SBRF | ■ BRF | ■ OQF | ■ EMTF |
| ■ MRF | ■ Vacuum | ■ CRF | |

ALLOCATED TO SPACE STATION OR PLATFORM — 17

- | | | | |
|---------|--------------|-------|-------|
| ■ SIRTF | ■ Soft X-Ray | ■ STO | ■ XRO |
| ■ HRS | ■ ASO | ■ RFI | ■ SAR |
| ■ MFM | ■ SMA | ■ FLD | ■ ATM |
| ■ ICE | ■ Ocean | ■ SPP | ■ MEA |
| ■ MEC | | | |

ALLOCATED TO SPACE PLATFORM — 8

- | | | | |
|--------|--------|---------|--------|
| ■ XTE | ■ AXAF | ■ LAMAR | ■ VLBI |
| ■ ALOS | ■ IS | ■ MET | ■ LARS |

ALLOCATED TO FREE FLYER SERVICE — 8

- | | | | |
|-------------------|-------------|--------|---------|
| ■ Space Telescope | ■ SIDM | ■ LDR | ■ GRO |
| ■ LANDSAT D-D' | ■ GRAVSAT-B | ■ UARS | ■ TIROS |

D19

MISSION REQUIREMENTS (TASK 1)

CANDIDATE COMMERCIAL

MISSIONS

Dr. Harry Wolbers

COMMERCIAL USES OF SPACE PAST EXPERIENCE

VFY168

- Substantial Survey Work Since 1972
 - (e.g.: GE, TRW, SAI, RI, MDAC)
- Typical Results of Prior Studies
 - Few Concepts Stand Up Under Scrutiny
 - Attractive Alternatives - Less Expensive
 - Products and Markets Poorly Defined
 - Risks High - Many Unknowns
 - Long Time Delay - Concept to Implementation
 - Payback Period Long for Capital Outlay Required
 - Protection of Proprietary Rights Critical

E1

COMMERCIAL USES OF SPACE CURRENT STATUS

VFY169

- Only One Potential Product for Space Manufacturing Has Reached Testing Stage — (Electrophoresis Operations)
- Interest in Space Exists
- Companies Want to Stay Ahead of Competition
- Risk Reducing/Sharing Policies Would Spur Interest
- Continuing Dialog With Potential Users Required

E2

COMMERCIAL USERS INTERACTION STRATEGY

VFY139

PROCEDURE

- Develop Case Study As Example
- Establish Continuing Dialog and In-Depth Interviews With Selected Users

RESULTS TO DATE

- Relationships With 18 Corporate Entities Established (8 Booz Allen, 10 MDAC)
- 25 In-Depth Interviews (15 Booz Allen, 10 MDAC)
- 12 Potential Commercial Missions Identified
- 59 Products/Processes Identified to Date

E3

TWELVE POTENTIAL COMMERCIAL MISSIONS IDENTIFIED TO DATE

VFY006

| MDAC Data Bank Identifier | Areas of Responsibility | | |
|------------------------------------|-------------------------|-------------|-----|
| | MDAC HB | MDAC STL | BAH |
| CIR001 Materials Research Facility | ● | ○ | |
| CMP001 Electrophoretic Processes | | ● | |
| CMP002 Silicon Ribbon Manufacture | | ● | |
| CMP003 Crystals/Diffractors | ○ | | ● |
| CMP004 Melting/Refreezing | ○ | | ● |
| CMP005 Homogeneous Mixtures | ○ | | ● |
| CMP006 Directional Crystal Growth | ○ | | ● |
| CMP007 Hot/Cold Processes | ○ | | ● |
| CMP008 Unidirectional Processes | ○ | | ● |
| CMP009 Earth Observations | ○ | | ● |
| CMP010 Materials Production | ○ | | ● |
| CMP011 Misc Operations | ○ | | ● |

- Product Definition
- System Support

E4

MISSION REQUIREMENTS (TASK 1)

SELECTED COMMERCIAL

MISSIONS

Jim Rose — MDAC St. Louis

CASE HISTORY OF A COMMERCIAL SPACE MISSION

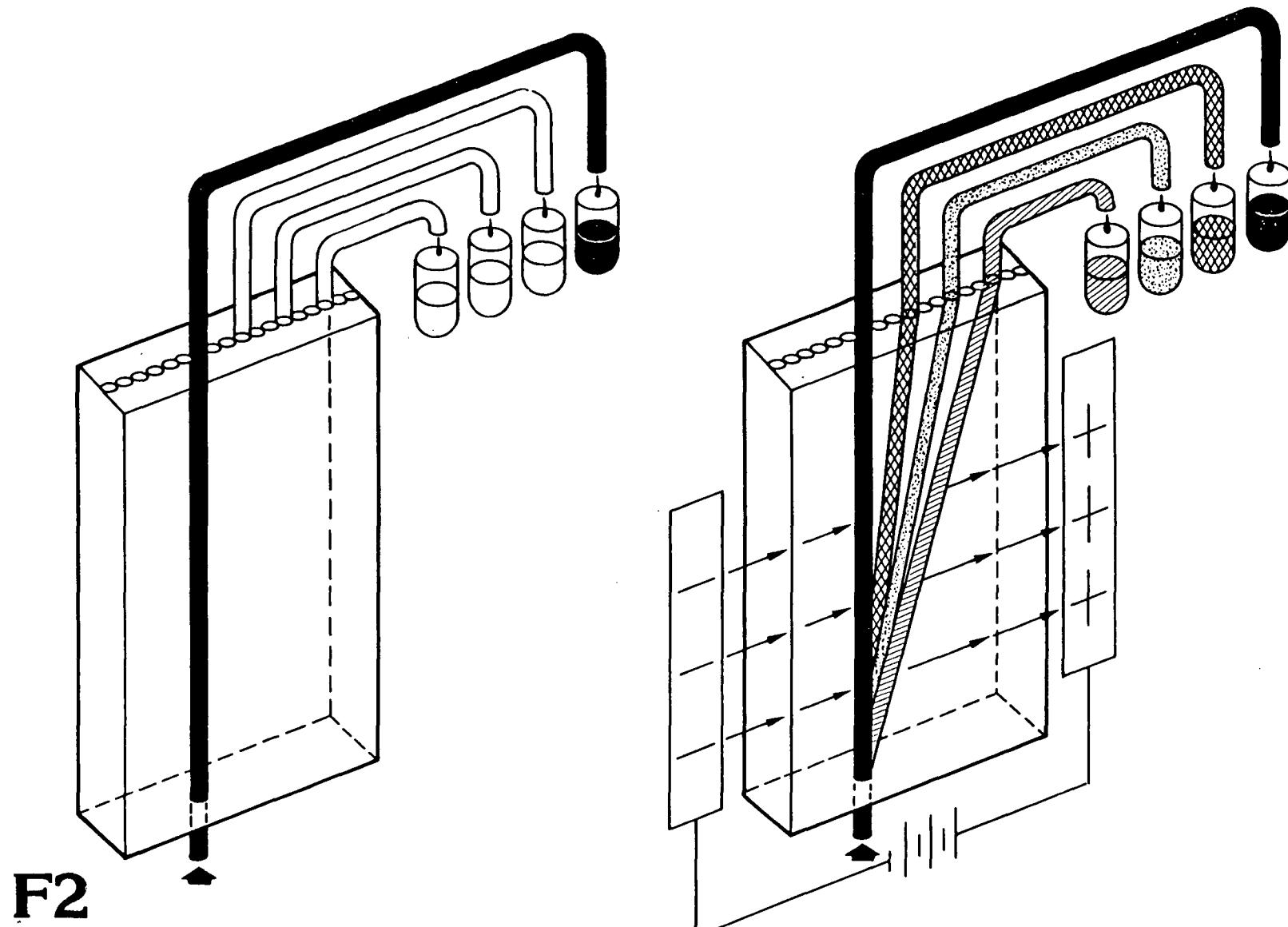
VFY047



- Potential Benefits
- Program Characteristics
- Development Considerations
- Manned Space Station Operations

VFY048

CONTINUOUS FLOW ELECTROPHORESIS



POSSIBLE PRODUCTS UTILIZING EOS TECHNOLOGY

VFY049

| Field | Types of Products |
|------------------------|--------------------------------------------------------------------|
| Pharmaceuticals | Enzymes, Hormones, Other Proteins, Cells |
| Diagnostics | Monoclonal Antibodies, Hormones for Radioimmunoassays |
| Veterinary | Enzymes, Hormones, Other Proteins, Spermatozoa, Other Cells |
| Agrichemicals | Growth Stimulants, Pathogens |
| Food Products | Additives |

F3

EXAMPLES OF BENEFICIAL BIOLOGICAL PRODUCTS

VFY050

| Product | Product Objective | Current Status |
|--------------------------------------------|----------------------------------------------------------------|---------------------------------------|
| Growth Hormone (850,000)* | Stimulates Juvenile Bone Growth, Promotes Healing of Ulcers | Research Quantities, Low Purity |
| Beta Cells (3,200,000) | Single Injection Cure for Diabetes | Clinical Quantities Not Separable |
| α - Antitrypsin (500,000)* | Limit Emphysema Disease State, Enhance Cancer Chemotherapy | Research Quantities, Low Purity |
| Epidermal Growth Factor (1,100,000)* | Skin Burn and Wound Healing | Research Quantities, Low Purity |
| Interferon (20,000,000)* | Viral Infection Immunity | Low Yield and Purity |
| Antihemophilic Factor (15,000)* | Eliminate Immunological Reactions for Hemophilia | Low Purity and Loss of By-Products |

*Annual Patient Load — U.S. Market

F4

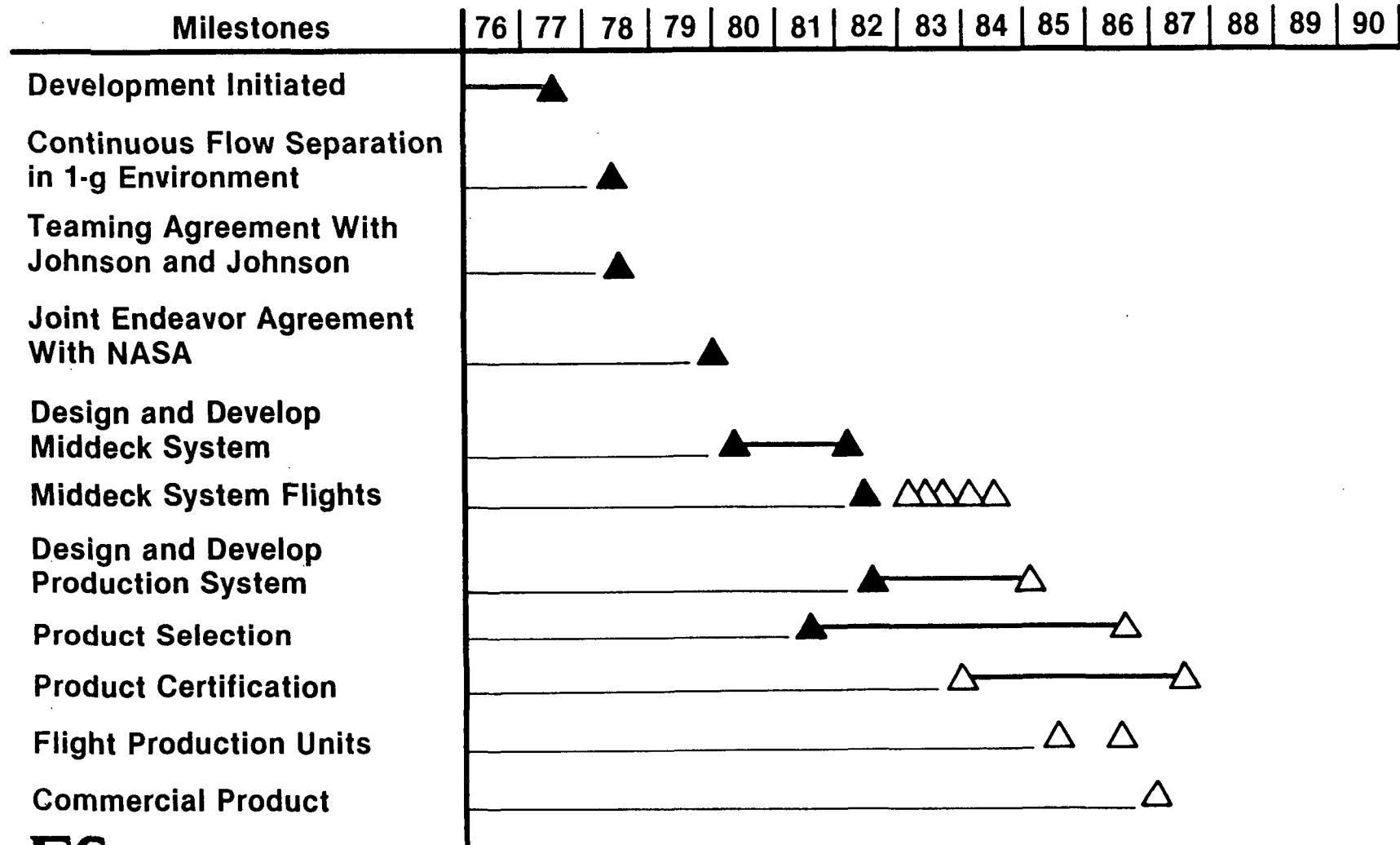
CHARACTERISTICS OF THE EOS PROGRAM

- Cooperative Venture
 - (1) Within Industry: MDAC for Aerospace, Johnson and Johnson (J&J) for Pharmaceuticals
 - (2) With Government: NASA for Shuttle Launch and Support
- High Technology Activity
 - (1) New Process Development Required
 - (2) New Product Development Required
- Proprietary Marketable Products
 - (1) Impressive Medical Benefits Possible
 - (2) Low Weight/High Value
 - (3) Projected Positive Return on Investment (ROI)
- Successful Early Development Effort

F5

EVOLUTION OF A COMMERCIAL SPACE PROGRAM

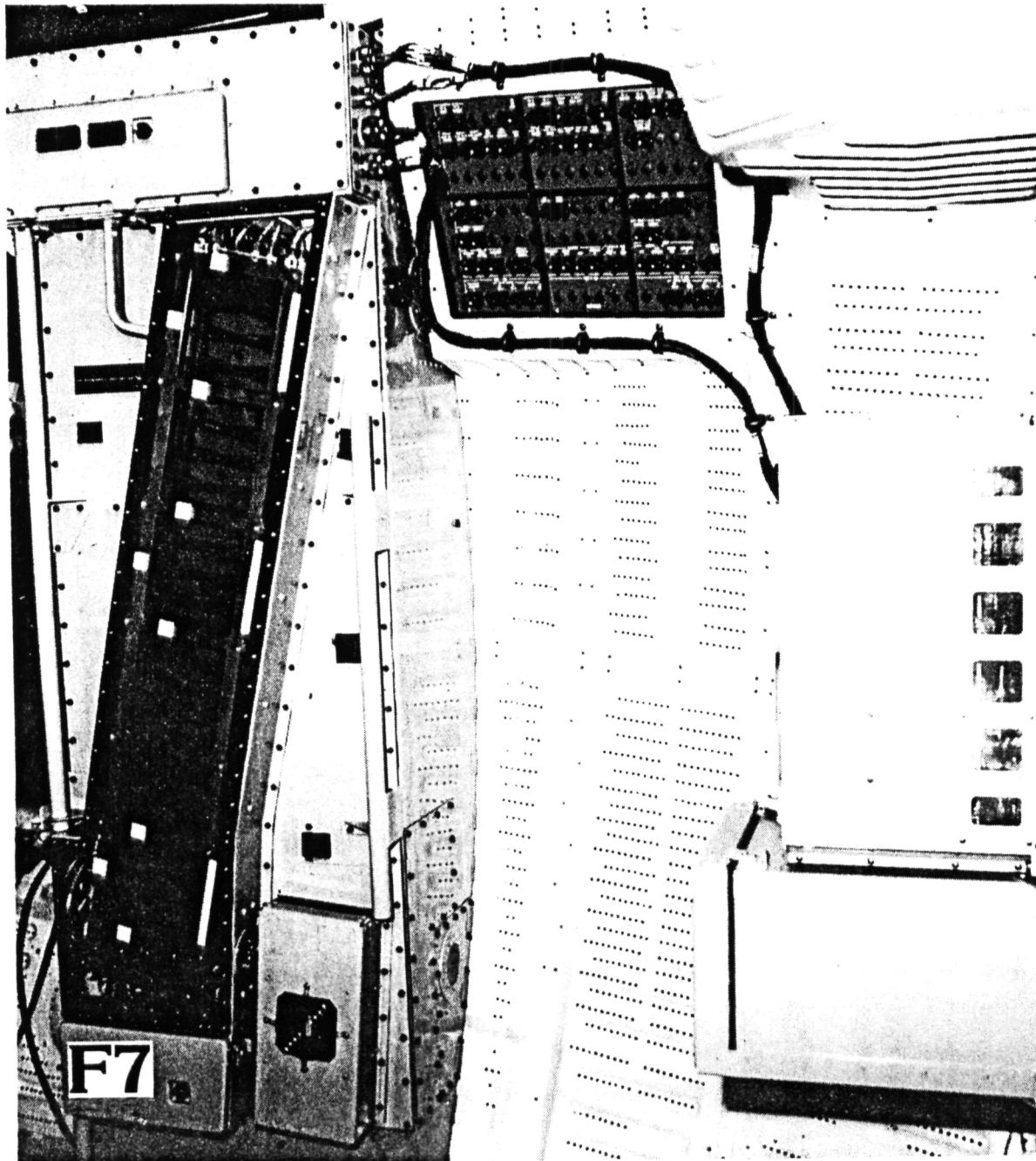
VFY052



F6

EOS MIDDECK SYSTEM

VFY053



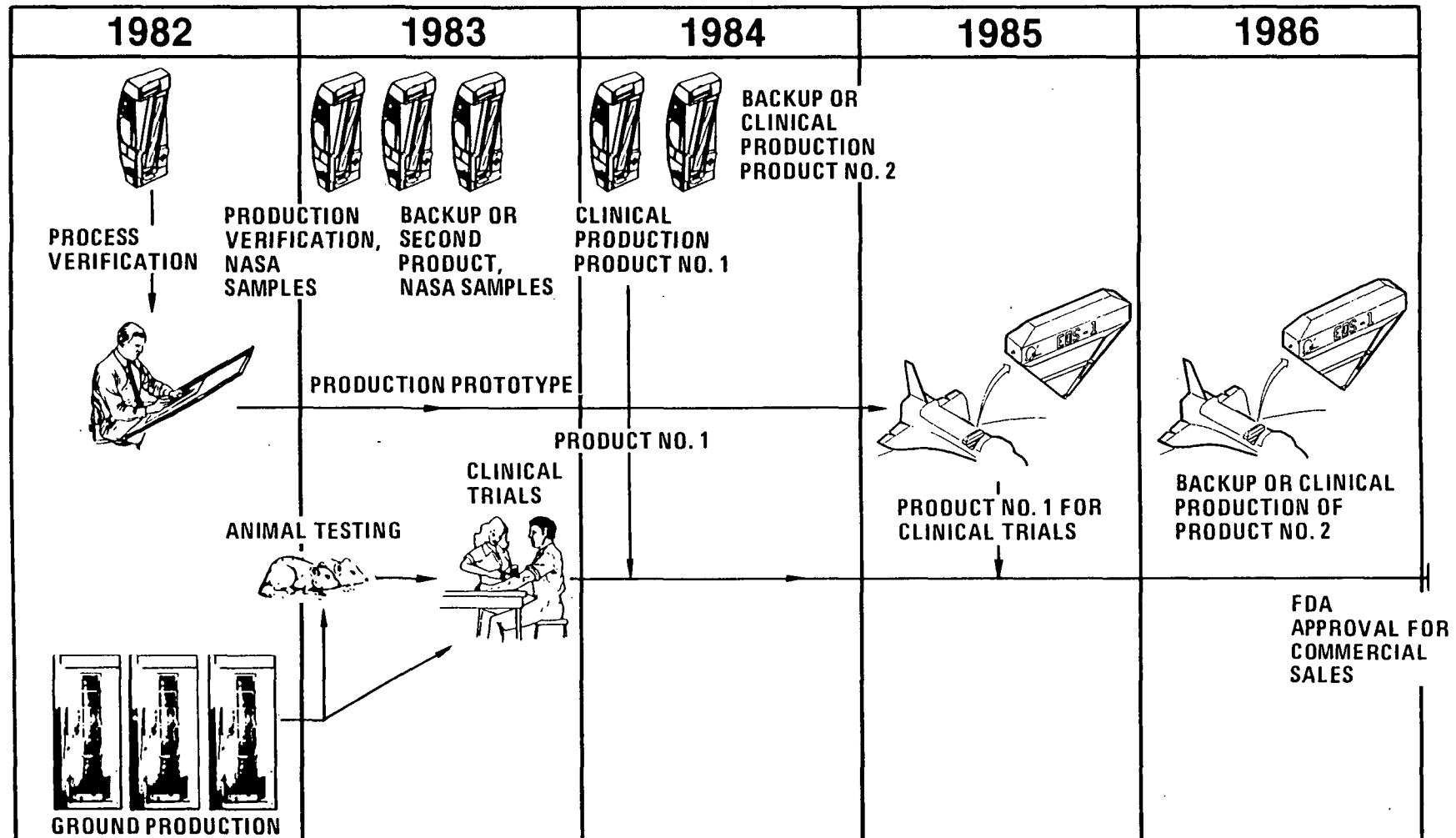
Flight Dates Under Joint Endeavor Agreement

| | | | |
|-----|----|-------|------|
| STS | 4 | July | 1982 |
| | 6 | Jan | 1983 |
| | 7 | April | 1983 |
| | 8 | July | 1983 |
| | 11 | Jan | 1984 |
| | 14 | May | 1984 |

Results From First STS Flight

1. 500 Times Increase in Yield
2. Quantitatively Repeatable Separation
3. Validated Design Concepts

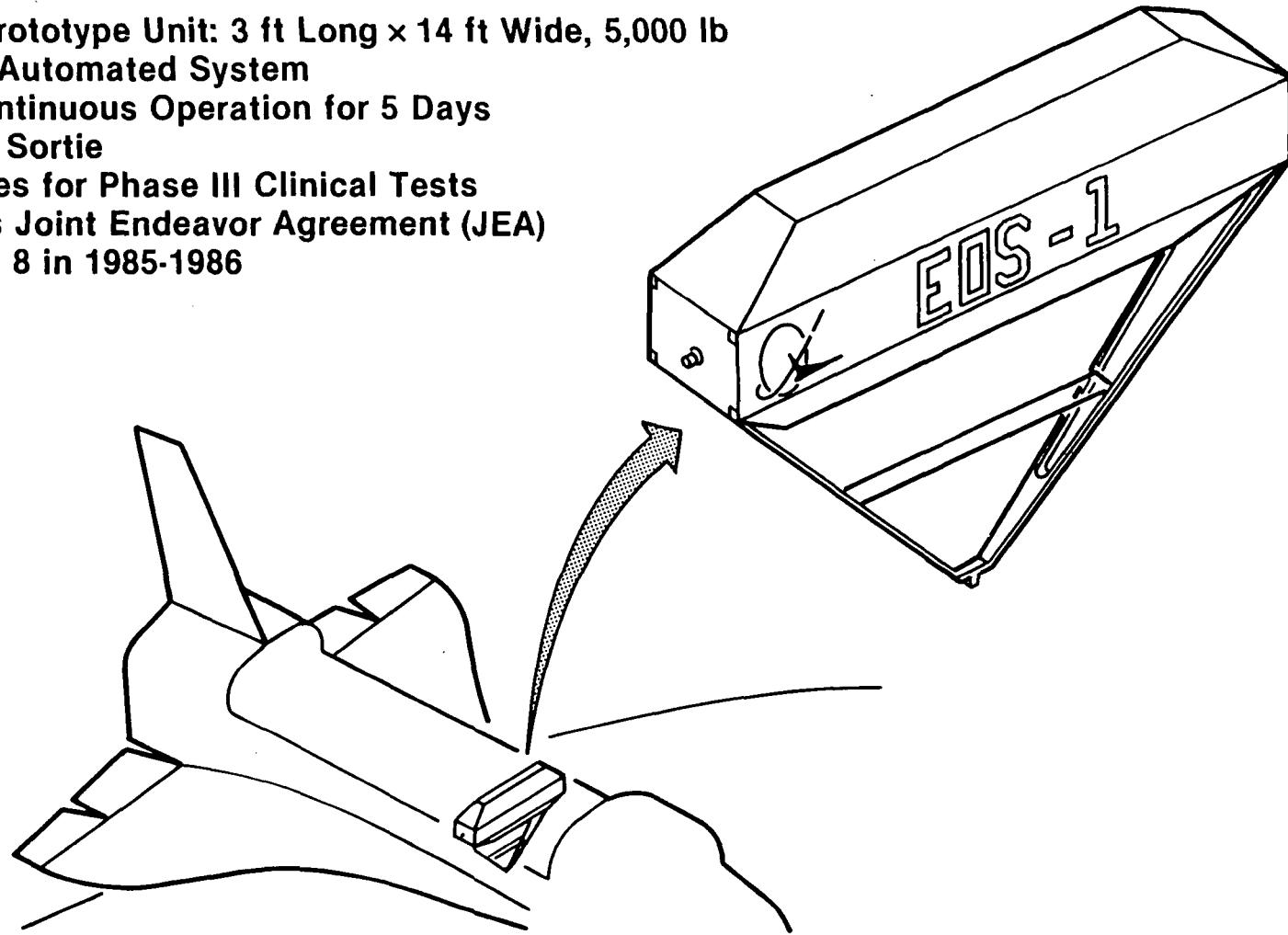
STEPS TO COMMERCIAL OPERATIONS



F8

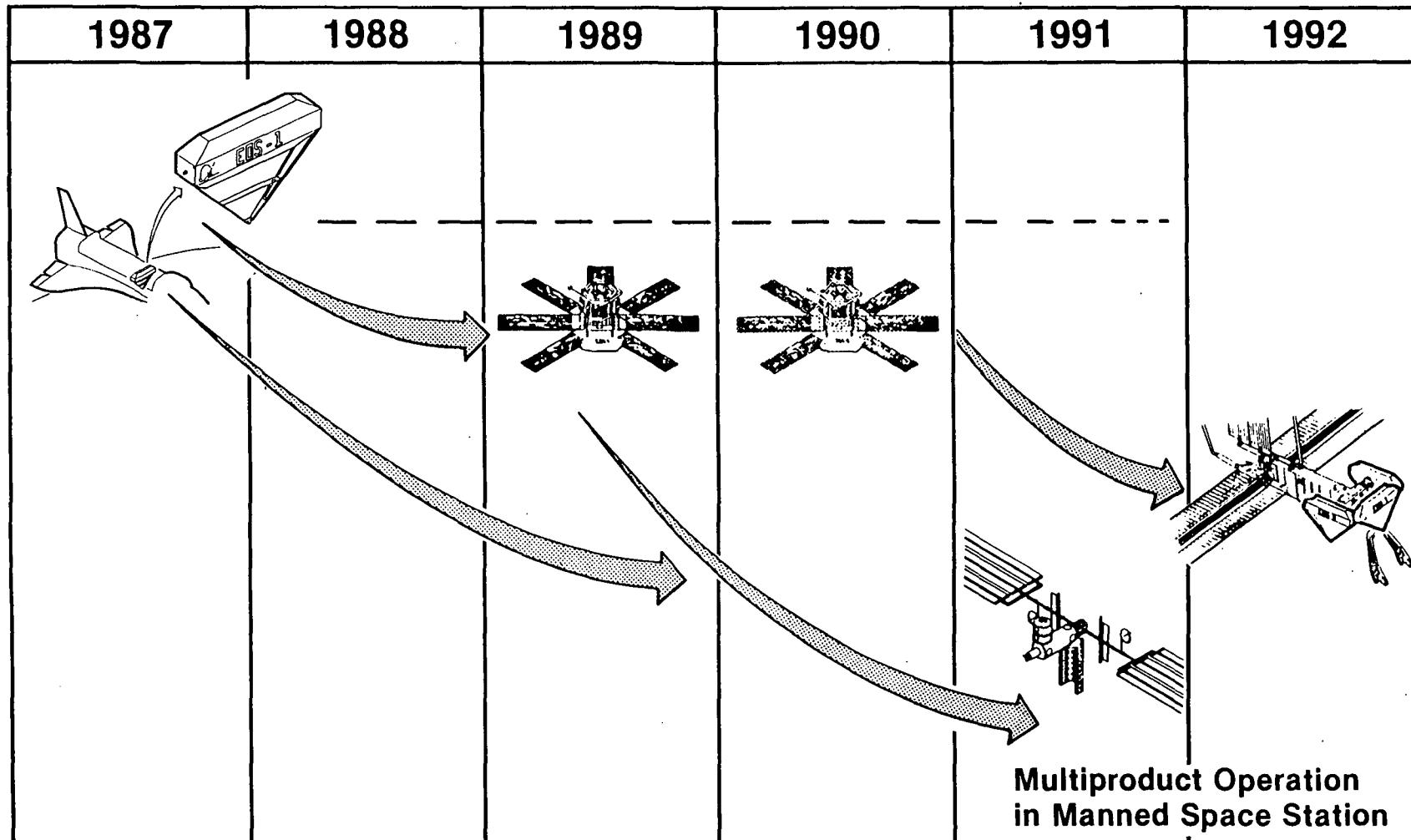
PRODUCTION PROTOTYPE IN SHUTTLE PAYLOAD BAY

- Production Prototype Unit: 3 ft Long × 14 ft Wide, 5,000 lb
- 24-Chamber, Automated System
- Checkout Continuous Operation for 5 Days During 7-Day Sortie
- Produce Doses for Phase III Clinical Tests
- Scheduled as Joint Endeavor Agreement (JEA) Flights 7 and 8 in 1985-1986



F9

COMMERCIAL OPERATIONS ALTERNATIVES



F10

CONSIDERATIONS IN DEVELOPMENT OF NEW COMMERCIAL SPACE PRODUCTS

VFY057

- Requires Verification and Development of Space Processes and Products
- Significant Investment Capital at High Risk
- Unproven Markets for New Products
- Elapsed Time to Marketable Product Is Significant
- Vulnerable Proprietary Rights
- Dependence on Supporting Government Space Facilities
- Technology Obsolescence

F11

STEPS FOR NEW BIOLOGICAL PRODUCT DEVELOPMENT

VFY058

| | Unmanned Free-Flyer Mode | | | Manned Space Station Mode |
|-------------------------------------|--------------------------|----------------|------------------------|---------------------------------|
| | Middeck or Spacelab | Payload Bay | Unmanned Free-Flyer | |
| Characterization | ✓ | | | ✓ |
| Clinical Trial Materials | | ✓ | | ✓ |
| Initial Commercial Production | | ✓ (Interim) | ✓ | ✓ |
| Expanded Production | | | ✓ | ✓ |

F12

**MANNED SPACE STATION
SHOWS SIGNIFICANT IMPROVEMENT OVER
UNMANNED PLATFORM ON PRODUCT PRICE TO PATIENT**

VFY059

| | Unmanned Operations Mode | Manned Operations Mode |
|----------------------------------------------------------|-----------------------------------------|---------------------------------------|
| Relative Cost of Front-End Expense | 1 | 0.38 |
| Relative Cost of Operating Expense | 1 | 0.77 |
| Relative Number of New Products Developed | 1 | 5.0 |

F13

MANNED SPACE STATION OPERATION

Enhances Rate of New Product Additions

- Fifteen Products Can Be Added in 10 Years With Space Station Compared With Three Products for Unmanned Free-Flyer
- Product Characterization Time Is Reduced From 1 or 2 Years to a Few Months
- Production Time for Clinical Materials Is Reduced From 1 or 2 Years to a Few Months
- Dedicated Section of Plant and Manned Operation Allow Many Products to Be Evaluated or Produced in Short Runs in Same Time Frame

F14

MISSION REQUIREMENTS (TASK 1)

SELECTED COMMERCIAL

MISSIONS

Dr. Myron Weinberg — Booz, Allen and Hamilton

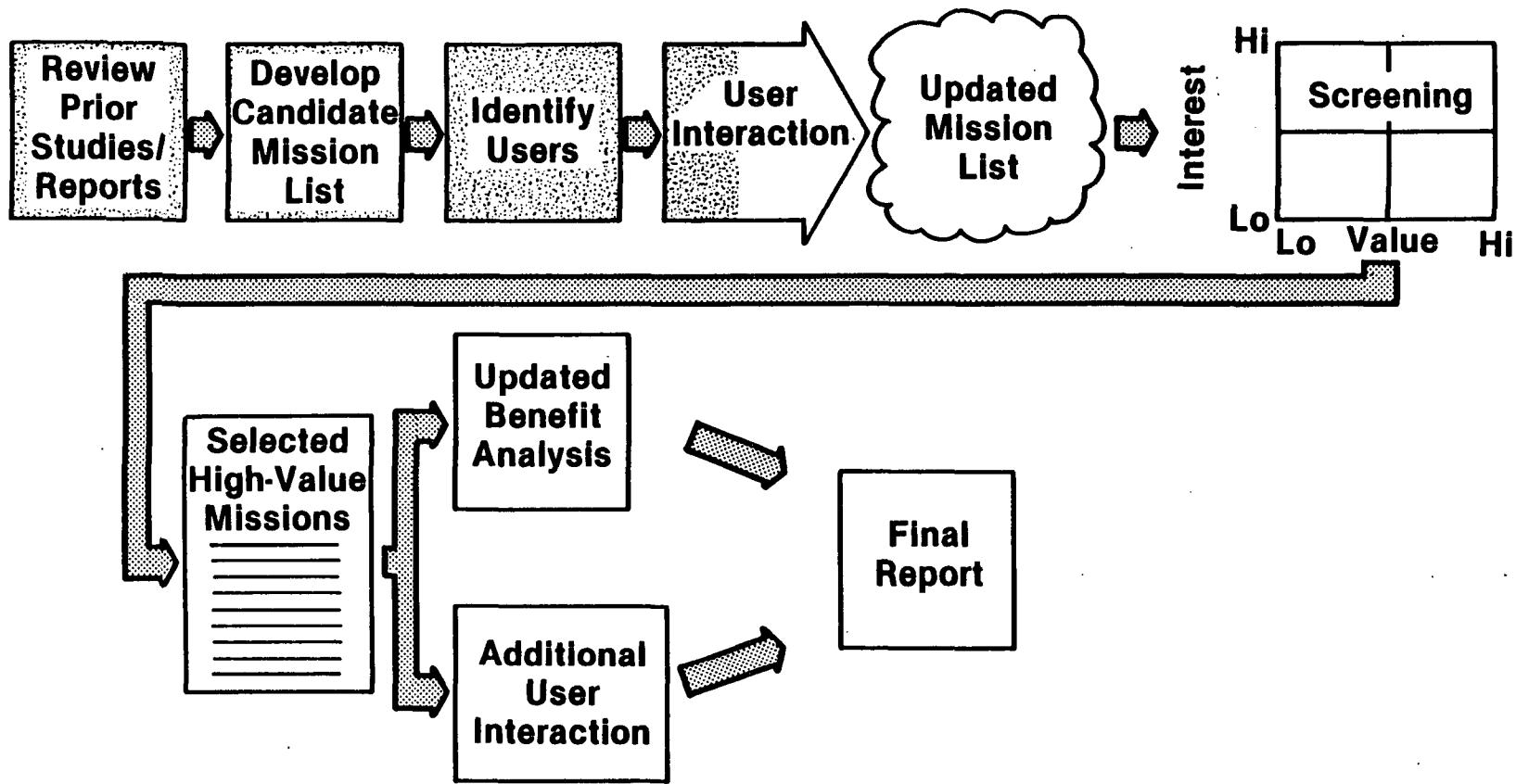
OBJECTIVES

- Identify Missions and Introduce Space Station Opportunities to Potential Commercial Users
- Develop “Real World” Qualified List of Commercial Missions
- Identify Broad Space Station Requirements for the Selected Missions

G1

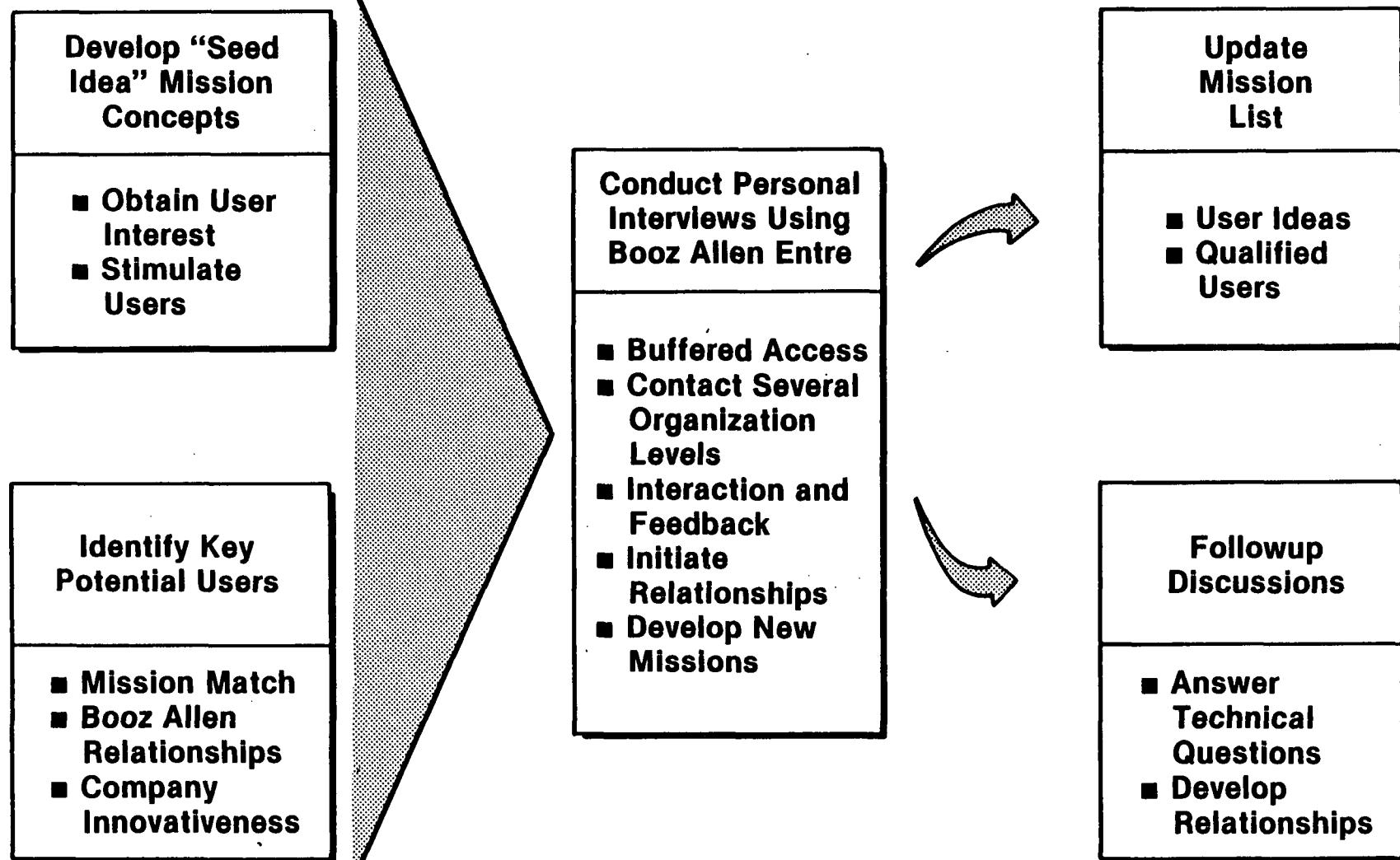
OVERALL APPROACH TO COMMERCIAL MISSION SELECTION AND ANALYSIS

VFX846



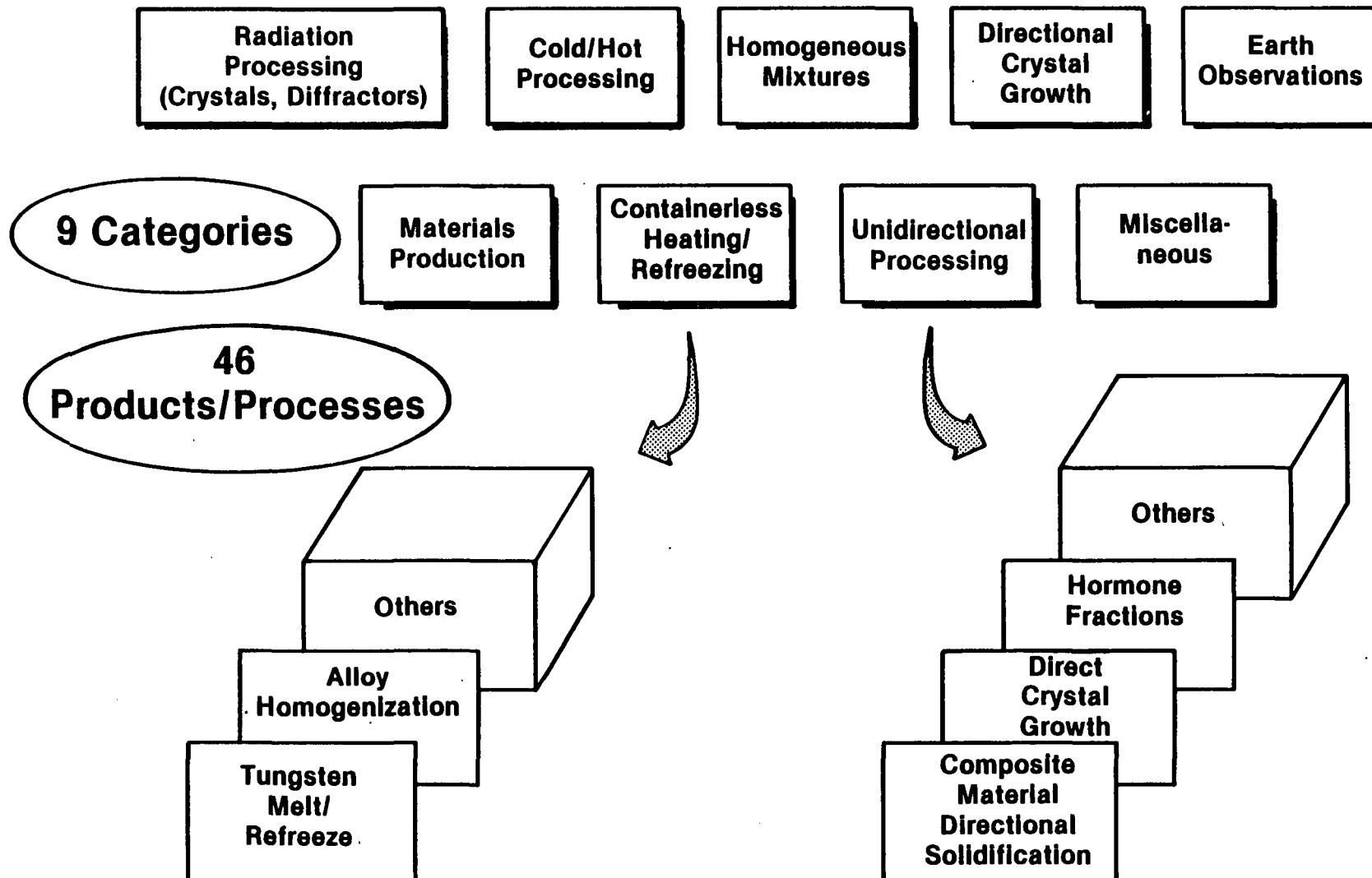
G2

KEY ELEMENTS OF BOOZ ALLEN APPROACH



G3

COMMERCIAL AREAS OF INTEREST



G4

CANDIDATE MISSION SUMMARY

| Category | Product/Process | File Reference |
|----------------------------------|------------------------------------------------------|-----------------|
| Unidirectional Processing | Composite Material Directional Solidification | A-MP-017 |
| | Directed Crystal Growth as in Silicons | B-005 |
| | Directed Quartz Crystal Growth | B-032 |
| | Other Crystal Growth | B-033 |
| | Orientation in Heterogeneous Composites | B-034 |
| | Protein Purification as in Immunoglobulins | B-036 |
| | Cellular or Protein Fractionation | B-035 |
| | Other Hormone Fractions | B-019 |

G5

46 CANDIDATE PRODUCTS/PROCESSES AND POTENTIAL USERS

VFX849

| Miscellaneous | | | | |
|----------------------------------|-----------------------------------------------|----------------|--------------------------|----------------------------------------------|
| Containerless Heating/Refreezing | | | | |
| Materials Production | | | | |
| Earth Observations | | | | |
| Directional Crystal Growth | | | | |
| Homogeneous Mixtures | | | | |
| Cold/Hot Processing | | | | |
| Radiation Processing | | | | |
| Unidirectional Processing | | | | |
| Category | Product/Process | File Reference | Potential Users | Remarks |
| UNIDIRECTIONAL PROCESSING | COMPOSITE MATERIAL DIRECTIONAL SOLIDIFICATION | A-MP-017 | | ABSENCE OF OXYGEN, DUST, OPERATION IN VACUUM |
| | DIRECTED CRYSTAL GROWTH AS IN SILICONS | B-005 | MONSANTO | |
| | DIRECTED QUARTZ CRYSTAL GROWTH | B-032 | U.S. TIME | |
| | OTHER CRYSTAL GROWTH | B-033 | | GLASS FIBERS ARE THE SIGNIFICANT TARGET |
| | ORIENTATION IN HETEROGENEOUS COMPOSITES | B-034 | BELL LABS | |
| | PROTEIN PURIFICATION AS IN IMMUNOGLOBULINS | B-036 | MONSANTO SCHERING PLOUGH | |
| | CELLULAR OR PROTEIN FRACTIONATION | B-035 | HYLAND | |
| | OTHER HORMONE FRACTIONS | B-019 | ELI LILLY | |

G7

CANDIDATE MISSION SUMMARY

| Category | Product/Process | File Reference | Potential Users |
|----------------------------------|---------------------------------------------------|----------------|-------------------------------------------|
| Unidirectional Processing | Composite Material | A-MP-017 | |
| | Directional Solidification | B-005 | Monsanto |
| | Directed Crystal Growth as in Silicones | B-032 | U.S. Time |
| | Directed Quartz Crystal Growth | B-033 | |
| | Other Crystal Growth | B-034 | Bell Labs |
| | Orientation in Heterogeneous Composites | B-036 | Monsanto Schering Plough |
| | Protein Purification as in Immunoglobulins | B-035 | Hyland |
| | Cellular or Protein Fractionation | | |
| | Other Hormone Fractions | B-019 | Eli Lilly |

G6

TARGET USERS 1

American Telephone and Telegraph Company

Homogenous Mixtures, Directional Crystal Growth

E. I. Dupont & Company

Homogenous Mixtures, Directional Crystal Growth, Undirectional Processing

Monsanto Company

Homogenous Mixtures, Directional Crystal Growth, Undirectional Processing

Allegheny International

Cold/Hot Processing, Homogenous Mixtures, Containerless Heating/Refreezing

Johnson Matthey Company

Cold/Hot Processing, Containerless Heating/Refreezing

Perkin Elmer, Inc.

Directional Crystal Growth, Radiation Processing

TARGET USERS 2

Celanese, Inc.

Homogenous Mixtures, Cold/Hot Processing

Eli Lilly Co., Inc.

Homogenous Mixtures, Unidirectional Processing

Union Carbide, Inc.

Earth Observations

The Fluor Corporation

Cold/Hot Processing, Homogenous Mixtures, Containerless Heating/Refreezing, Unidirectional Processing

International Business Machines

Directional Crystal Growth, Containerless Heating/Refreezing

Eastman Kodak

Directional Crystal Growth

Baxter Travenol

Unidirectional Processing, Homogenous Mixtures, Containerless Heating/Refreezing

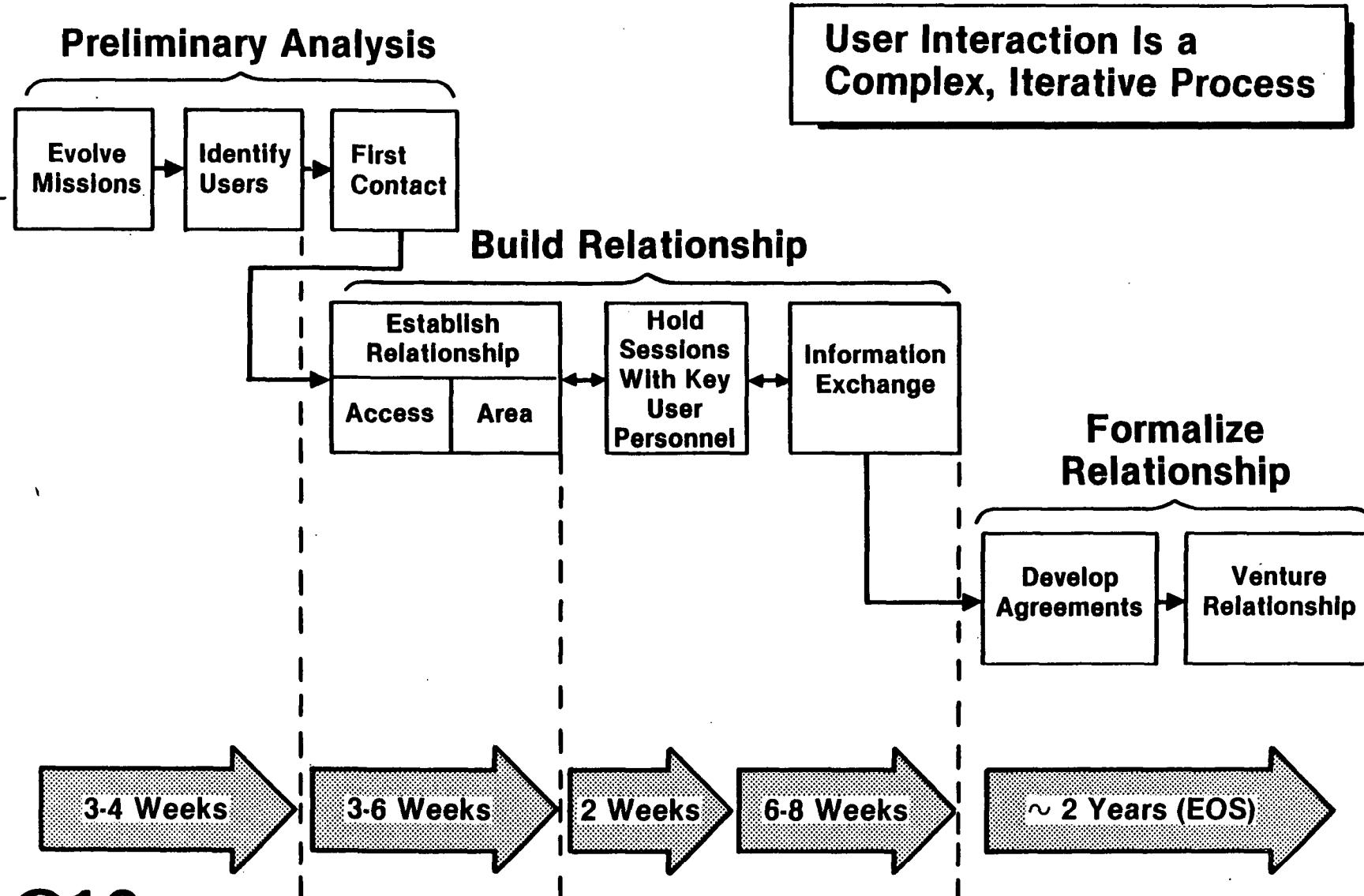
Department of Defense

Miscellaneous — Medical Uses

Environmental Protection Agency

Earth Observations

USER INTERACTION



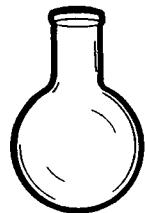
G10

USER INTERACTION RESULTS TO DATE

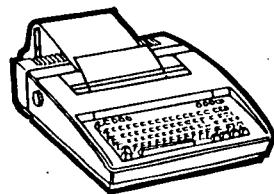
USER



American Telephone
and Telegraph Company



Monsanto



International
Business Machines



Baxter Travenol

RESPONSE

■ Interest in New Product Identified —
Up-Coming Concept Meeting

■ Three Major Areas of Processing
Identified — Planning Meeting to Be Set

■ Unique Area of Interest Identified —
Concept Meeting Week of
November 16, 1982

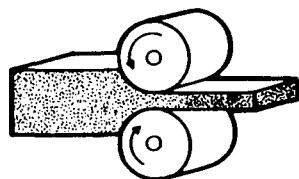
■ Three Major Areas of Interest
in New Products — Concept
Meeting on November 16, 1982

G11

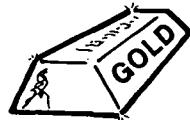
USER INTERACTION RESULTS TO DATE (CONT)

USER**Eli Lilly and Company****RESPONSE**

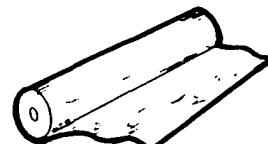
- Two Areas of Processing Interest Identified — Concept Meeting on November 17, 1982

**Allegheny International**

- Specific New Product Idea Identified by Allegheny — Meeting on Environmental Requirements Set for Week of December 1, 1982

**Johnson Matthey**

- Identified — Meeting Week of November 22, 1982

**Celanese**

- First Concept Meeting Completed — Technical Analysis of a New Concept Generated

G12

TYPICAL USER INTERACTION RESULTS (CELANESE MEETING)

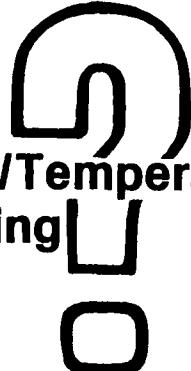
VFX857

Our Initial Meeting With Celanese Is an Example of the Results of Our Contacts to Date

- The Meeting Involved Discussion of Our Concepts — One Major New Idea Was Developed by Celanese**
- Several Detailed Technical Questions Were Asked**
- Technical Analysis of These Questions Is Required Before Followup Is Possible**
- Proprietary Rights Were Discussed**

CELANESE... TECHNICAL ANALYSIS

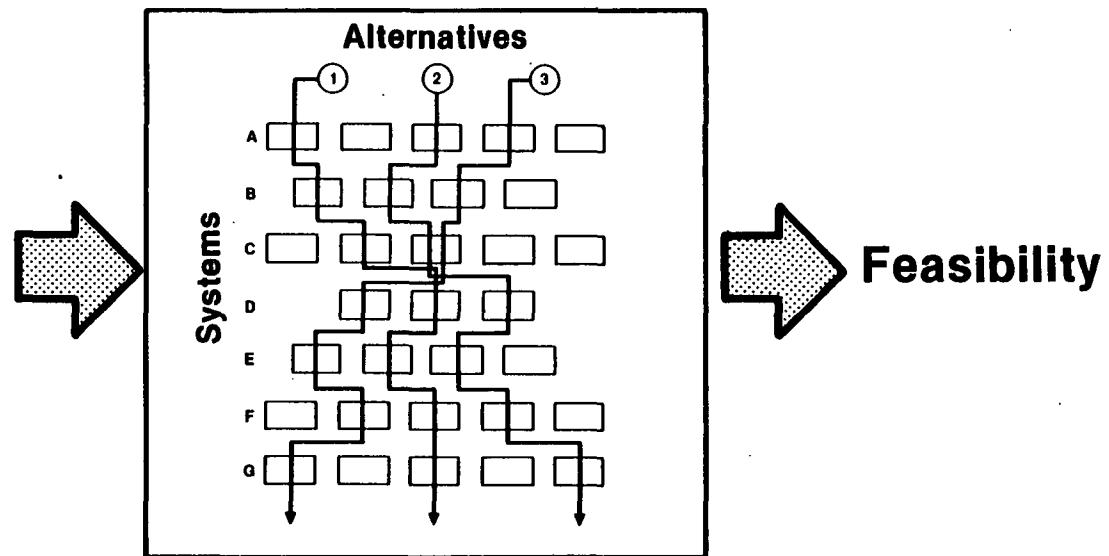
- Handling of Materials



- Time/Temperature/
Cooling



- Effects of Microgravity
on Stretched Molecules



G14

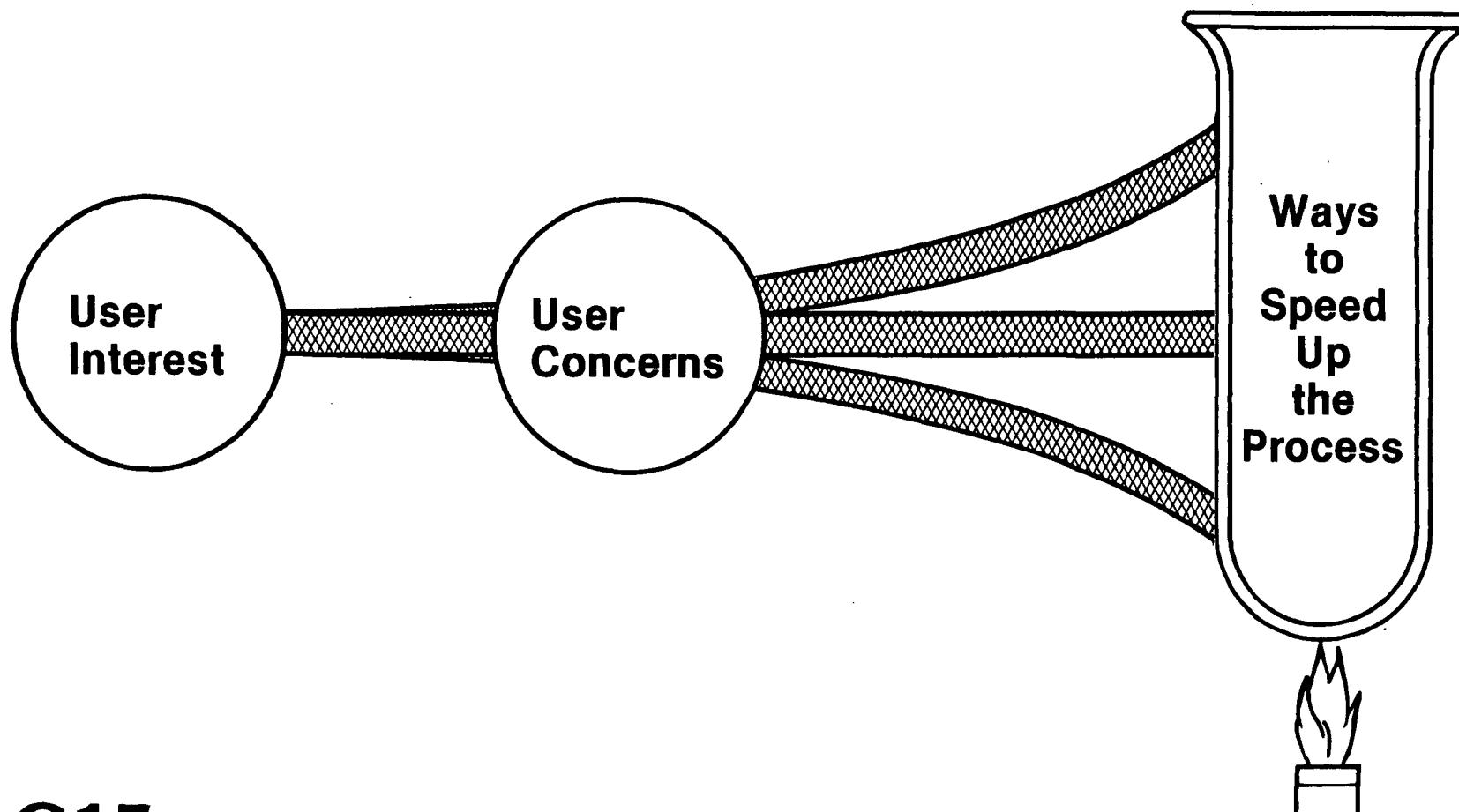
"WHAT WE HAVE LEARNED TO DATE"

We have completed two or more contacts with eight of the identified potential users and have drawn some general observations and conclusions from these discussions.

These observations may be summarized under three topic headings:

- Current user interest
- Current user concerns
- Ways to speed up the generation of user interest and, ultimately,
user-sponsored development

WHAT WE HAVE LEARNED TO DATE



G15

WHAT WE HAVE LEARNED TO DATE - USER INTEREST

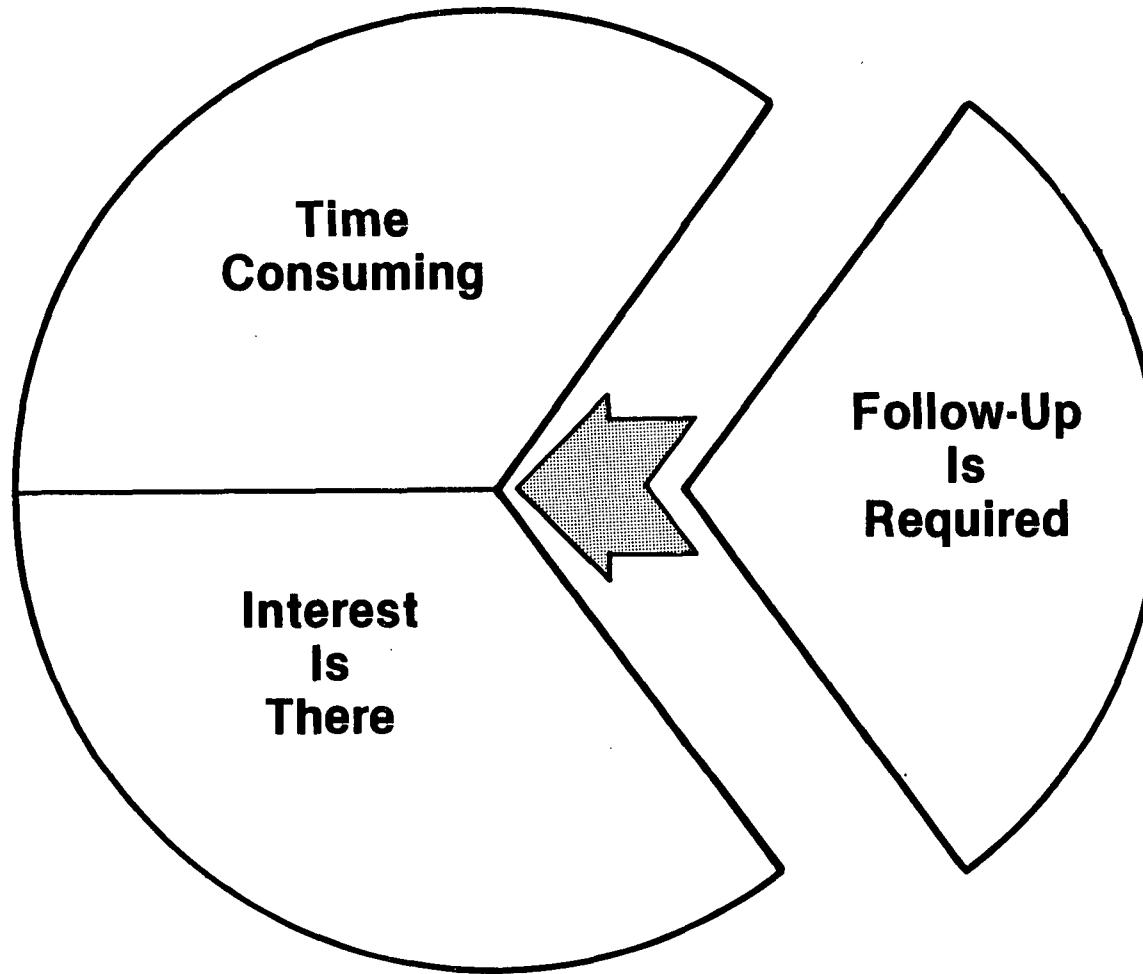
We have reached three general conclusions about user interest:

- (1) Developing the basic relationship required to explore user interest is a multi-step, complex and time consuming process, necessarily involving all of the following steps:
 - Initial contacts at levels where decisions can be made
 - Follow up contacts to establish
 - the framework for controlled exchange of information
 - the range of business and technical areas to be explored
 - Meetings to exchange concepts
 - Follow up analyses to
 - validate feasibility of concepts
 - demonstrate commitment to support user needs
 - Follow up meetings
- (2) Once there is real user interest
 - Potential users will commit significant resources in establishing relationships, conducting meetings and following up
 - An incubation period is required to develop product and process concepts of real value
 - Users will wish to participate in the follow up studies of the conceptual ideas
- (3) User interest must be nurtured by continued exchanges involving:
 - Technological analyses
 - Additional contacts
 - Information exchange

WHAT WE HAVE LEARNED TO DATE

— USER INTEREST —

VFX859



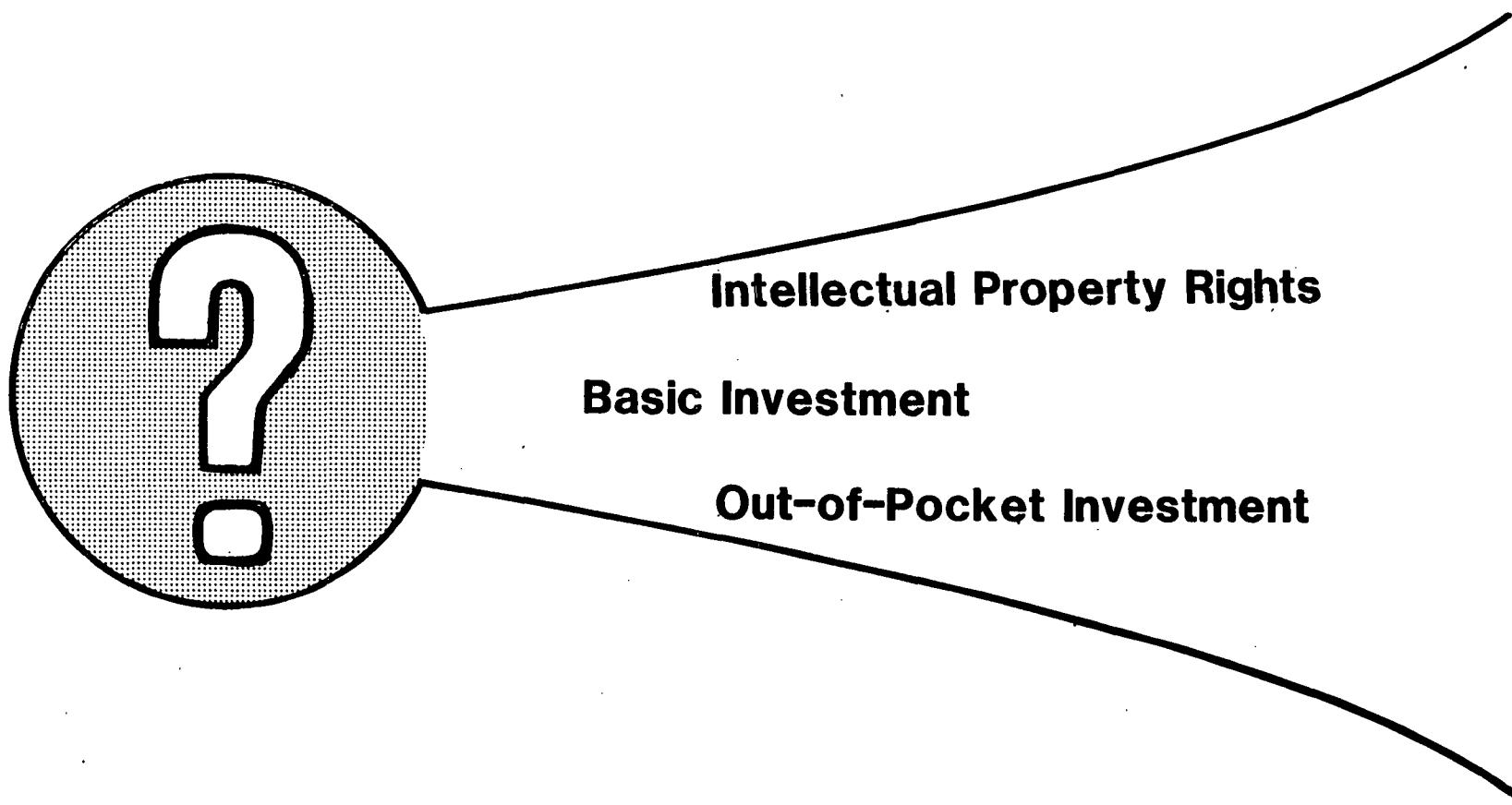
G16

WHAT WE HAVE LEARNED TO DATE - USER CONCERNS

Potential users have expressed a number of concerns which interfere with user commitments to and support of programs -- These include

- The concern that intellectual property rights developed in the conceptualization, investigation and reduction to practice of processes and products cannot be safeguarded in a vehicle involving major government effort
- A question as to who will make the basic investment to develop and launch the vehicle which will support the enterprise which could arise from reduction of a concept to practice
- A need for assurance that the concept can be reduced to practice with use of in-house resources. Users are not interested in making significant out-of-pocket investments in others to do research or hardware development.

WHAT WE HAVE LEARNED TO DATE – USER CONCERNS –



G17

WHAT WE HAVE LEARNED TO DATE - SPEED UP THE PROCESS

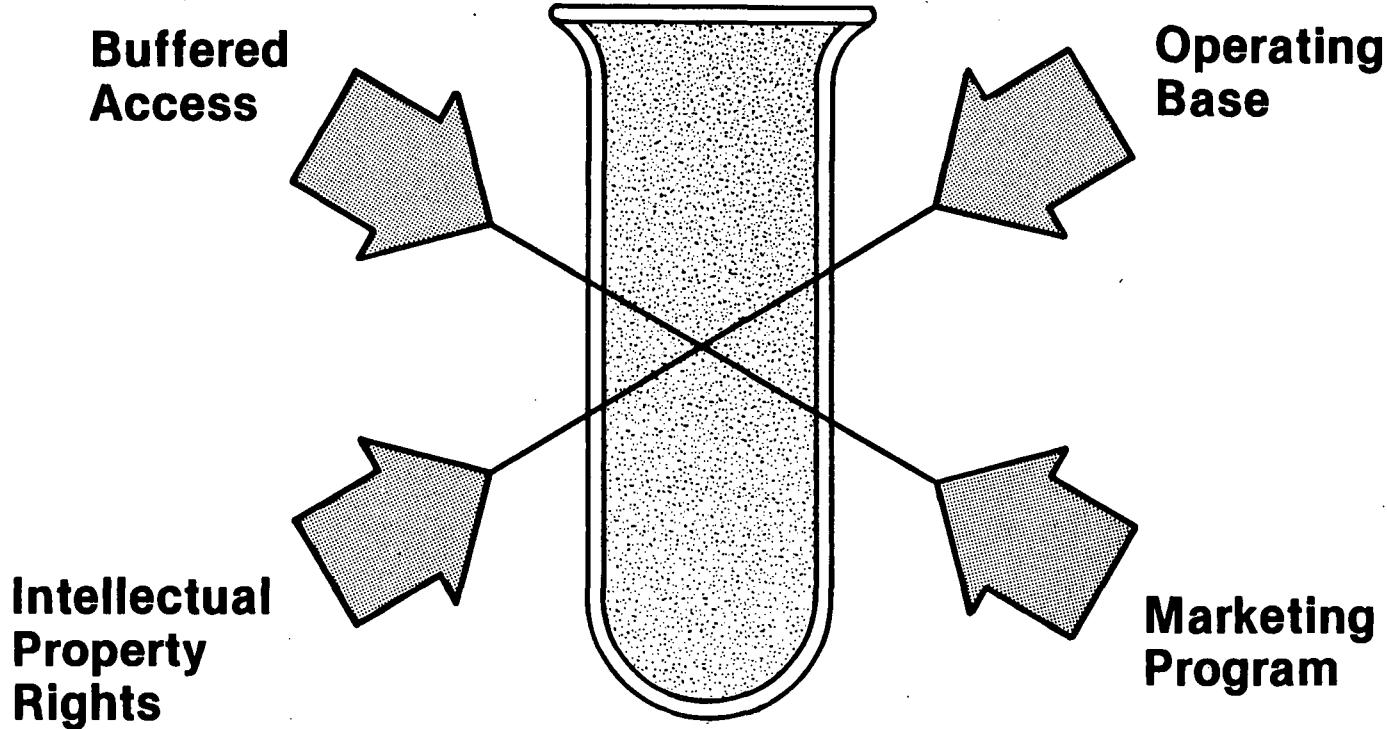
We feel that there are ways for an initializing agency to improve communication with and improve the opportunity to identify, attract and maintain interested, committed commercial users -- These include:

- Development and institutionalization of buffered access techniques which assure user protection of proprietary interests in process, product and service concepts during identification, investigation, analysis and development
- Communication of commitments to protect intellectual property rights after the concept has been developed and is in practice
- Demonstration that there is a program to provide the operating base on which the final enterprise (mission) will be carried out
- Development and application of a formalized, ongoing marketing program which will provide support to the interested user and maintain a visible point of focus for stimulating and developing potential future users.

WHAT WE HAVE LEARNED TO DATE

- SPEED UP THE PROCESS -

VFX861



G18

MISSION REQUIREMENTS (TASK 1)

Technology Development Missions

Space Operations Missions

National Security Missions

Mission Requirements Summary

Dave Riel

TECHNOLOGY DEVELOPMENT MISSIONS

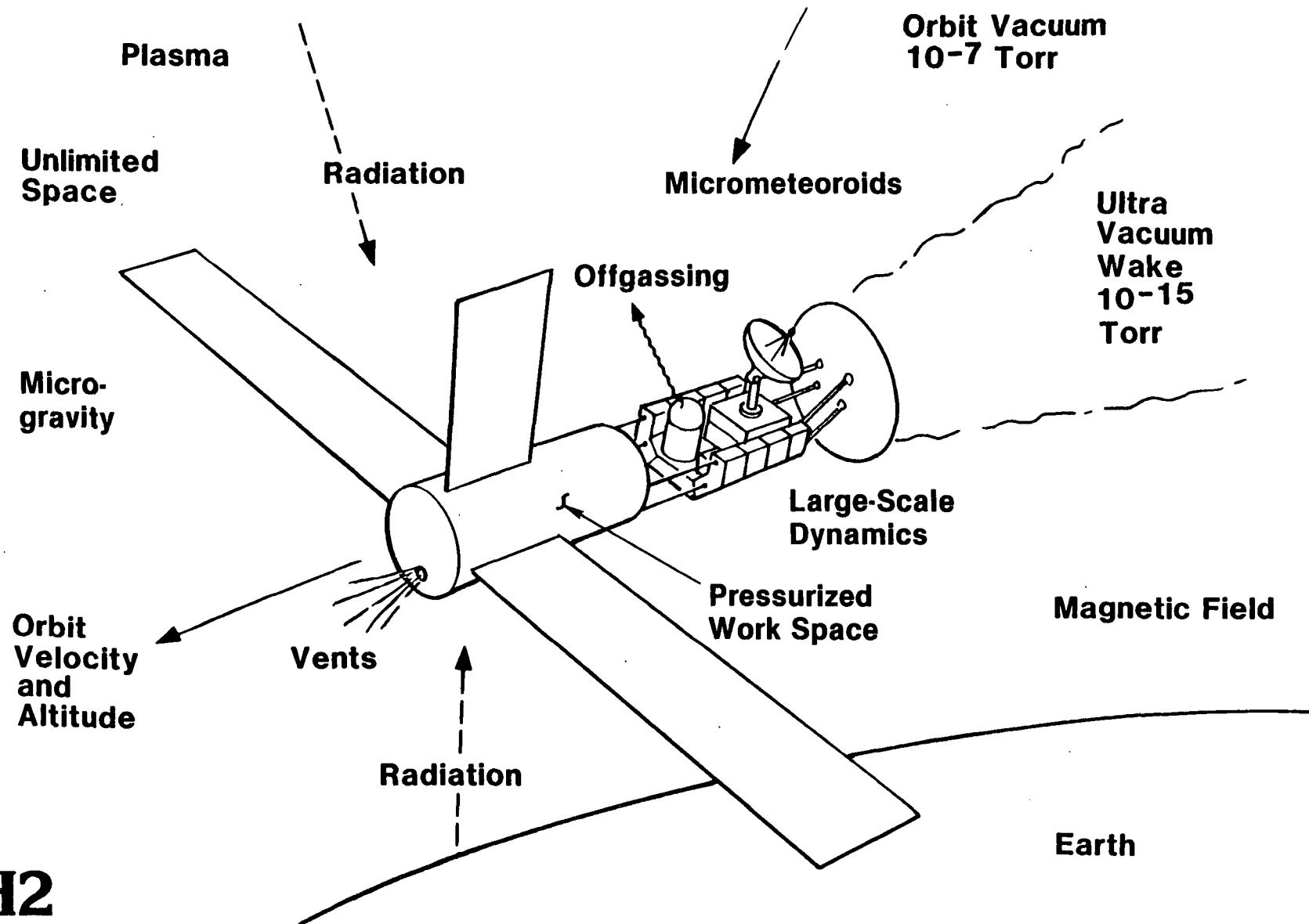
**TECHNOLOGY DEVELOPMENT MISSIONS PROVIDE
ON-ORBIT TESTING WHICH ENABLES:**

- **Generic Mission and Payload Equipment for Future Applications**

- **Technology for Space Station Growth Applications**

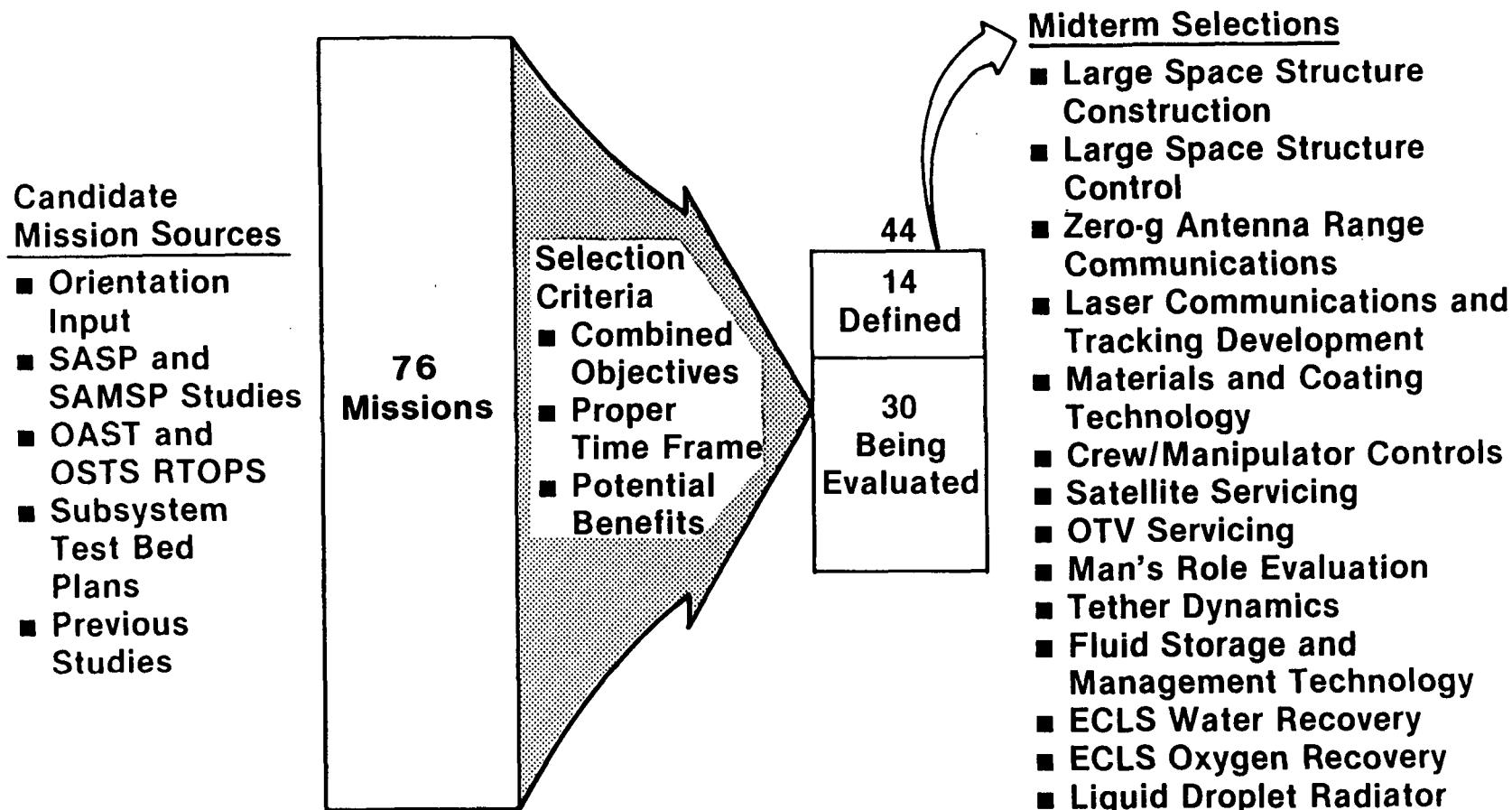
SPACE STATION ENVIRONMENTAL TEST FACILITY

VFY026



TECHNOLOGY DEVELOPMENT MISSION SELECTION

VFY084



H3

MISSION/ENVIRONMENT INTERACTION

| MISSIONS | ENVIRONMENTAL ATTRIBUTES | | | | | | | | | | | |
|-------------------------------------|--------------------------|--------|-----------------|----------------|----------------|----------------|--------------------|----------------------|--------|-----------------|-----------|---------------|
| | MICROGRAVITY | VACUUM | UNLIMITED SPACE | ORBIT ALTITUDE | ORBIT VELOCITY | MAGNETIC FIELD | PRESSURIZED VOLUME | LARGE-SCALE DYNAMICS | PLASMA | MICROMETEOROIDS | RADIATION | CONTAMINATION |
| LARGE SPACE STRUCTURE CONSTRUCTION | X | X | X | | | | | | X | | | |
| LARGE SPACE STRUCTURE CONTROL | X | X | X | | | | | | X | | | |
| ZERO-g ANTENNA RANGE COMMUNICATIONS | | X | | X | | | | | | | | |
| LASER COMMUNICATIONS AND TRACKING | | X | X | | | | | | | | X | |
| MATERIALS AND COATINGS TECHNOLOGY | | X | | | X | | | | X | X | X | X |
| CREW/MANIPULATOR CONTROLS | X | X | | | | | X | X | | | | |
| SATELLITE SERVICING | X | X | | | | | X | X | | | | X |
| OTV SERVICING | X | X | | | | | X | X | | | | |
| MAN'S ROLE EVALUATION | X | X | X | | | | X | X | | | X | X |
| TETHER DYNAMICS | X | X | | | | X | | | X | X | | X |
| FLUID STORAGE AND MANAGEMENT | X | X | | | | | | | | | | |
| ECLS WATER RECOVERY | X | | | | | | X | | | | | |
| ECLS OXYGEN RECOVERY | X | | | | | | X | | | | | |
| LIQUID DROPLET RADIATOR | X | X | X | | | | | X | X | | X | X |

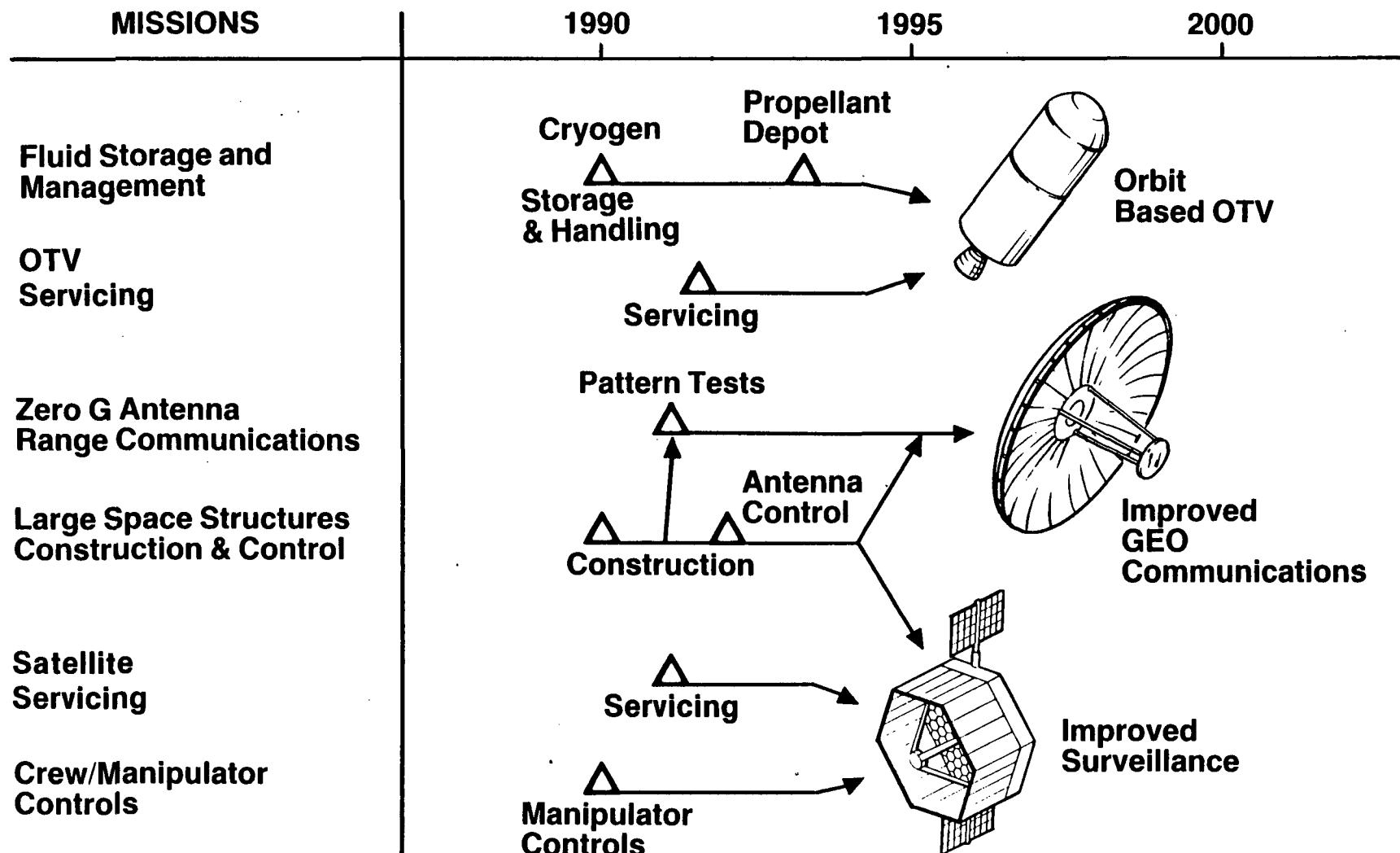
H4

GENERIC PAYLOAD DEVELOPMENT

| MISSION | FUNCTIONAL VALUE | POTENTIAL USE | MISSION CATEGORY |
|-------------------------------------------------------------------------|----------------------------------------------------------------|---------------------------------------------------------------------------------------------|-----------------------------------------------------------------------------------------|
| • LARGE SPACE STRUCTURES CONSTRUCTION AND CONTROL | • HIGH-RESOLUTION ANTENNAS AND MIRRORS | • EARTH OBSERVATION • ASTRONOMY • SURVEILLANCE • COMMUNICATIONS | • SCIENCE AND APPLICATIONS • NATIONAL SECURITY • COMMERCIAL |
| • ZERO-g ANTENNA RANGE COMMUNICATIONS | • IMPROVED DIRECTIONALITY AND ISOLATION | • COMMUNICATIONS | • COMMERCIAL • NATIONAL SECURITY |
| • LASER COMMUNICATIONS AND TRACKING | • HIGHER DATA RATES • NARROW BEAM TRANSMISSION | • EARTH OBSERVATION • COMMUNICATIONS • SURVEILLANCE • SPACE STATION COMMUNICATIONS | • SCIENCE AND APPLICATIONS • COMMERCIAL • NATIONAL SECURITY • SPACE OPERATIONS |
| • CREW MANIPULATOR CONTROLS • SATELLITE SERVICING • OTV SERVICING | • IMPROVED SPACE ROBOTICS • REDUCED MISSION COST | • SATELLITE SERVICE • OTV SERVICE | • ALL MISSION CATEGORIES |
| • FLUID STORAGE AND MANAGEMENT TECHNOLOGY | • IMPROVED ON-ORBIT TRANSFER VEHICLES • REDUCED STS FLIGHTS | • ON ORBIT DEPOT | • SPACE OPERATIONS |

H5

POTENTIAL HIGH VALUE MISSIONS



H6

GROWTH SPACE STATION IMPROVEMENTS

| Subsystem | Technology Development Mission | Major Benefit Areas | | | | |
|-----------------|-----------------------------------|---------------------|-----------------|------|-------------|----------|
| | | Initial Weight | Resupply Weight | Cost | Performance | Security |
| ECLS | O ₂ Recovery | | ▲ | ▲ | | |
| | Water Recovery | | ▲ | ▲ | | |
| CDMS | Laser Communications and Tracking | | | | ▲ | ▲ |
| Thermal Control | Liquid Droplet Radiator | ▲ | | | ▲ | |
| | Material and Coating Technology | ▲ | | | ▲ | |
| ACS | Large Space Structure Control | | | | ▲ | |
| | Tether Thrust and Drag Control | | ▲ | | ▲ | |
| Crew Systems | Man's Role Evaluation | | | ▲ | ▲ | |

H7

TECHNOLOGY DEVELOPMENT MISSION MIDTERM CONCLUSIONS

VFY021

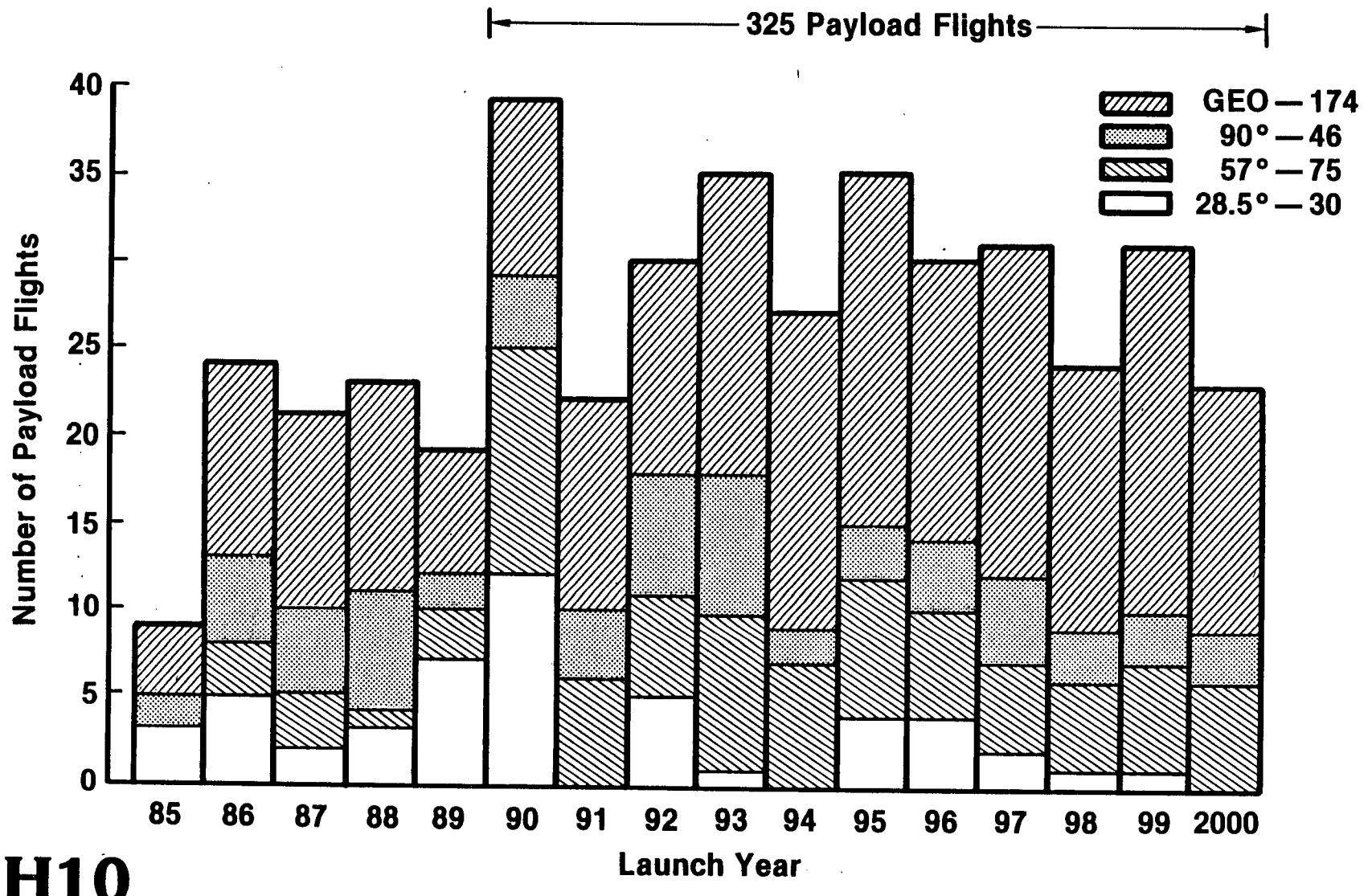
Technology Development Missions:

- Utilize the Unique Space Station Environment
 - Enable:
 - Advanced Mission Technology
 - Increased Space Station Capability
 - Provide Benefits to All Categories of Users
 - Require Manned Participation For the Majority
 - Are Relatively Short Term and Orbit Independent
 - Require Exterior Volume — 15 Pallets (Equiv)
- 

SPACE OPERATIONS MISSIONS

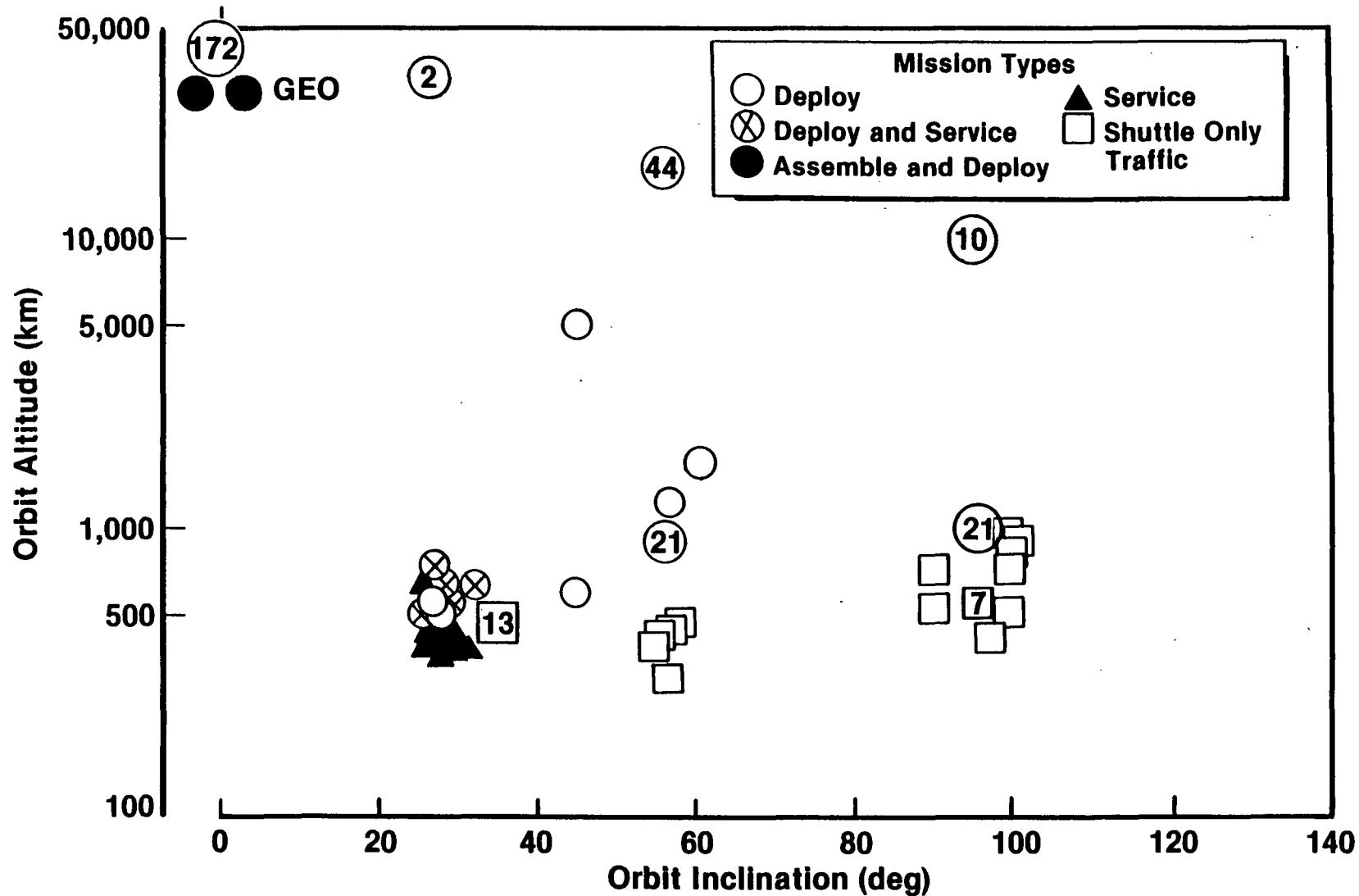
- Transportation (OTV, TMS, Other)
 - Deploy/Retrieve
 - Debris Collection
- Assembly, Integration, Checkout
 - Large Structures
 - Stage/Payload Mating
- Service
 - Maintain/Repair/Resupply
 - Instrument Reconfiguration
- Storage
 - Propellants (Cryo, Storables)
 - Spares
 - Payloads
- Space Utilization
 - Quarantine
 - Rescue

SPACE OPERATIONS MISSION MODEL



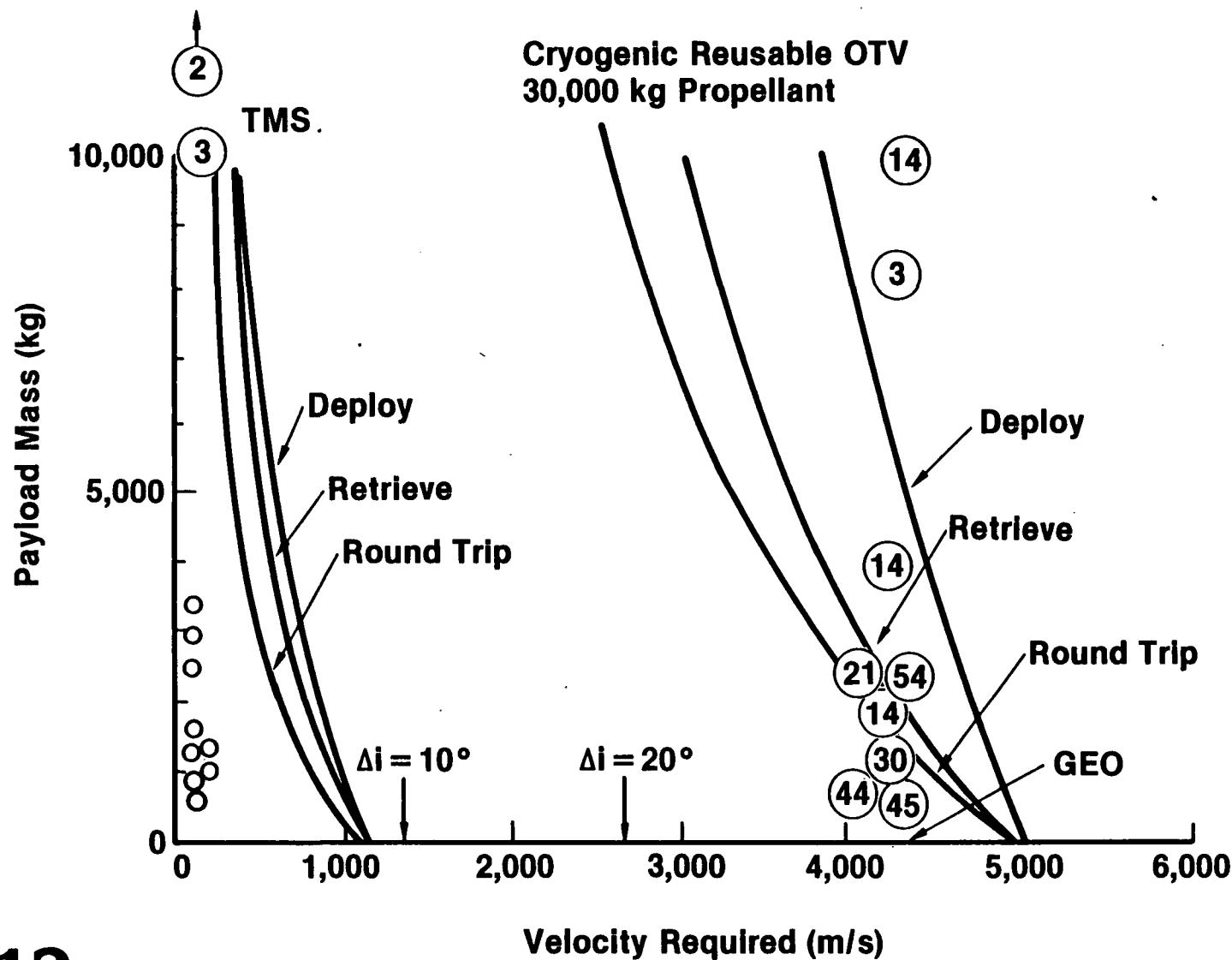
H10

SPACE OPERATIONS MISSIONS 1990-2000



H11

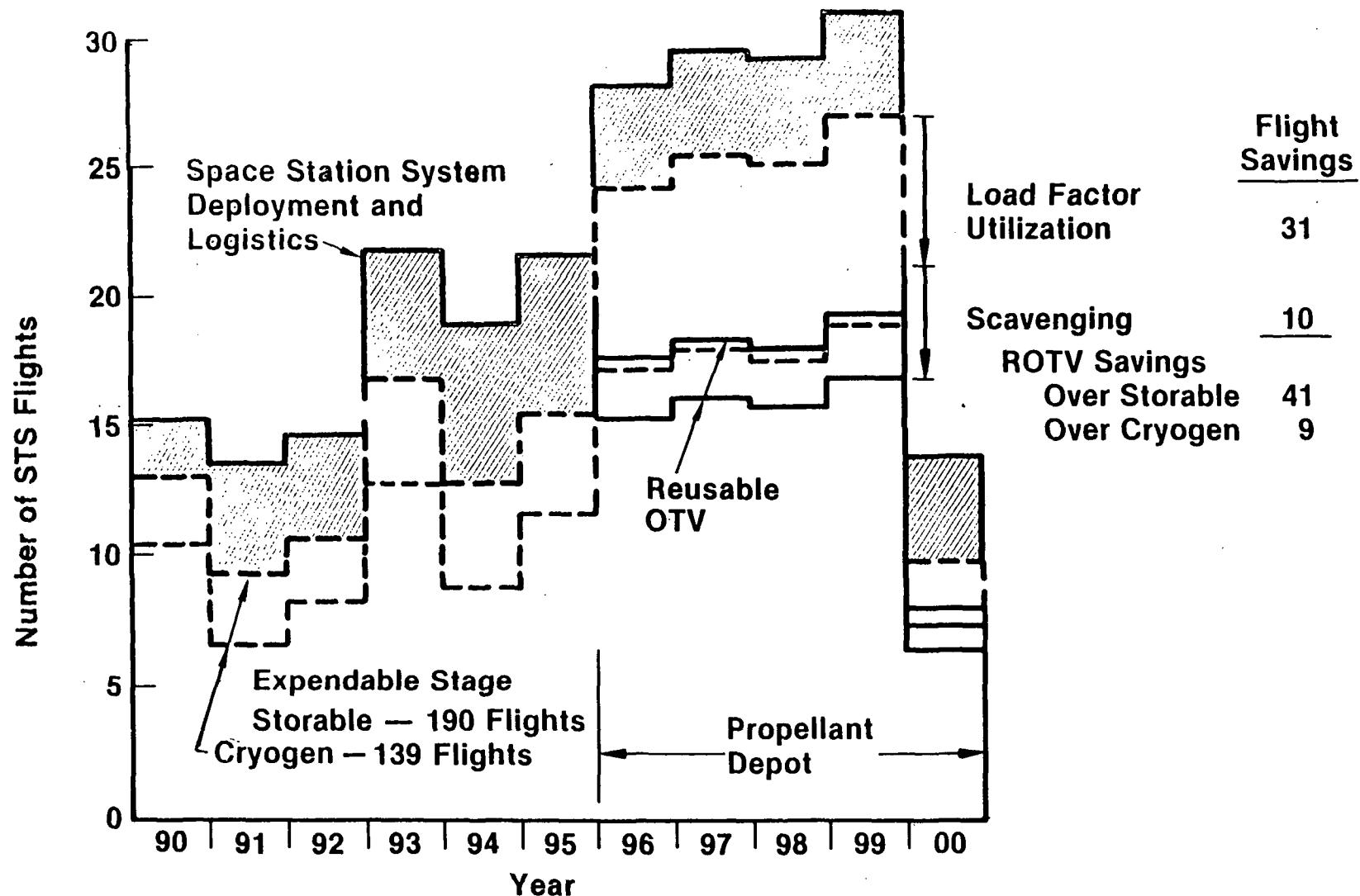
ORBIT TRANSFER VEHICLE CAPABILITY



H12

STS FLIGHT REQUIREMENTS REUSABLE OTV AND DEPOT EFFECTS

VFX979



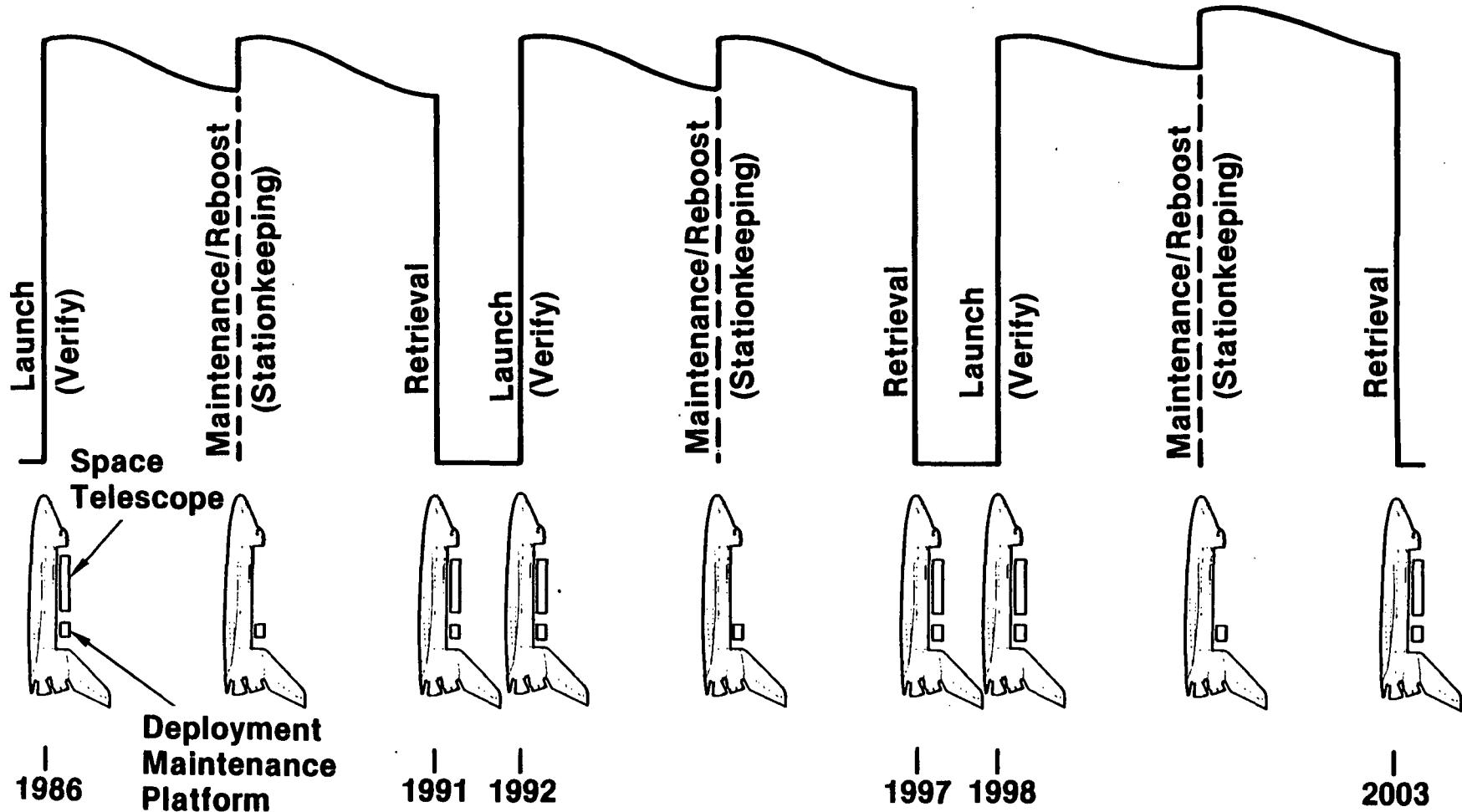
H13

SPACE OPERATIONS ISSUES

- Traffic Model Validation
- Shuttle Fleet Size
- Shuttle Utilization Factor
- Upper Stage Program Development
- Cryogen Scavenging Feasibility
- Orbital Propellant Depot Cost

H14

SPACE TELESCOPE MAINTENANCE



H15

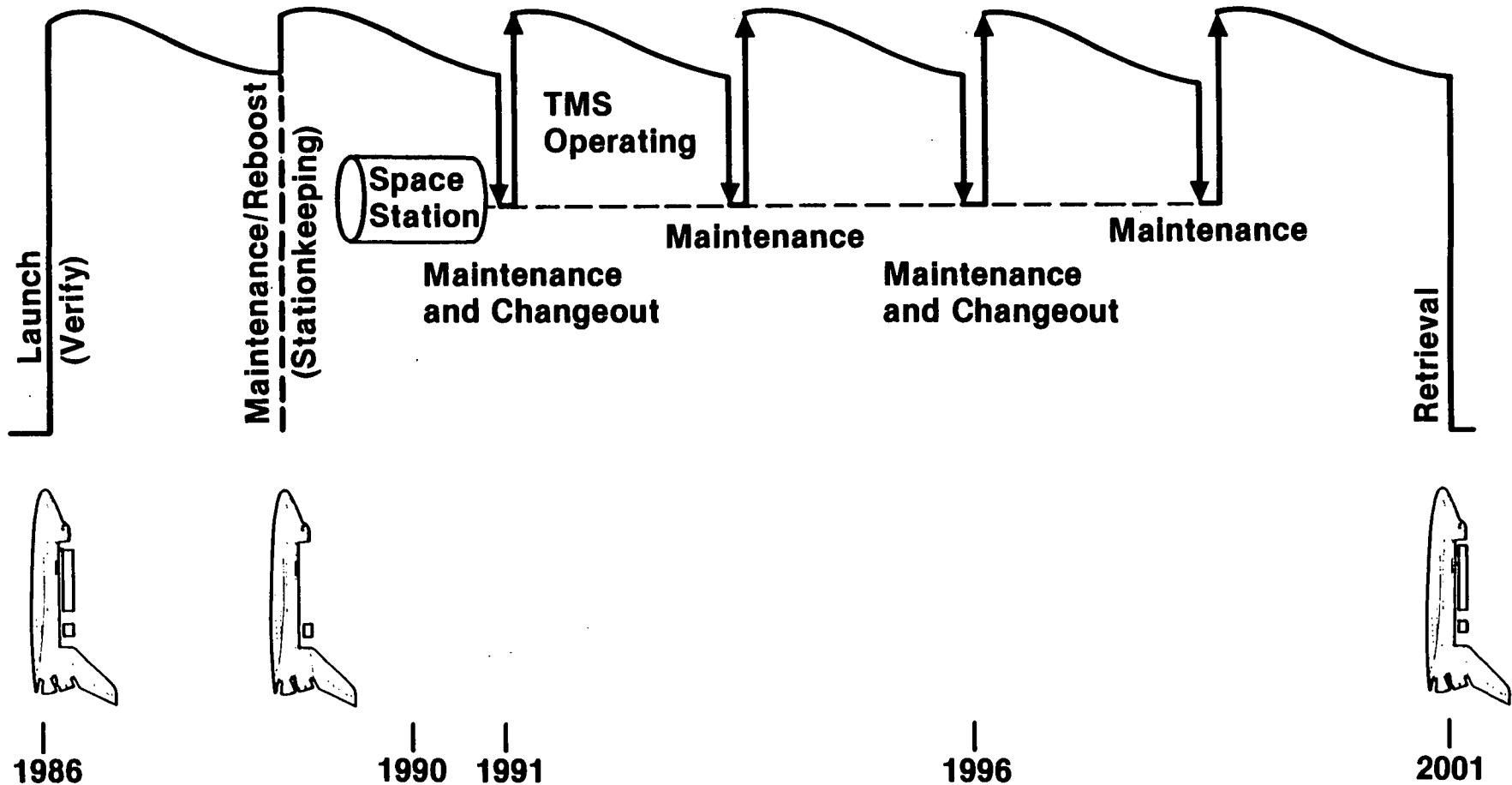
SPACE TELESCOPE MAINTENANCE

| | Planned Service Items | Designed Service Items | Potential Service Items |
|----------|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-----------------------------------------------------------------------------------------------------------------------------------------------------------------------------|------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| On-Orbit | <ul style="list-style-type: none"> ■ Batteries (6) ■ Fine Guidance Sensor (2) ■ Guidance Sensor Electronics (1) ■ Rate Sensor Unit (1) ■ Rate Sensor Electronics (1) ■ SI Control and Data Handling (1) ■ Axial Scientific Instrument (1) | | |
| Ground | <ul style="list-style-type: none"> ■ FH Star Tracker ■ Radial Scientific Instr ■ ACS Units ■ Tape Recorder ■ Sun Sensors | <ul style="list-style-type: none"> ■ FH Star Tracker ■ Radial Scientific Instr ■ ACS Units ■ Tape Recorder ■ Sun Sensors | <ul style="list-style-type: none"> ■ Comm/Data Mgmt Unit ■ Solar Array Wings ■ Solar Array Drive Electronics ■ Change Current Controller |

H16

SPACE TELESCOPE MAINTENANCE SPACE STATION

VFX819

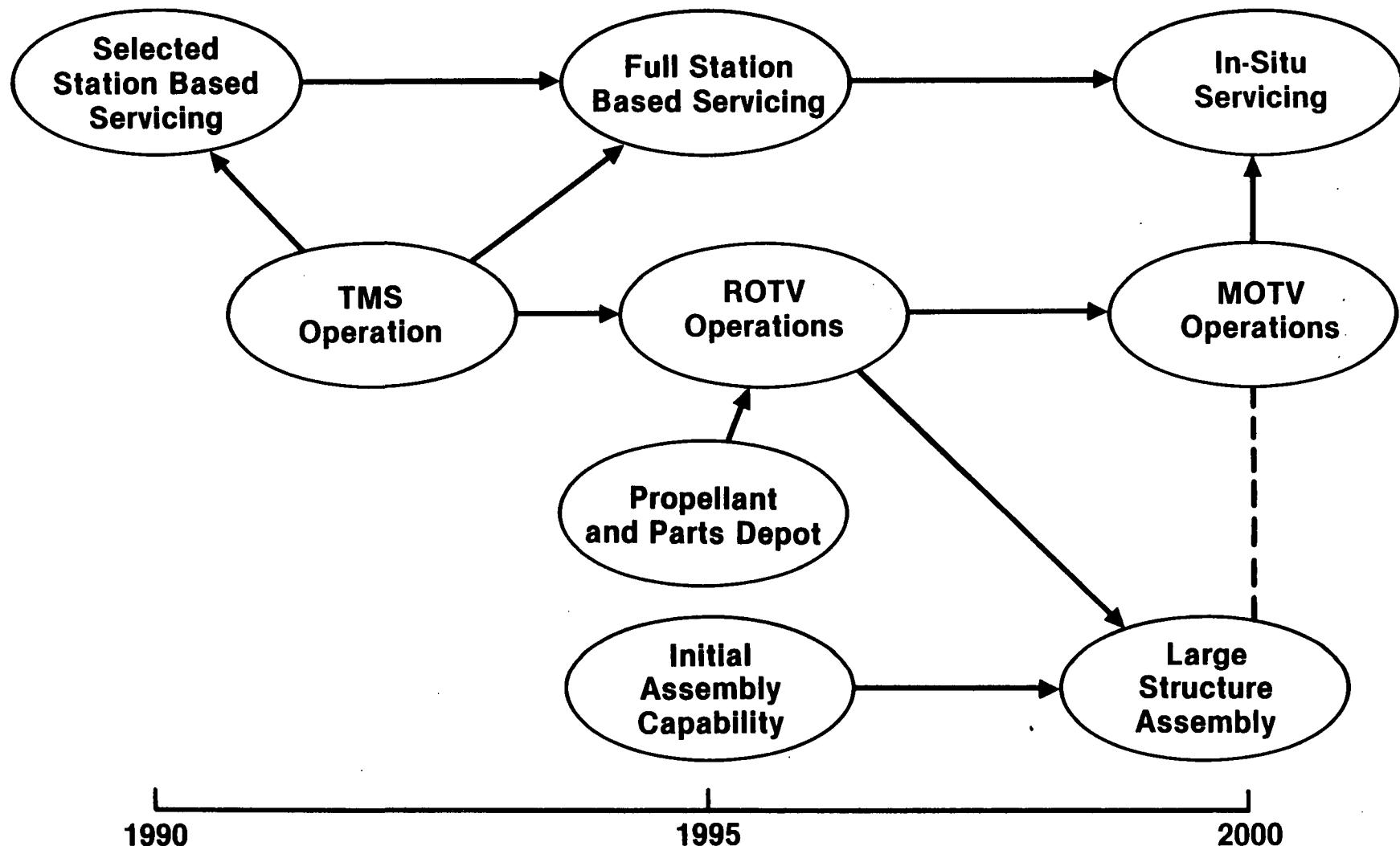


- Orbiter Requires DMP

H17

REQUIRED SPACE OPERATIONS CAPABILITIES

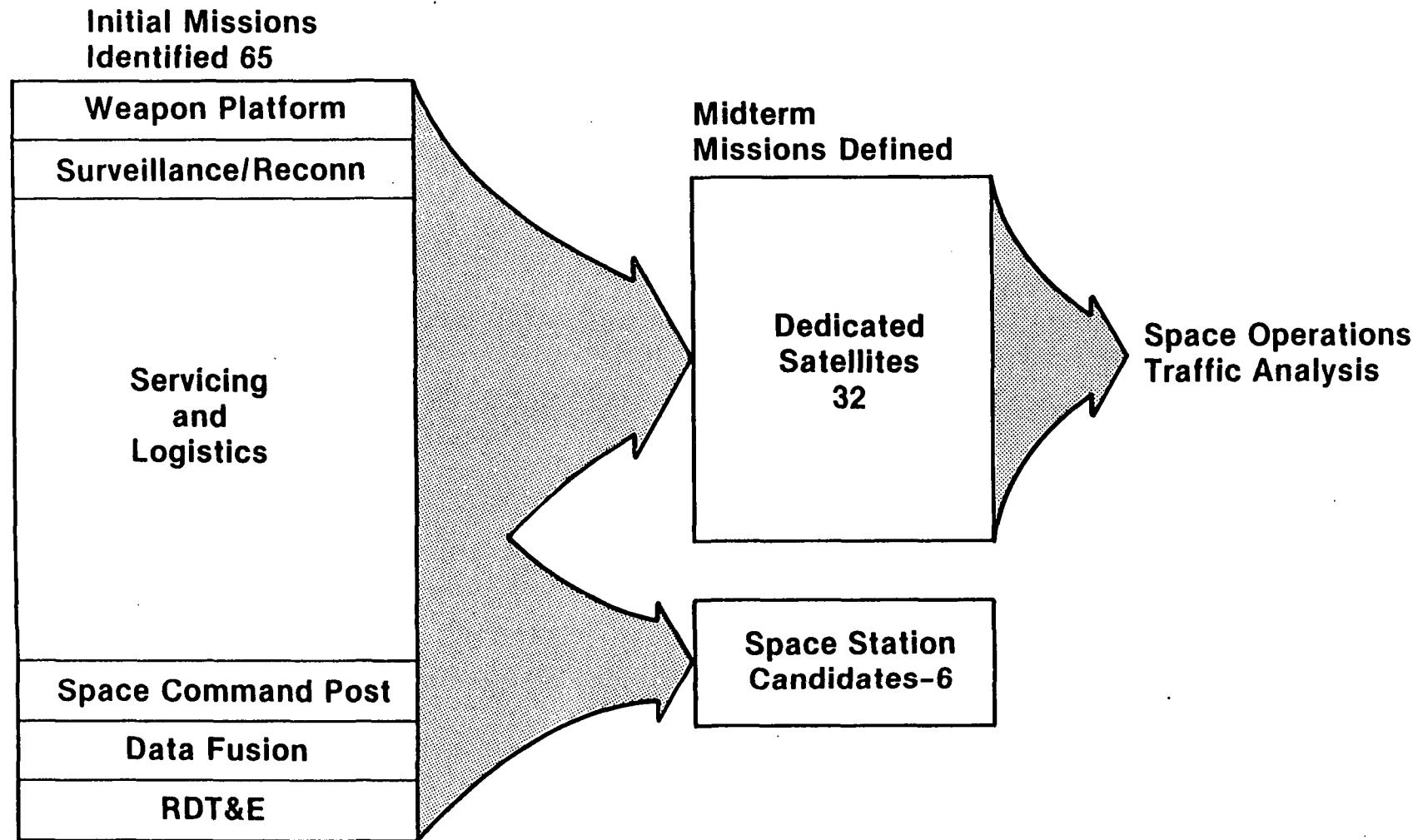
VFY019



H18

NATIONAL SECURITY MISSIONS

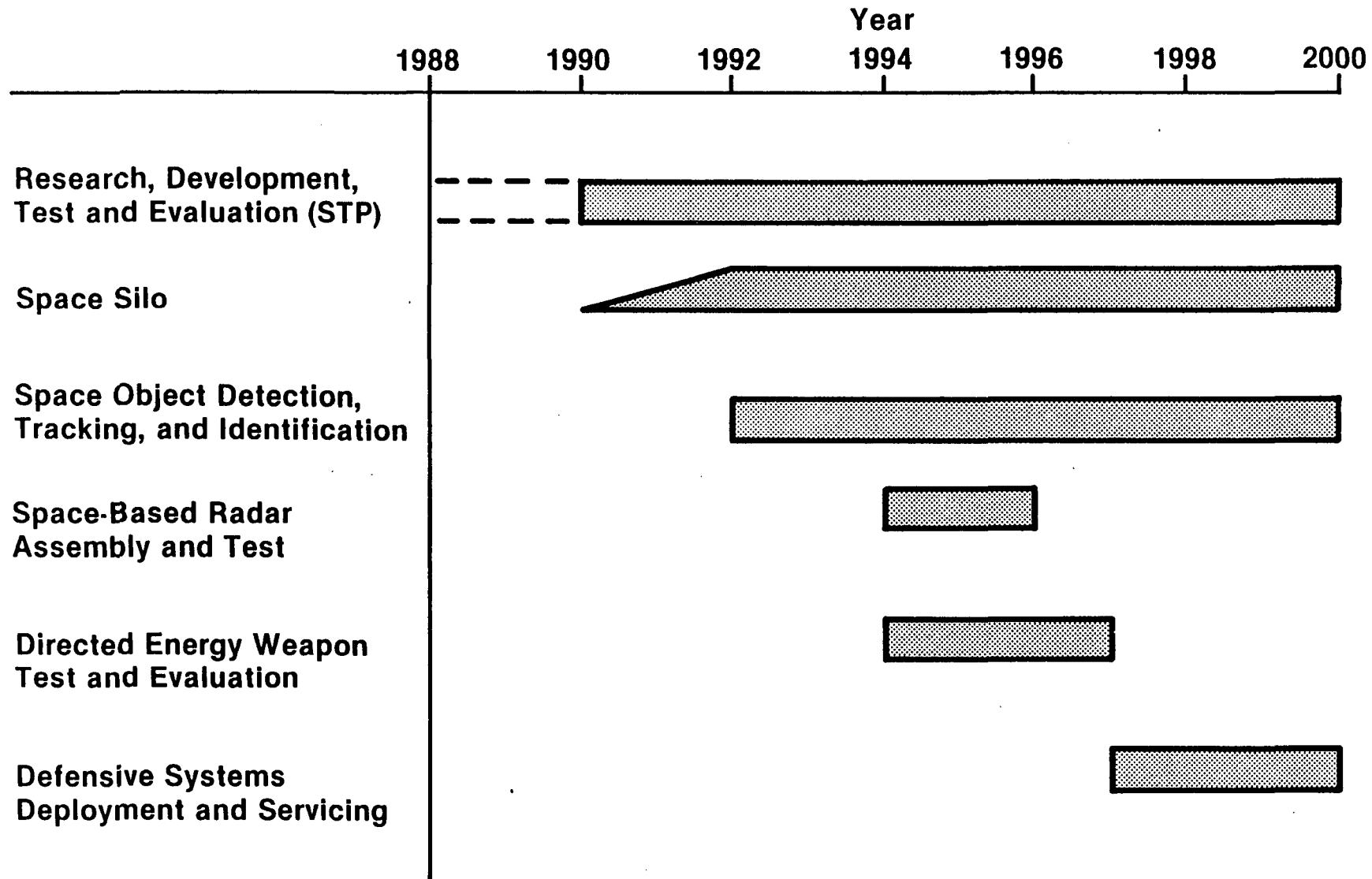
VFX972



H19

NATIONAL SECURITY MISSIONS MIDTERM

VFY144



H20

NATIONAL SECURITY MISSION REQUIREMENTS

VFY018

| | <u>INITIAL</u> | <u>ULTIMATE</u> |
|-----------------------|------------------------|------------------------|
| Orbit | 28.5° Initial | Polar |
| Facilities (Equiv) | 2 Modules 2 Pallets | 3 Modules 4 Pallets |
| Crew | 1-2 | 2-5 |
| Power | 3 kW | 10 kW |

H21

NATIONAL SECURITY MISSIONS MIDTERM CONCLUSIONS

- Minimal Influence on Overall Requirements
- Prime Continuing Missions Are RDT&E
 - Additional Duration, Power, and Crew
 - Reduced Support System Requirements
- Space Station Enhances Current Capabilities
 - Reduced Satellite Response/Replacement Time
 - Enhanced Space System Survivability/Endurance
- Space Station Offers New Mission Capabilities
 - Low-Cost Space Sensor Base
 - Orbital Assembly of Large Structures
 - Potential Depot Advantages

MISSION REQUIREMENTS SUMMARY

Dave Riel

MIDTERM MISSION SET — 95 TOTAL

| CODE | NAME | N ^o | DUR | INC | INC | INC | ALT | ALT | ALT | STAT | SER | NO | HRS | POWER | MASS G | KG | |
|--------|----------------------|----------------|-----|------|-----|-----|-----|-----|-----|-------|------|----|------|-------|--------|-------|---|
| | | | | | | | | | | | | | | | DATA | KBPS | |
| | | | | DATE | DEG | MAX | MIN | KM | MAX | MIN | PLAT | CR | DAY | | | | |
| SAS001 | SOLAR OPT TELE | 88 | 1 | 33 | 57 | 28 | 400 | 435 | 370 | R | 1.00 | 4 | 5000 | 8175 | 0 | 3082 | 0 |
| SAS002 | SIRTF | 89 | 10 | 28 | 57 | 0 | 400 | 430 | 350 | R | 1.00 | 4 | 1045 | 8103 | 0 | 1300 | 0 |
| SAS003 | STARLAB | 90 | 1 | 28 | 57 | 20 | 400 | 800 | 350 | R | 1.00 | 4 | 2220 | 3280 | 0 | 16559 | 0 |
| SAS004 | COMP SPEC COSRAY NUC | 94 | 2 | 57 | 57 | 28 | 400 | 435 | 370 | R | 1.00 | 8 | 731 | 10000 | 0 | 3550 | 0 |
| SAS005 | SOL SOFT XRAY TS | 88 | 5 | 57 | 98 | 28 | 430 | 600 | 350 | D A M | 1.00 | 2 | 240 | 11070 | 0 | 0 | 0 |
| SAS006 | SOLAR TERR OBS | 93 | 2 | 57 | 57 | 57 | 400 | 350 | 600 | D A L | 1.00 | 8 | 3000 | 1668 | 0 | 240 | 0 |
| SAS007 | PINHOLE XRAY | 91 | 1 | 97 | 97 | 80 | 370 | 800 | 350 | R | 1.00 | 8 | 0 | 1000 | 0 | 0 | 0 |
| SAS008 | XRAY OBSE | 92 | 2 | 28 | 57 | 0 | 400 | 370 | 370 | A A | 1.00 | 2 | 900 | 2600 | 0 | 0 | 0 |
| SAS009 | SPACE TELESC | 85 | 15 | 28 | 28 | 28 | 600 | 600 | 600 | M | 3.00 | 8 | 2100 | 11000 | 0 | 0 | 0 |
| SAS010 | HIPRES X&G-RAY SPEC | 90 | 2 | 28 | 45 | 0 | 400 | 500 | 350 | A A M | 1.00 | 2 | 530 | 9516 | 0 | 0 | 0 |
| SAS011 | XRAY TIMING EXPL | 88 | 2 | 28 | 57 | 0 | 400 | 600 | 300 | A M | 1.00 | 4 | 600 | 1354 | 0 | 0 | 0 |
| SAS012 | SOLAR INT DYNAMICS | 91 | 5 | 28 | 28 | 28 | 575 | 575 | 575 | M | 2.00 | 8 | 800 | 20500 | 0 | 0 | 0 |
| SAS013 | ADV XRAY ASTROFAC | 91 | 10 | 28 | 57 | 0 | 500 | 600 | 400 | A | 1.00 | 4 | 2000 | 0 | 0 | 0 | 0 |
| SAS014 | LAMAR | 92 | 6 | 28 | 57 | 0 | 400 | 435 | 200 | A M | 1.00 | 2 | 3400 | 0 | 0 | 0 | 0 |
| SAS015 | VLBI | 93 | 3 | 28 | 57 | 28 | 400 | 600 | 350 | A B | 2.00 | 8 | 900 | 0 | 0 | 0 | 0 |
| SAS016 | LRG AMB DEPL IRTSC | 93 | 10 | 28 | 50 | 28 | 700 | 800 | 400 | M | 3.00 | 8 | 3000 | 0 | 0 | 0 | 0 |
| SAS017 | ADV SOLAR OBSR | 93 | 5 | 57 | 57 | 0 | 400 | 600 | 250 | P | 1.00 | 4 | 4000 | 0 | 0 | 0 | 0 |
| TGN001 | LSS CONTR EXP | 92 | 1 | 28 | 90 | 0 | 400 | 999 | 300 | R | 1.00 | 1 | 1000 | 250 | 0 | 1000 | 0 |
| TGN002 | ZERO G ANT RANGE | 90 | 1 | 28 | 90 | 28 | 400 | 999 | 300 | R | 2.00 | 8 | 1000 | 9000 | 0 | 600 | 0 |
| TGN003 | MATERIALS&COAT TECH | 90 | 5 | 28 | 97 | 28 | 400 | 600 | 300 | D M | 1.00 | 2 | 0 | 200 | 0 | 0 | 0 |
| TGN004 | TETHER DYNAMICS | 92 | 1 | 28 | 97 | 0 | 400 | 600 | 300 | R | 1.00 | 2 | 1000 | 0 | 0 | 0 | 0 |
| TGN005 | LRG STRUCT CONSTR | 92 | 1 | 28 | 57 | 28 | 400 | 999 | 350 | R | 2.00 | 8 | 500 | 0 | 0 | 0 | 0 |
| TGN006 | FLUID STORE&MANAG | 90 | 1 | 28 | 97 | 28 | 400 | 999 | 300 | R | 1.00 | 4 | 500 | 0 | 0 | 0 | 0 |
| TGN007 | LIQUID DROPLET RAD | 92 | 1 | 28 | 97 | 0 | 400 | 999 | 350 | R | 1.00 | 4 | 200 | 0 | 0 | 0 | 0 |
| TOP001 | SATELL SERV TECH | 90 | 99 | 28 | 90 | 28 | 400 | 999 | 300 | R | 2.00 | 8 | 0 | 0 | 0 | 0 | 0 |
| H24 | TV SERVICE TECH | 90 | 1 | 28 | 90 | 28 | 400 | 999 | 300 | R | 2.00 | 4 | 1500 | 0 | 0 | 0 | 0 |

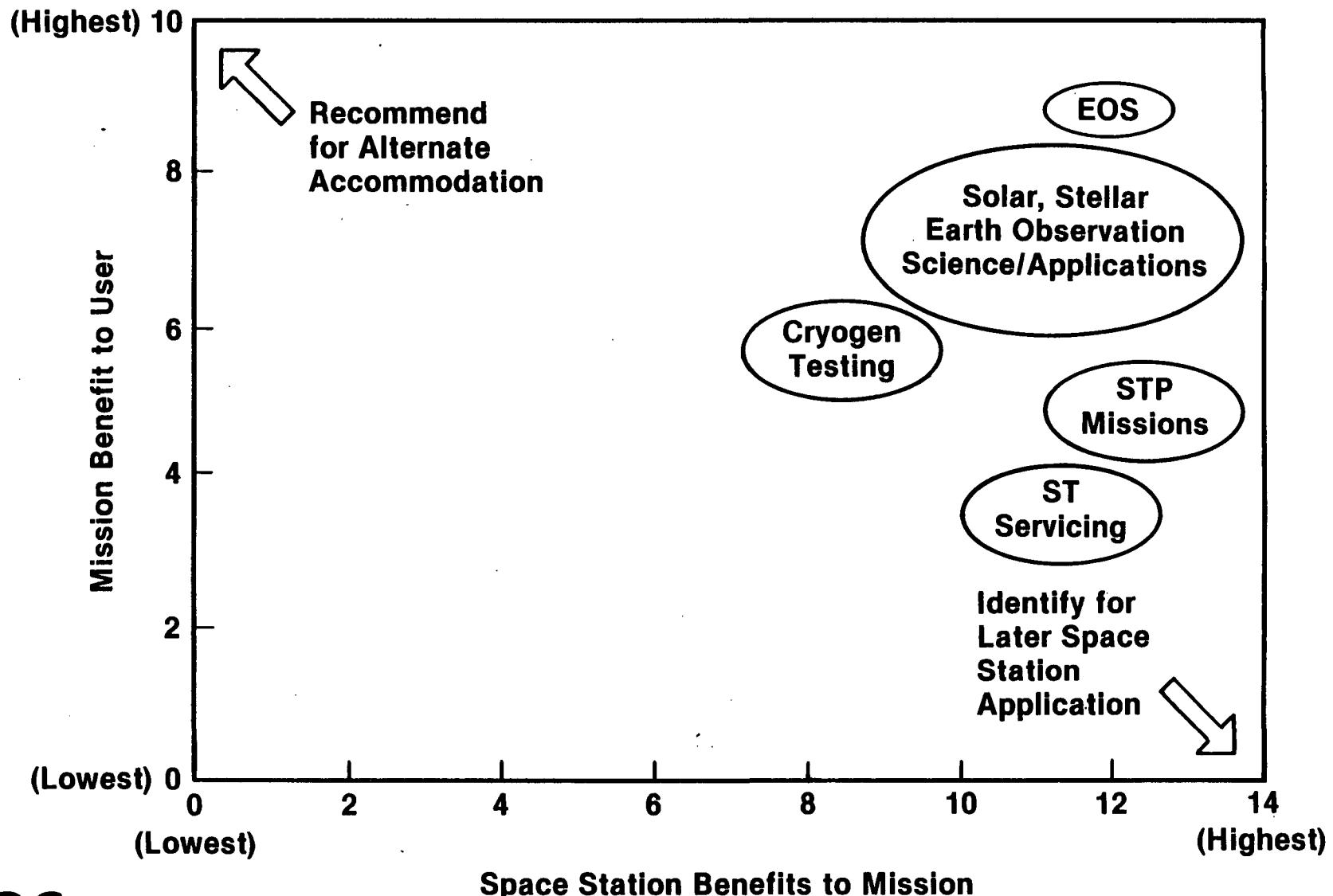
RELATIVE MISSION VALIDATION

| <u>VALIDATION SCALE</u> | <u>SCIENCE AND APPLICATION</u> | <u>COMMERCIAL</u> | <u>NATIONAL SECURITY</u> | <u>TECHNOLOGY DEVELOPMENT</u> | <u>SPACE OPERATIONS</u> |
|------------------------------------|-------------------------------------------------------------------------------------------|-------------------------|-----------------------------------------------------------------------|-------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-----------------------------------------------------------------------------------------------------|
| EXISTING SYSTEM | { ASTROPHYSICS { EARTH AND PLANETARY { ENVIRON OBSERVATORY { ASTROPHYSICS | | | | ● SERVICING ● PAYLOAD DEPLOYMENT |
| EXISTING SUPPORT CAPABILITY | | | ● SPACE TEST PROGRAM | | |
| SYSTEM DEVELOPMENT | | | | { SERVICING ● FLUID STORAGE | |
| TECHNOLOGY DEVELOPMENT | | ● EOS | | | |
| FUNDED DEFINITION STUDIES | ● EARTH AND PLANETARY ● COMMUNICATION ● LIFE SCIENCES ○ MATERIAL ● PROCESSING | ○ CRYSTAL FACILITY | ○ SPACE BASED RADAR ○ DIRECTED ENERGY WEAPON ○ DEFENSIVE SYSTEM | ● ADV ECLS ● ANTENNA RANGE ● MATERIALS ○ LARGE STRUCT ○ TETHER ● LASER COMM ○ DROPLETS ○ CREW MANIPULATOR ○ MAN'S ROLE EVAL ○ LIQUID DROP RADIATOR | ○ STORABLE OPS ○ ASSEMBLY ○ CRYO OPS ○ DEPOT ○ DEBRIS COLLECTION ○ PLANETARY SUPPORT |
| FUNDED STUDIES PLANNED STUDIES | ● ASTROPHYSICS ● EARTH AND PLANETARY ○ ENVIRON OBSERVATORY | ○ RESEARCH FAC | | | |
| NEEDS IDENTIFIED CONCEPTS PROPOSAL | ○ LIFE SCIENCES | ○ PRODUCTION FACILITIES | ● SPACE DETECTION ● SILO | | |
| CONCEPT IDEA | | | | | |

● 1990-1991
○ 1992-2000

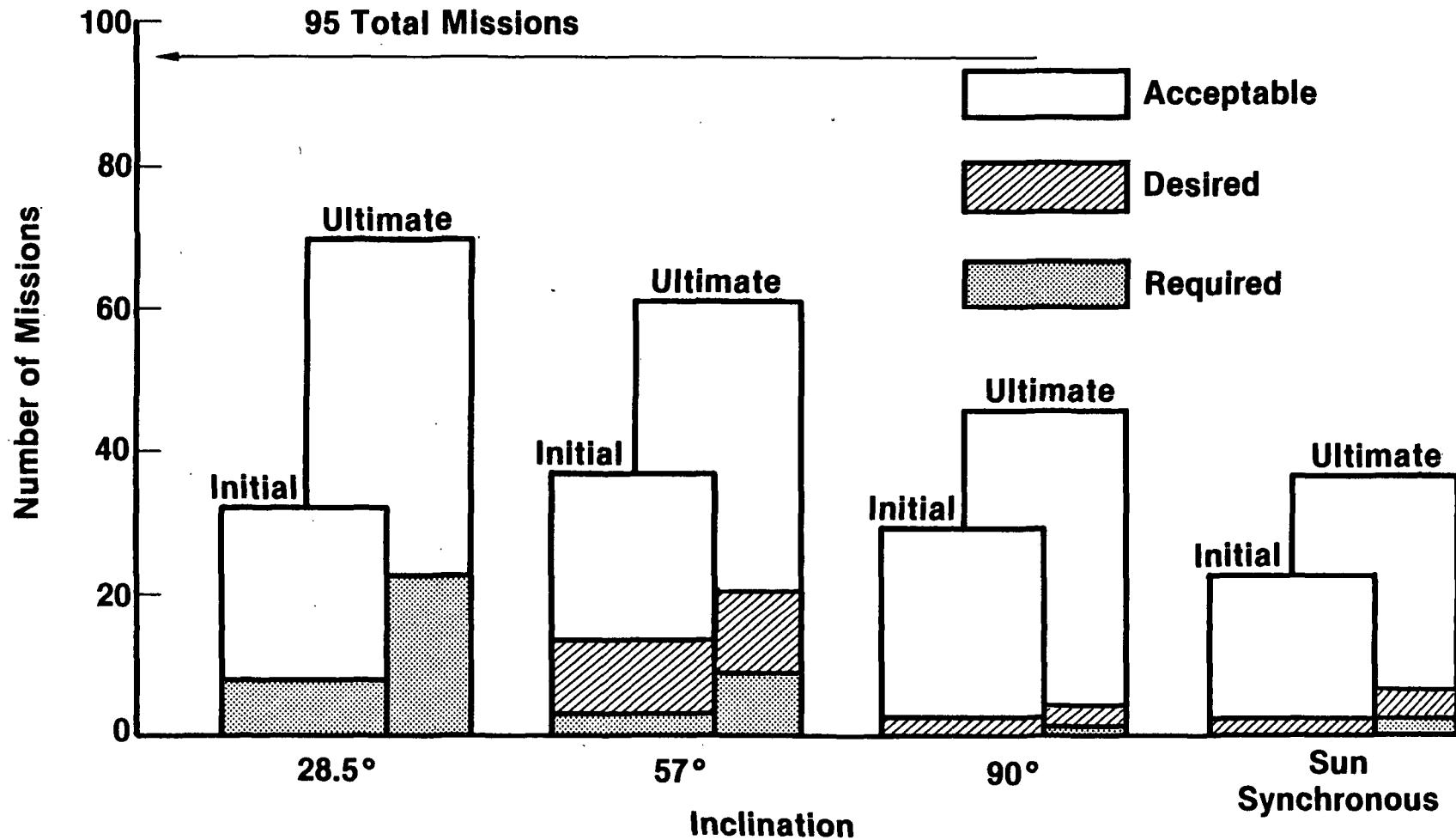
H25

MISSION PRIORITY



H26

MISSION INCLINATION DISTRIBUTION

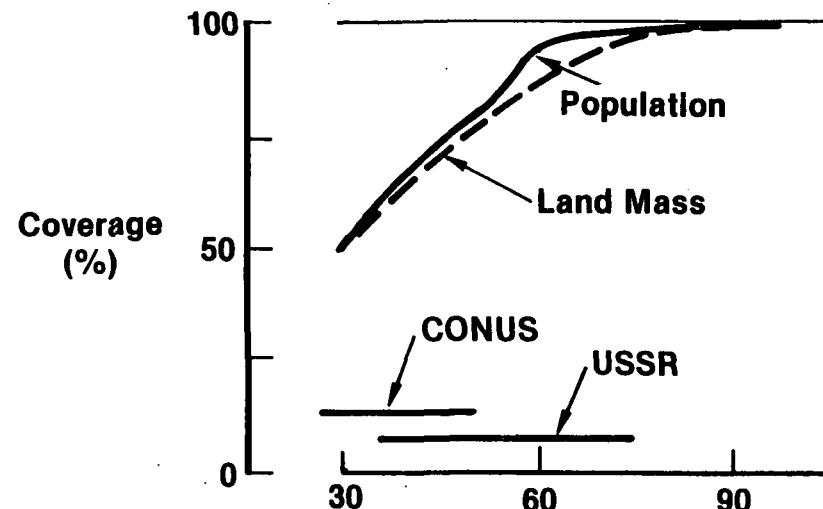


H27

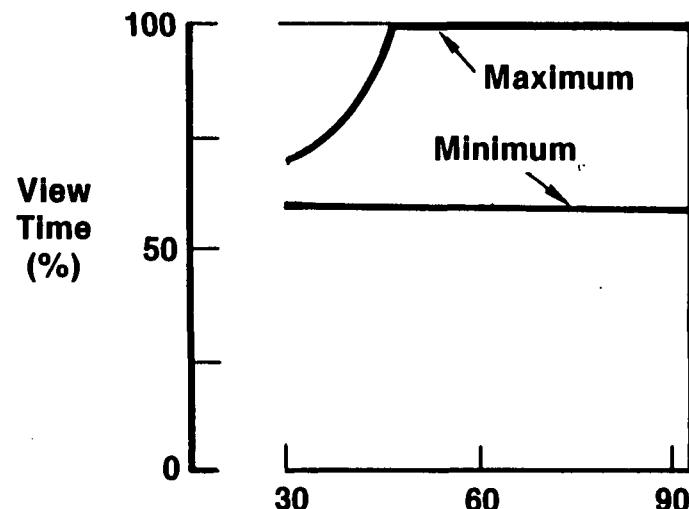
ORBIT INCLINATION SENSITIVITIES

VFX823

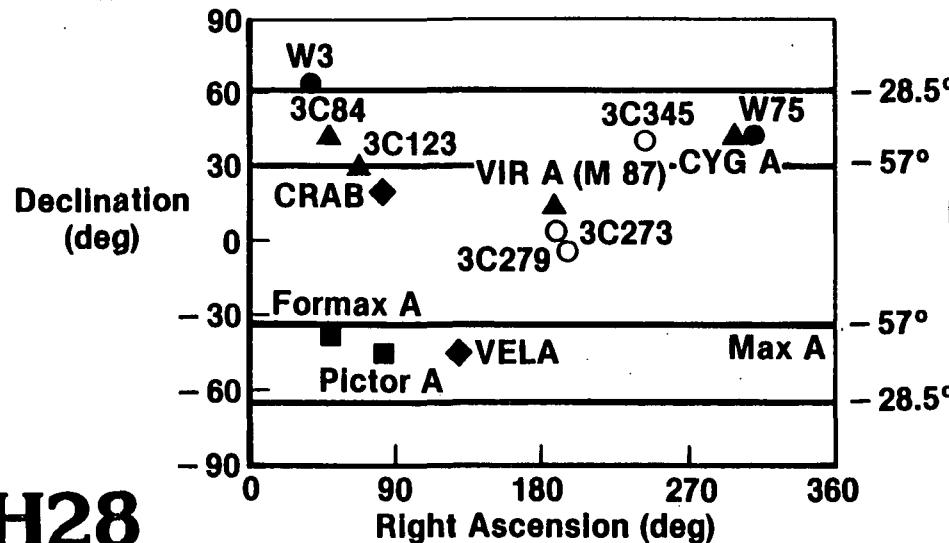
EARTH OBSERVATIONS



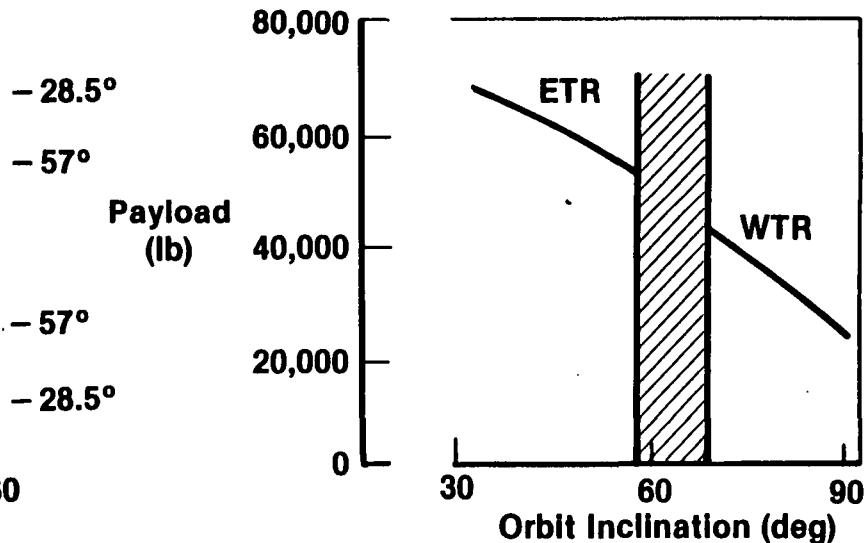
SOLAR VIEWING



STELLAR VIEWING



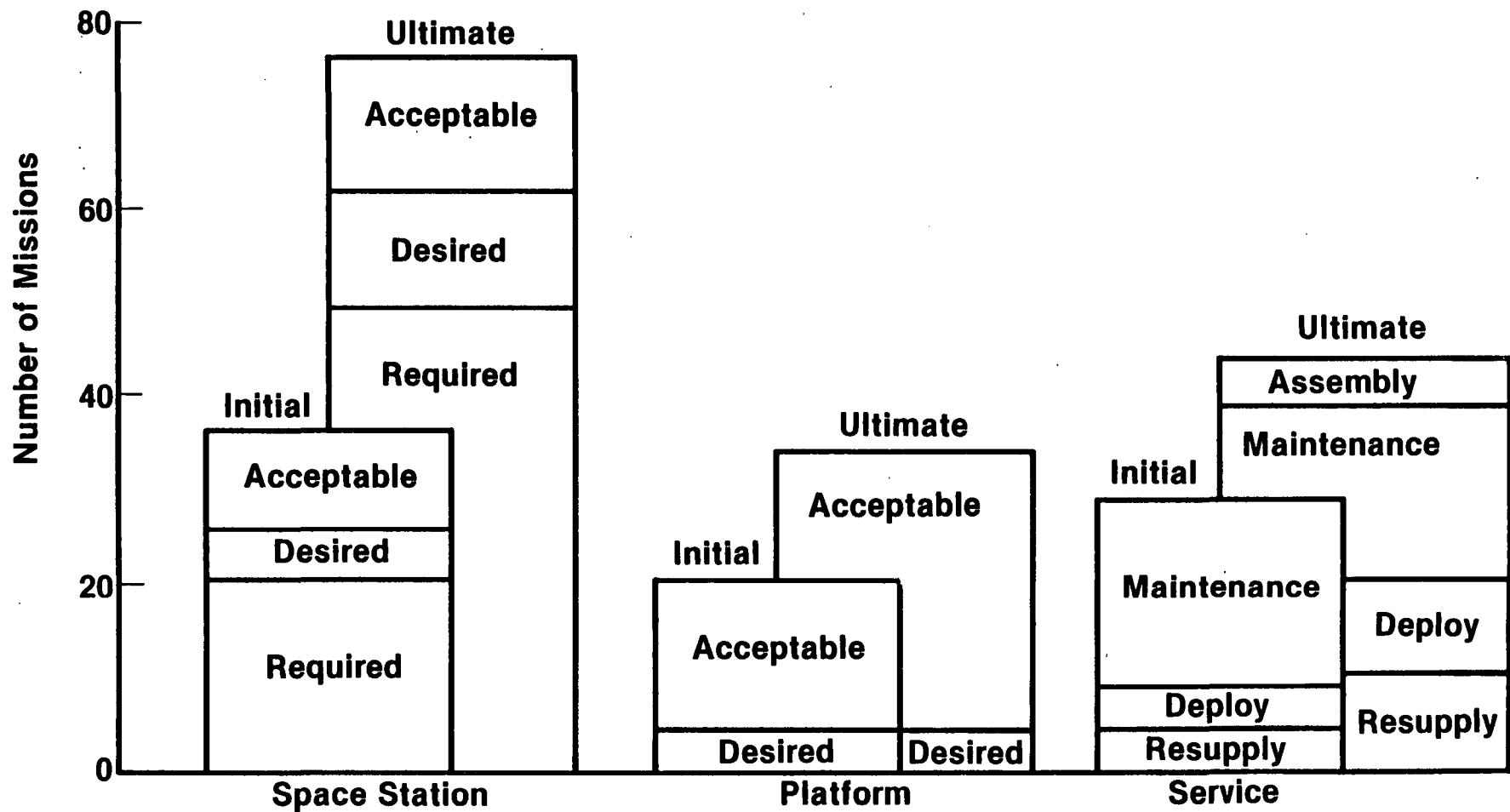
SHUTTLE PERFORMANCE



H28

SPACE STATION SYSTEM ALLOCATION REQUIREMENTS

VFY000

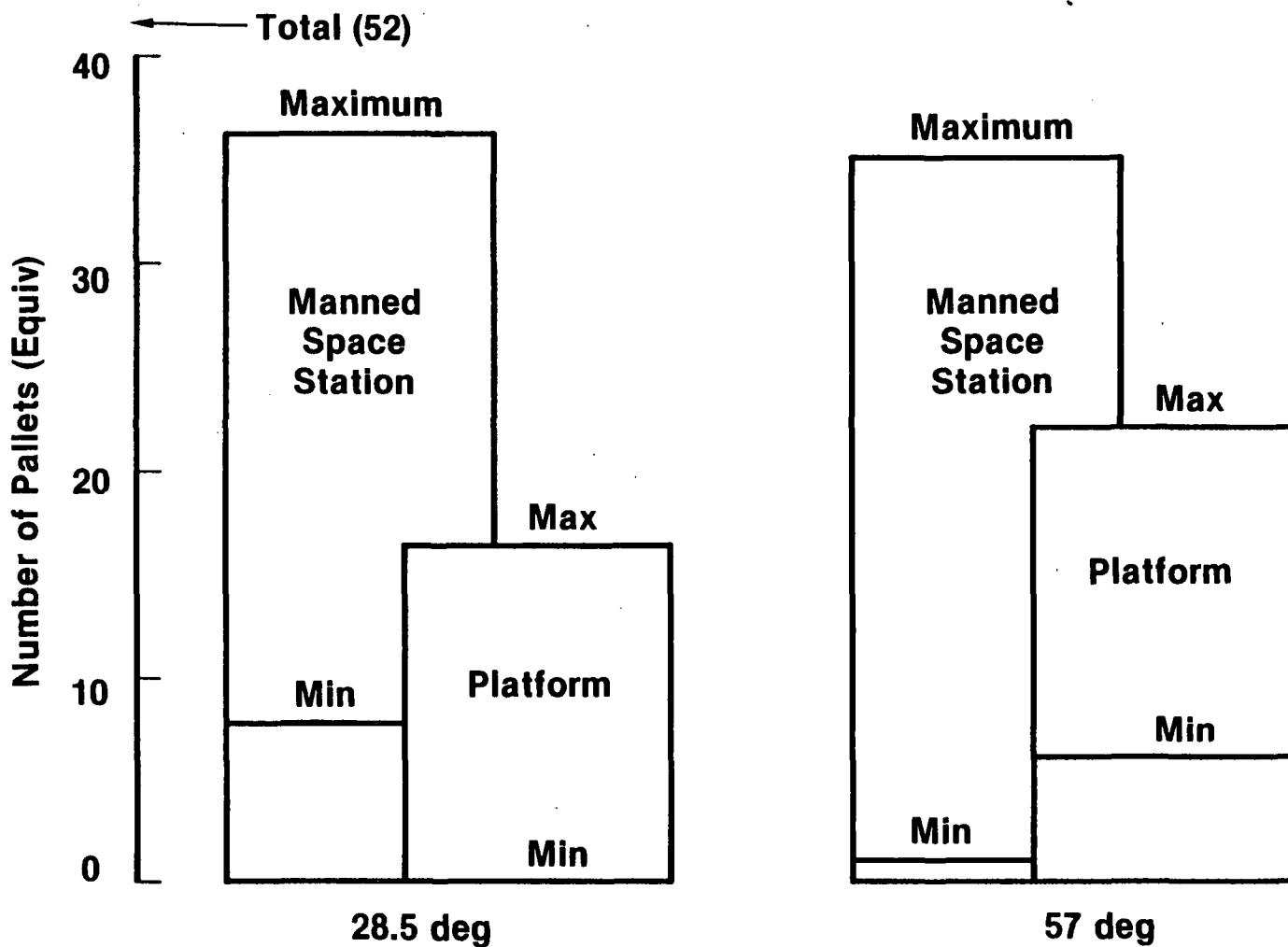


H29

EXTERNAL VOLUME ALLOCATION REQUIREMENTS

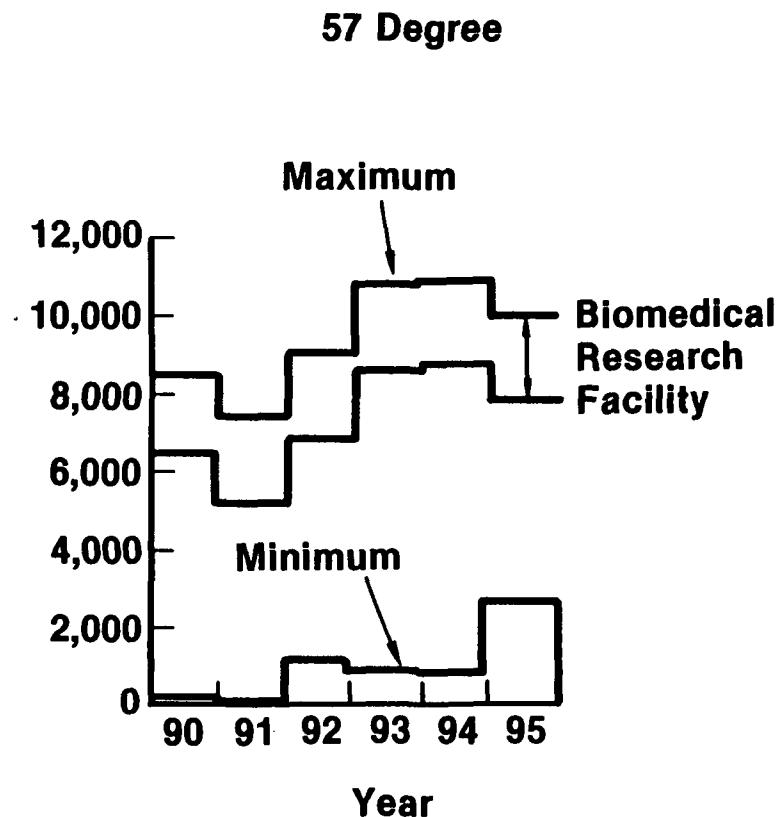
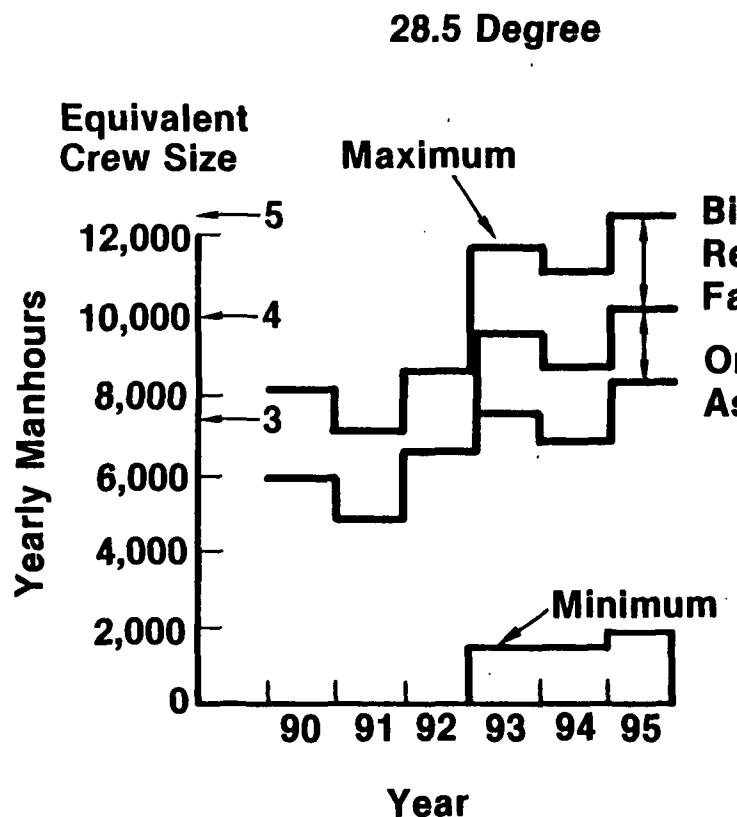
3M Pallets (Equivalent)

VFY147



H30

MISSION MANHOUR REQUIREMENTS

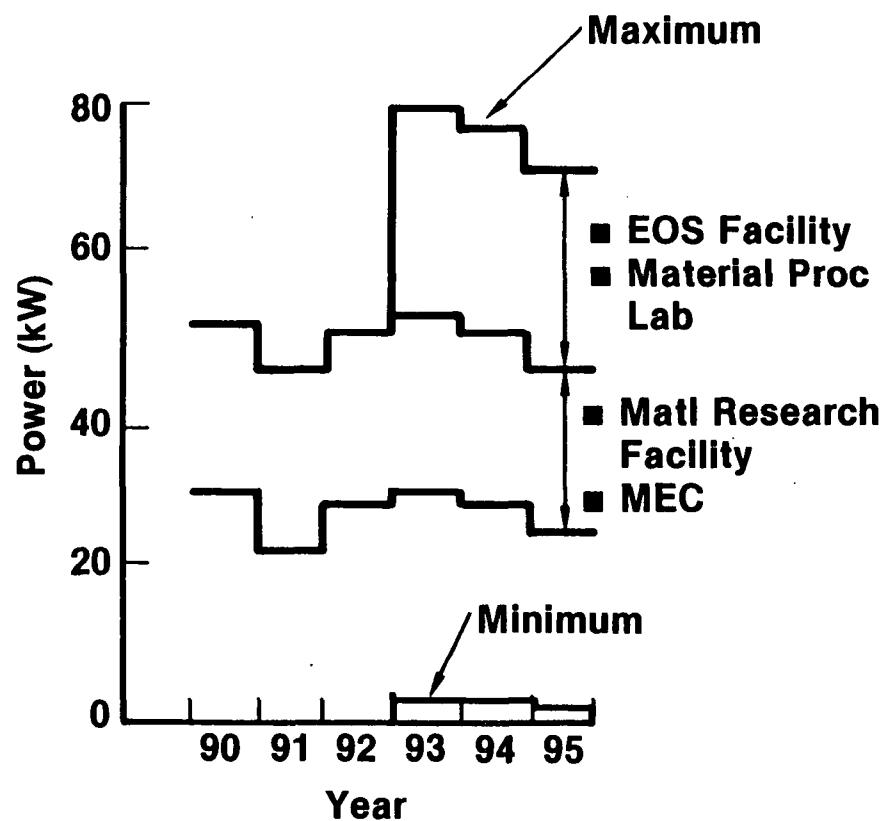


H31

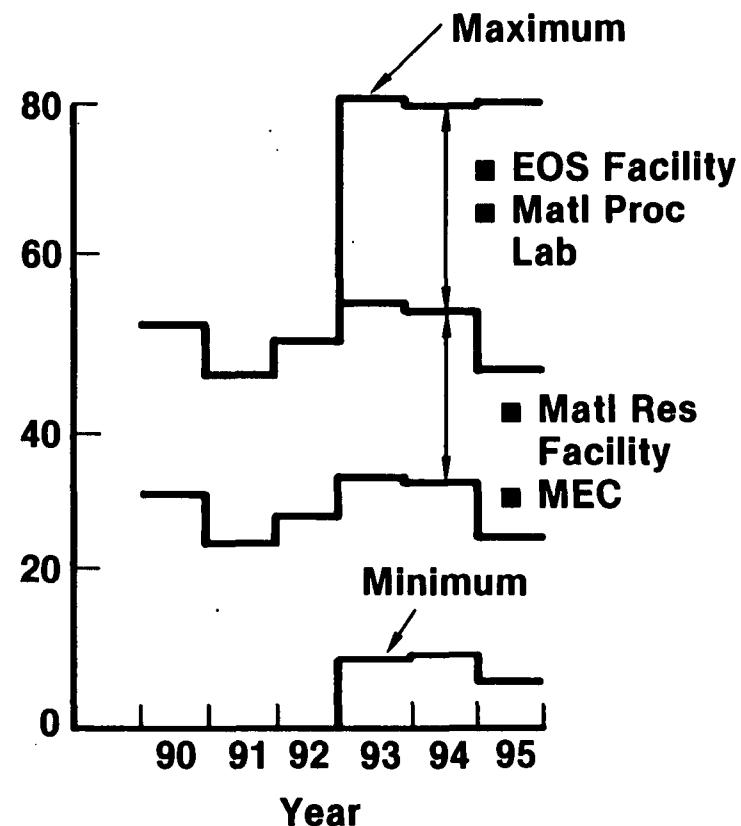
MISSION POWER REQUIREMENTS SPACE STATION INCLINATION ALLOCATION

VFY023

28.5 Degree



57 Degree

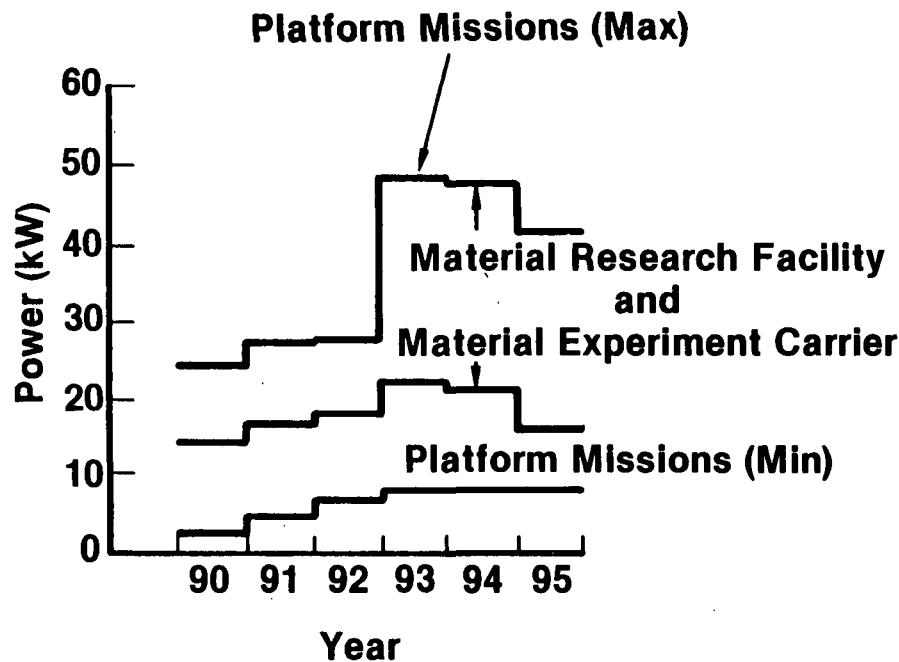


H32

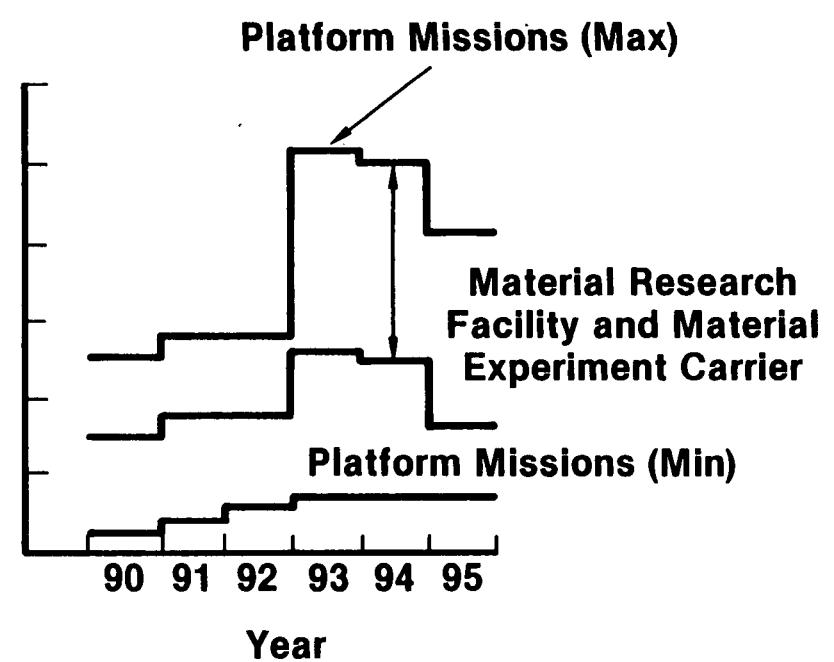
MISSION POWER REQUIREMENTS PLATFORM ALLOCATION

VFY083

28.5 Degrees



57 Degrees



H33

MISSION REQUIREMENTS ANALYSIS

MIDTERM CONCLUSIONS

VFY239

- **Manned Space Stations Required at 28.5 and 57 Degrees**
Combined Requirements:
 - Pressurized Modules⁽¹⁾: 8
 - External Pallets⁽¹⁾: Up to 45
 - Crew Size: 4-6
 - Power: 40 to 90 kW
- **Unmanned Platforms Desired at 28.5 and 57 Degrees**
Combined Requirements:
 - Pressurized Modules (Man-Tended): 2
 - External Pallets: Up to 25
 - Power: Up to 48 kW
- **STS Growth Needed:**
 - TMS: 1990
 - Propellant Depot and Reusable OTV: 1995

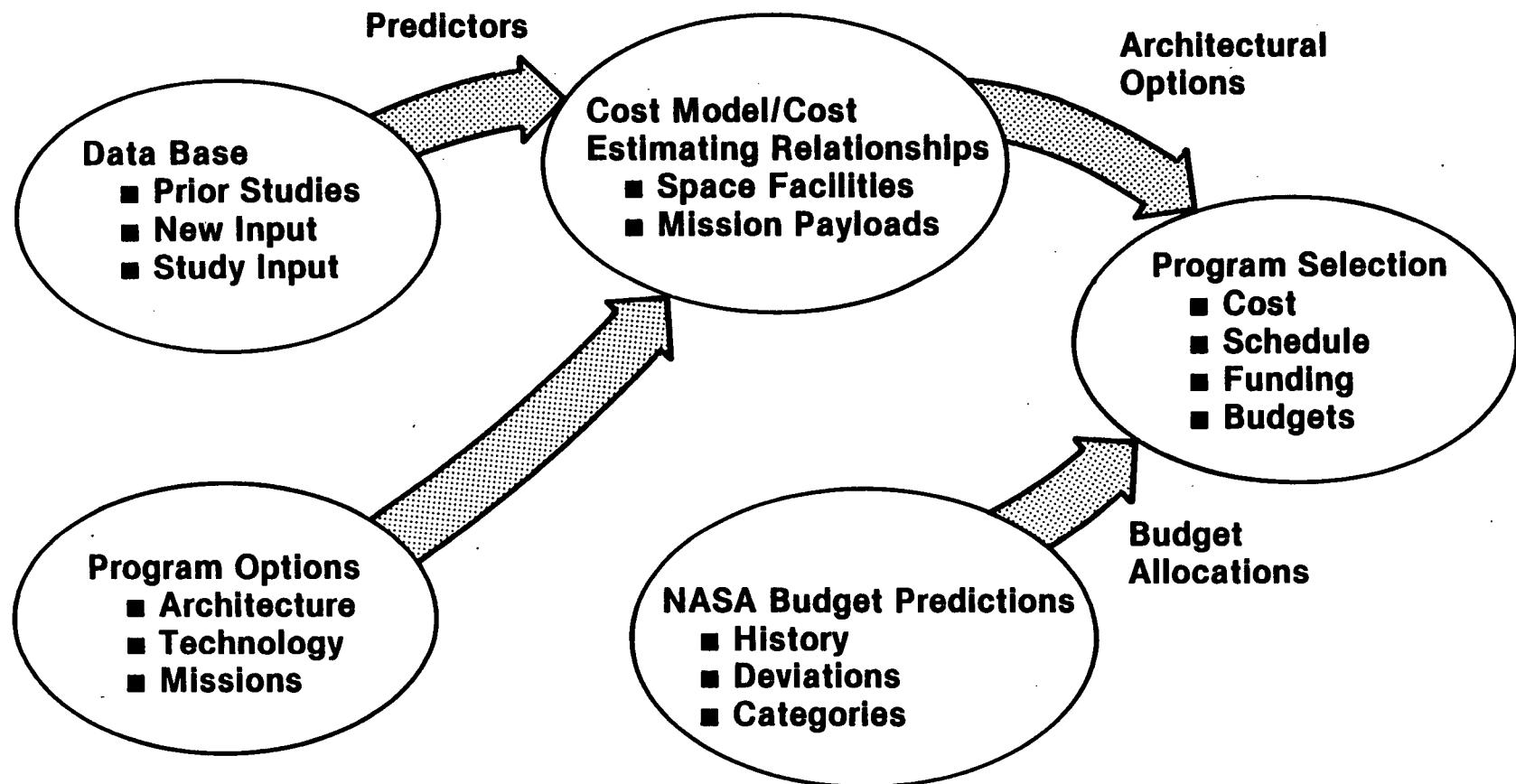
⁽¹⁾Equivalent Spacelab 3M Modules or Pallets

PROGRAMMATICS (TASK 3)

Bob Cowls

J1

MDAC APPROACH — PROGRAMMATICS

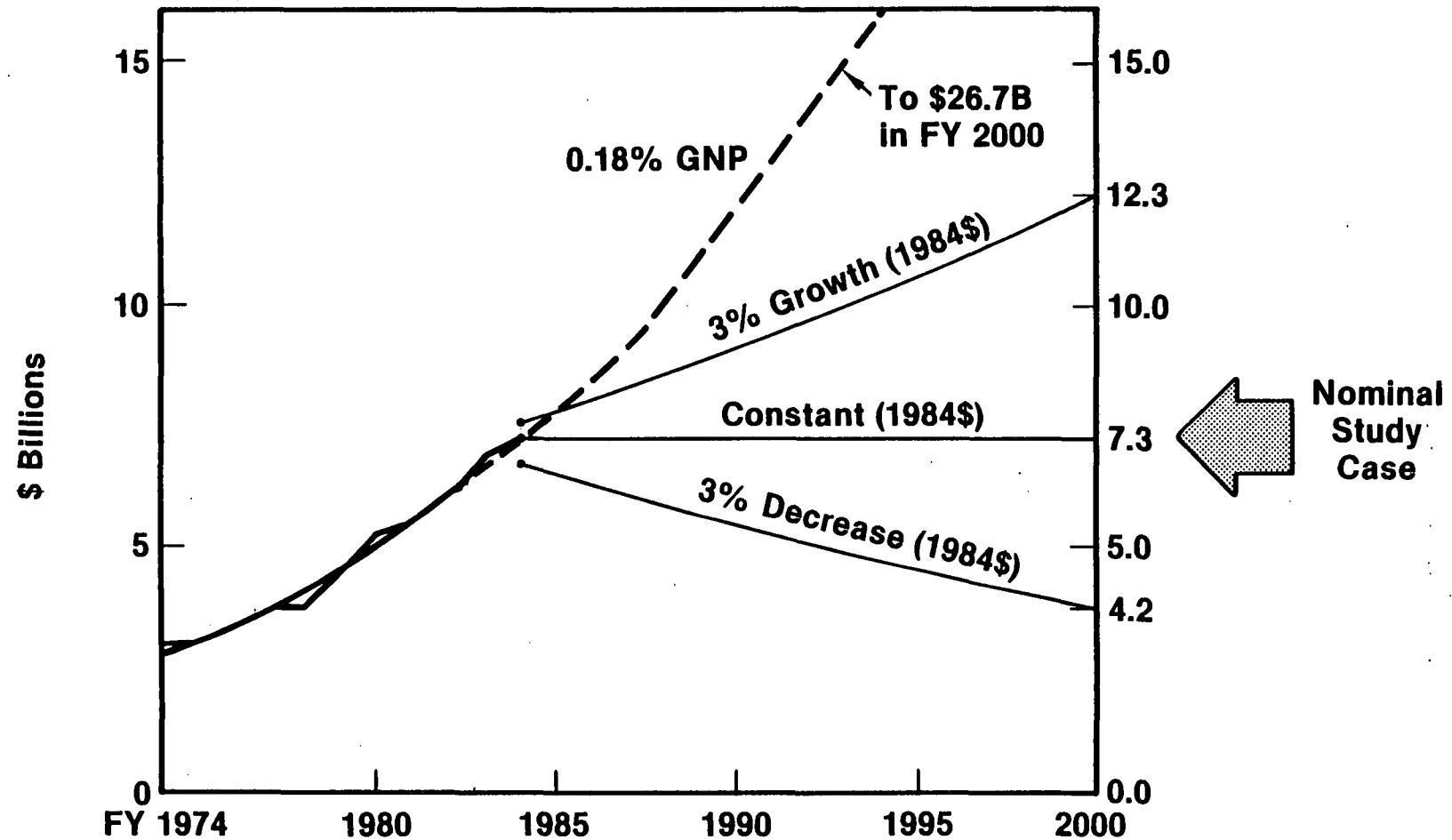


J2

NASA FUNDING PROFILE ASSUMPTIONS

- Continued Correspondence to GNP Growth (0.18% GNP)
- Need Balance Between Orbital Facility and Mission Expenditures (Recognizes Shortfall in Non-STS Areas Over Recent Years)
- No External Funds for Orbital Facilities Development
- Diminishing STS Requirement Produces Budget Wedge Opening

NASA BUDGET FORECAST



J4

SPACE STATION PROGRAM BUDGET ALLOCATION ASSUMPTIONS

| FUNDING SOURCE | SPACE STATION FACILITIES AND OPERATIONS* | MISSION EQUIPMENTS AND OPERATIONS* | | | | | |
|------------------|------------------------------------------|------------------------------------|---------------------------------|----------------------|----------------------------|---------------------|-------------------------|
| | | SCIENCE AND APP'S MISSIONS | TECHNOLOGY DEVELOPMENT MISSIONS | SPACE OPS** MISSIONS | NATIONAL SECURITY MISSIONS | COMMERCIAL MISSIONS | INTER-NATIONAL MISSIONS |
| NASA | | | | | | | |
| OSTS | ● | | O | ● | | | |
| OSSA | | ● | O | | | | |
| OAST | | | ● | | | | |
| DoD | | | | ● | | | |
| COMMERCIAL | | | | | | ● | |
| FOREIGN (BARTER) | | | | | | | ● |

● PRIMARY
○ SECONDARY

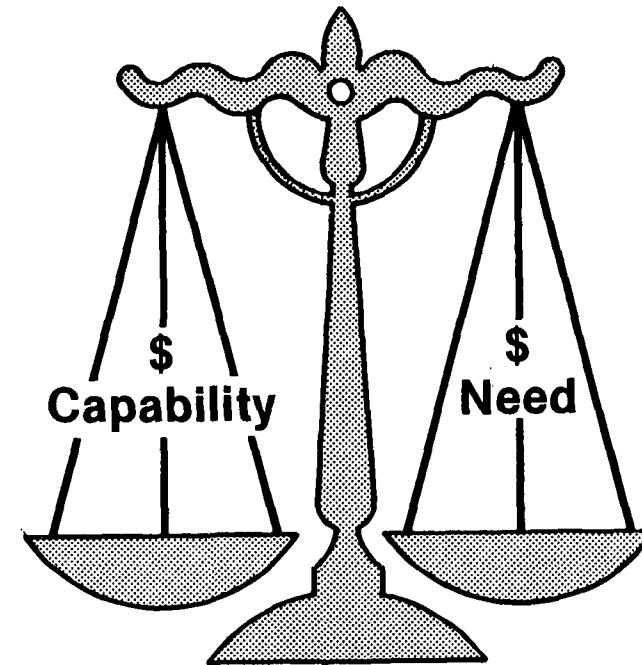
*INCLUDES HARDWARE COST (DEVELOPMENT, PRODUCTION) AND OPERATIONAL PHASE ACTIVITIES' COST

**OPERATIONAL MISSION, e.g., SPACECRAFT TRANSFER FROM LEO TO GEO, NOT MISSION OPERATIONAL PHASE ACTIVITIES, VIZ., ACTIVATION, RESUPPLY AND REFURBISHMENT

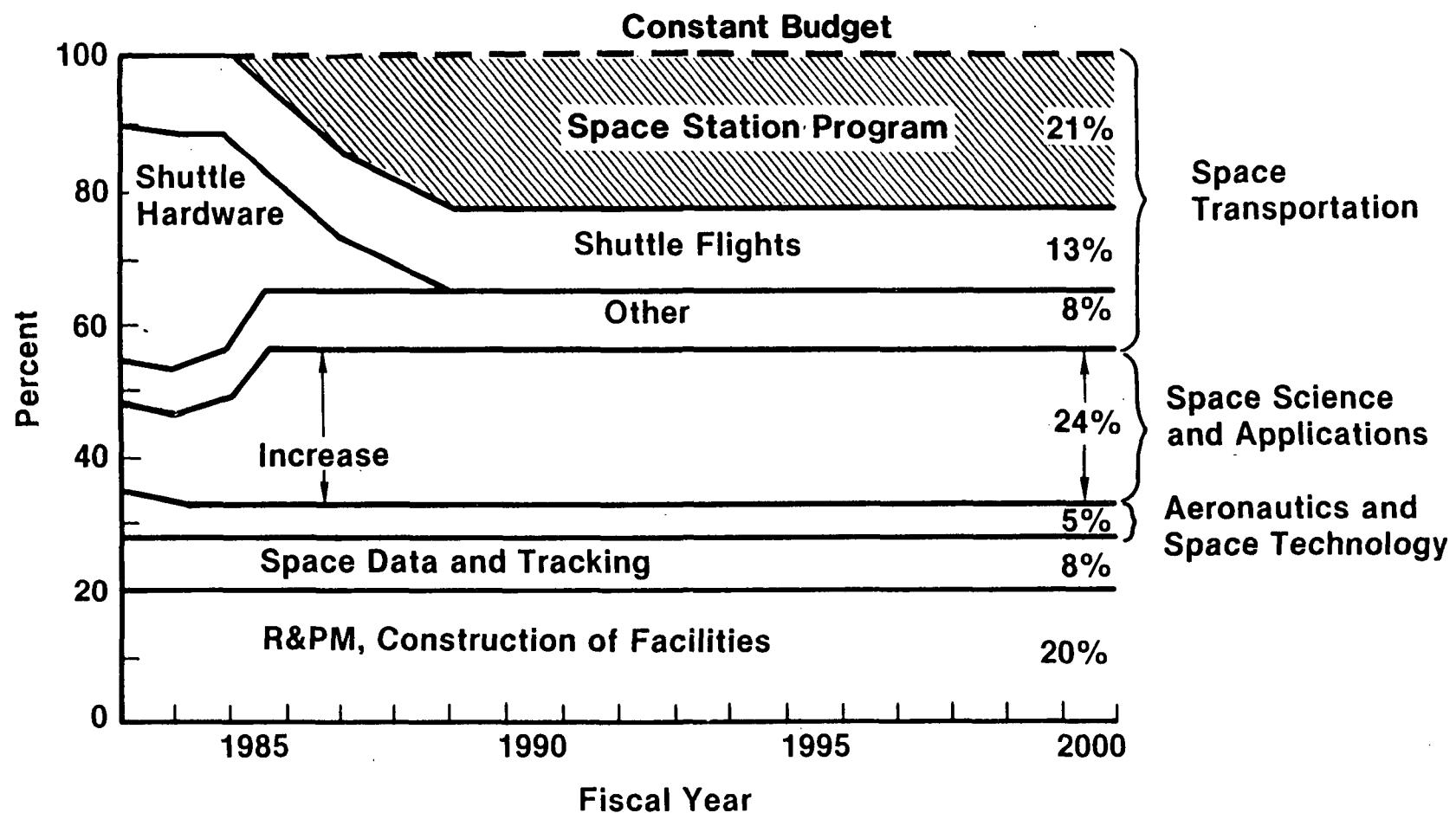
FUNDING ALLOCATION

Objective: Facility Capability = Mission Needs

**Orbital Location
Electrical Power
Crew Size
Volume
Data
Schedule
Equipment**

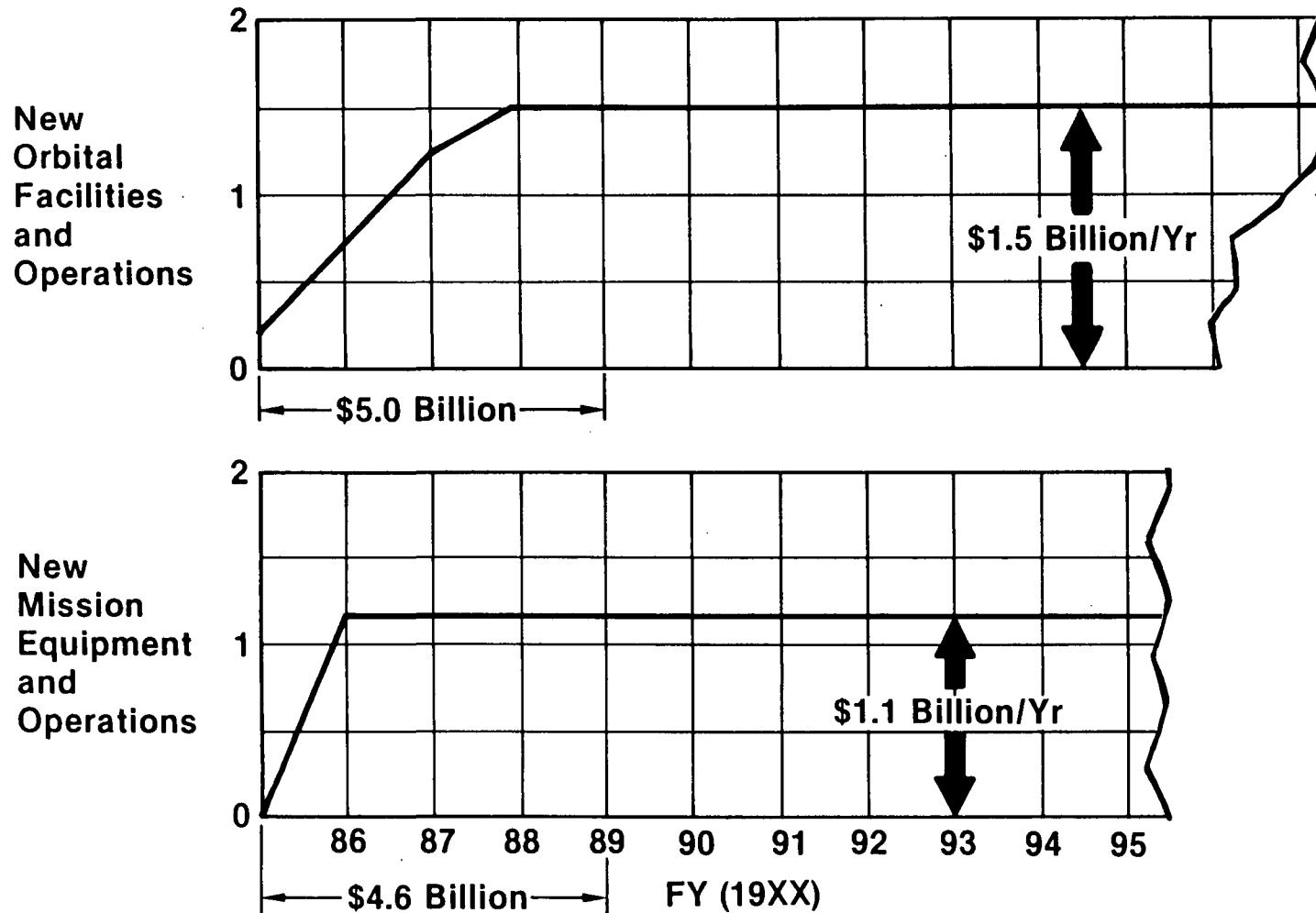


NASA BUDGET ALLOCATION ASSUMPTIONS



BUDGET MODEL NOMINAL CASE

(Billion Dollars, 1984)



Notes: (1) Science and Applications Budget Increased 60% Above 1983
 (2) Shuttle Flights Budgeted at \$0.9 Billion/Yr, Are Excluded
 (3) All NASA Funds; No Commercial, DoD or Foreign Funds

PROGRAMMATICS SUMMARY

- Funds for Space Station Program Are Available Due to Diminishing Shuttle Hardware Costs
- Funds for Fifth Orbiter Available
- Balance Between Mission Needs and Facility Capability Is Necessary

VFY040

MISSION IMPLEMENTATION (TASK 2)

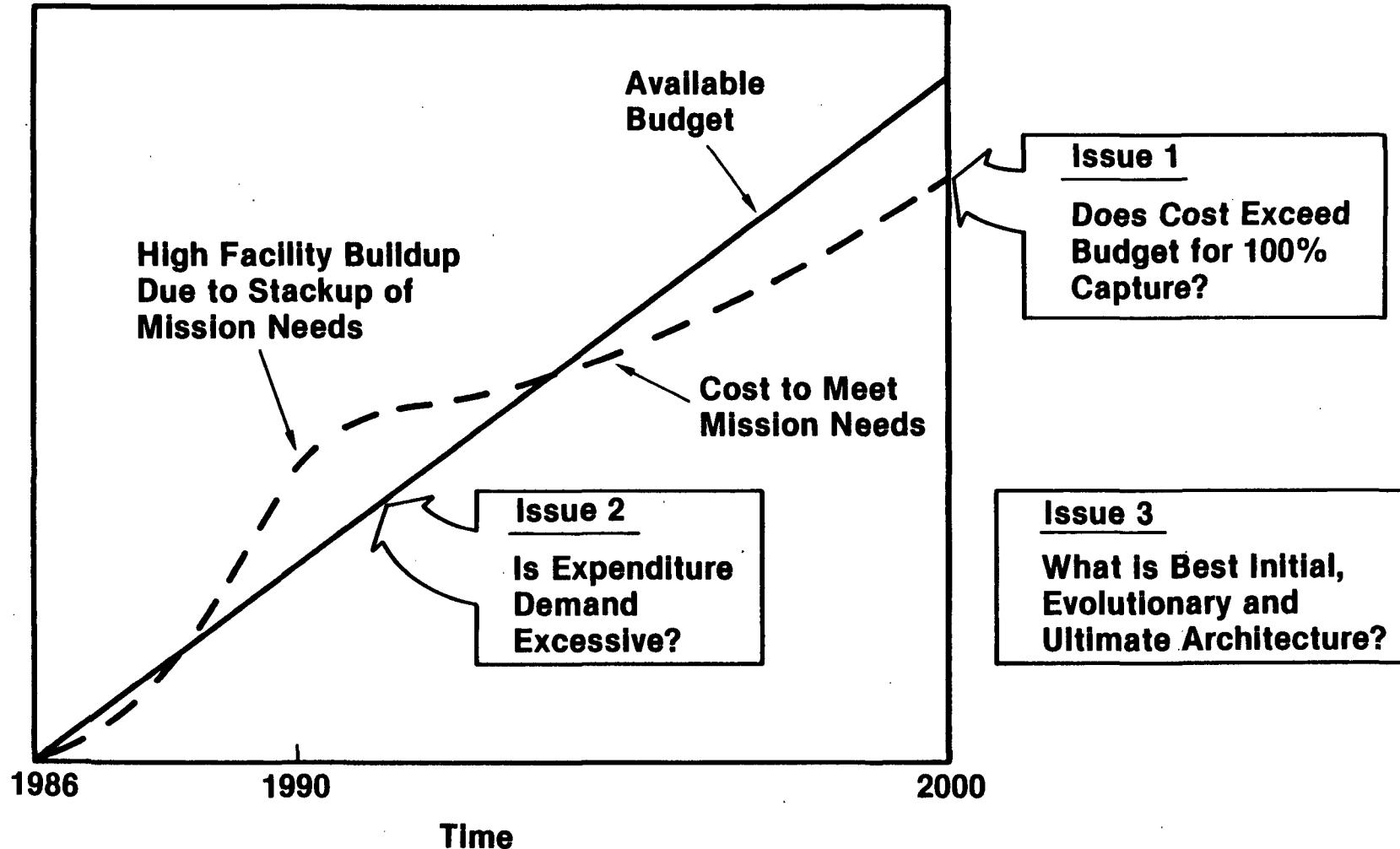
Bill Nelson

TASK 2

MISSION IMPLEMENTATION CONCEPTS

VFY027

Cummulative
Budget and Cost (\$)



K1

ARCHITECTURAL GOALS

- Maximize Mission Capture
 - Total Number
 - Number of Categories
- Provide Flexible Accommodations
 - Orbit Locations
 - Facility Types
- Maximize Cost Effectiveness
 - Cluster Facilities Preferred
 - Locate at Shuttle Traffic Lanes (Comanifesting)
 - Maximize Mission Capture per Unit Cost

K2

MISSION LOCATION AND FACILITY TYPE REQUIREMENTS

VFY008

| Inclination (deg) | Facility Type | | | | |
|----------------------|---------------|--------|--------|--------|----------|
| | Manned | Either | Either | Either | Unmanned |
| 28 | 17 | | 4 | 4 | 1 |
| 28-57 | 4 | | 2 | 2 | 4 |
| 28-90 | 4 | 0 | | 0 | 1 |
| 28-98 | 17 | 5 | | 5 | 1 |
| 57 | 4 | 3 | 3 | | 0 |
| 57-90 | 1 | 7 | | 7 | 0 |
| 57-98 | 2 | 3 | | 3 | 1 |
| 90 | 0 | 1 | | 1 | 0 |
| 90-98 | 1 | 0 | | 0 | 1 |
| 98 | 3 | 1 | | 1 | 0 |

**Number of Missions Requiring
Manned Facility at 57 Degree Inclination**

K3

MISSION CAPTURE FOR SPACE STATION AT 57° INCLINATION

VFY009

| Inclination (deg) | Facility Type | | | | |
|----------------------|---------------|--------|--------|--------|----------|
| | Manned | Either | Either | Either | Unmanned |
| 28 | 17 | | 4 | 4 | 1 |
| 28-57 | 4 | 2 | | 2 | 4 |
| 28-90 | 4 | 0 | | 0 | 1 |
| 28-98 | 17 | 5 | | 5 | 1 |
| 57 | 4 | 3 | | 3 | 0 |
| 57-90 | 1 | 7 | | 7 | 0 |
| 57-98 | 2 | 3 | | 3 | 1 |
| 90 | 0 | 1 | | 1 | 0 |
| 90-98 | 1 | 0 | | 0 | 1 |
| 98 | 3 | 1 | | 1 | 0 |

Mission Capture for
Space Station at 57 Degree
Inclination

K4

FACILITY NUMBER AND LOCATION FOR 100% MISSION CAPTURE

| Inclination (deg) | Facility Type | | | | |
|----------------------|---------------|--------|--------|--------|----------|
| | Manned | Either | Either | Either | Unmanned |
| 28 | 17 | | 4 | | 1 |
| 28-57 | 4 | | 2 | | 4 |
| 28-90 | 4 | 0 | | | |
| 28-98 | 17 | 5 | | | |
| 57 | 4 | 3 | | | 0 |
| 57-90 | 1 | 7 | | | 0 |
| 57-98 | 2 | 3 | | | |
| 90 | 0 | 1 | 57° | | 0 |
| 90-98 | 1 | 0 | 98° | | 1 |
| 98 | 3 | 1 | | | 0 |

Space Station Mission Capture Platform Mission Capture

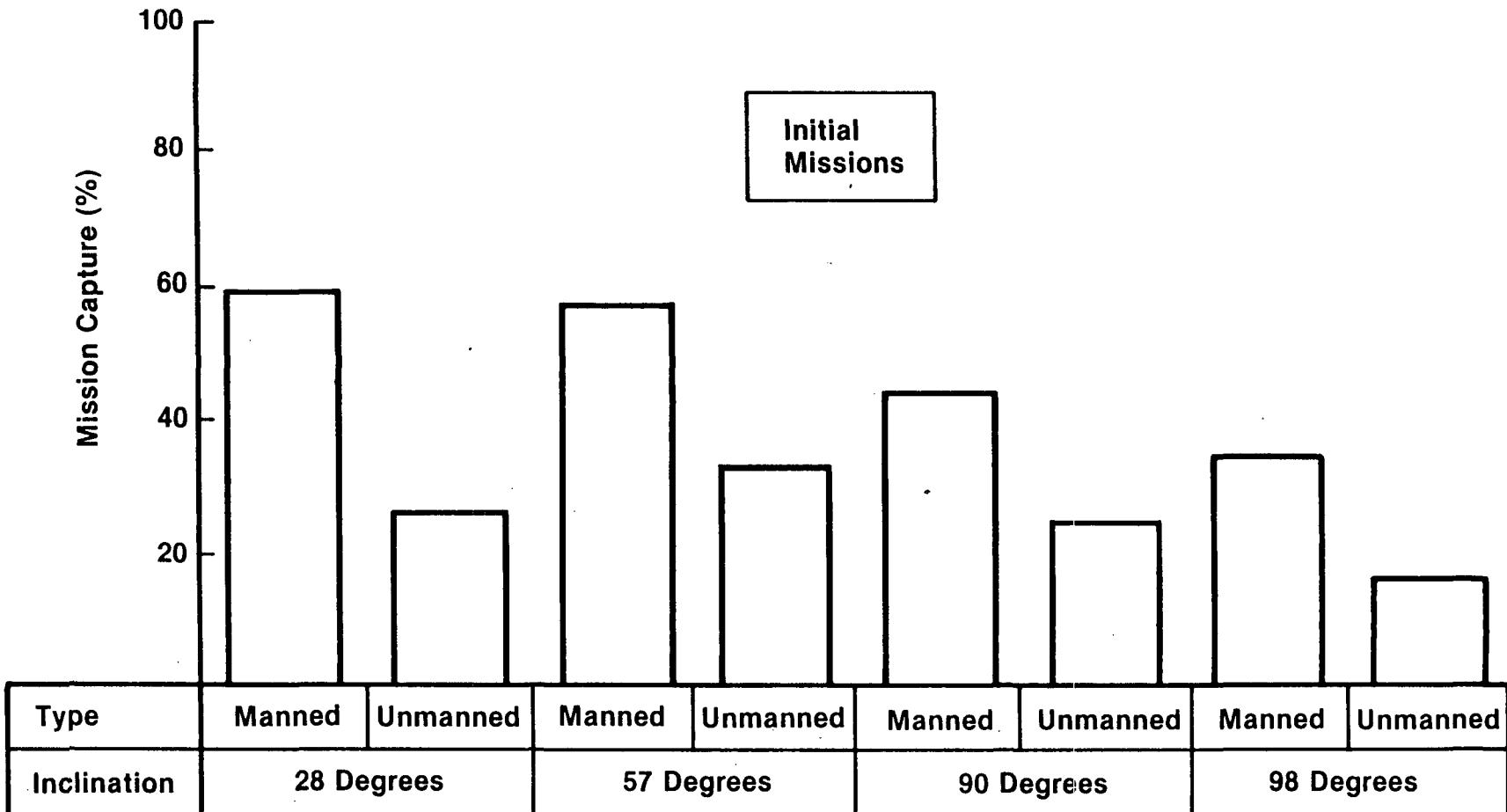
Mission Allocation for Capture

K5

SINGLE FACILITY MISSION CAPTURE

(Shows Maximum Mission % Capture
by Any Single Facility)

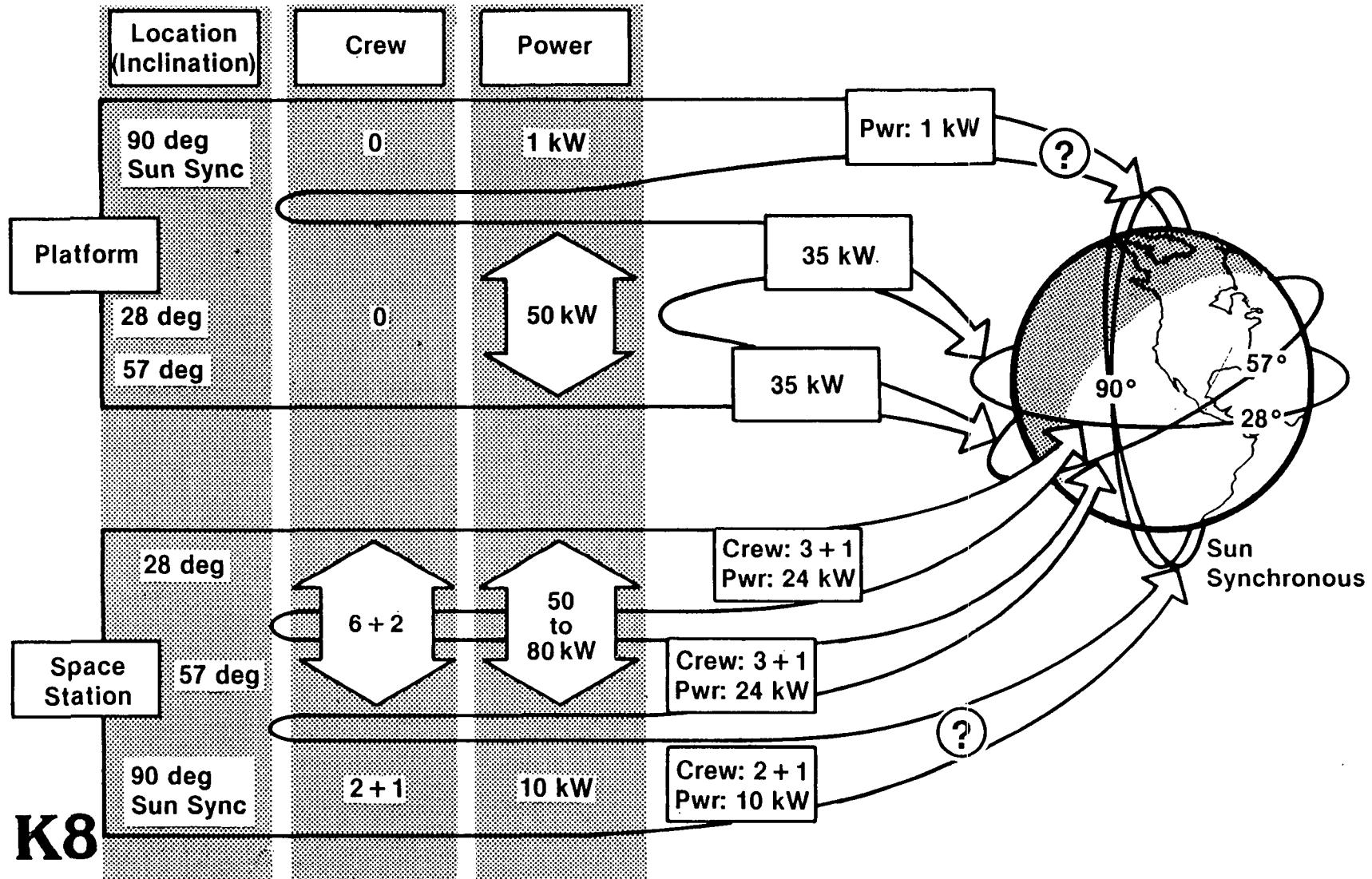
VFY167



K6

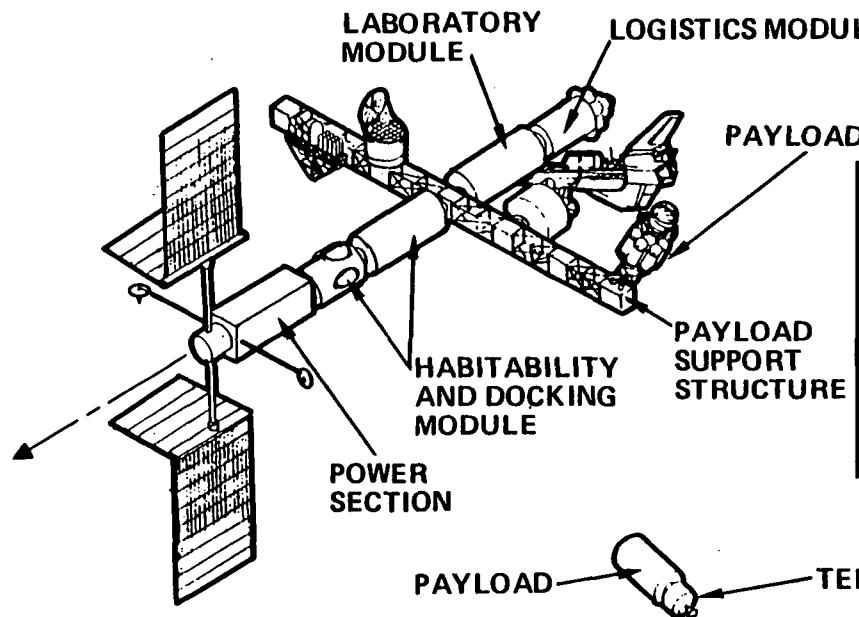
ALLOCATION OF FACILITY REQUIREMENTS ULTIMATE CAPABILITY

VFY135



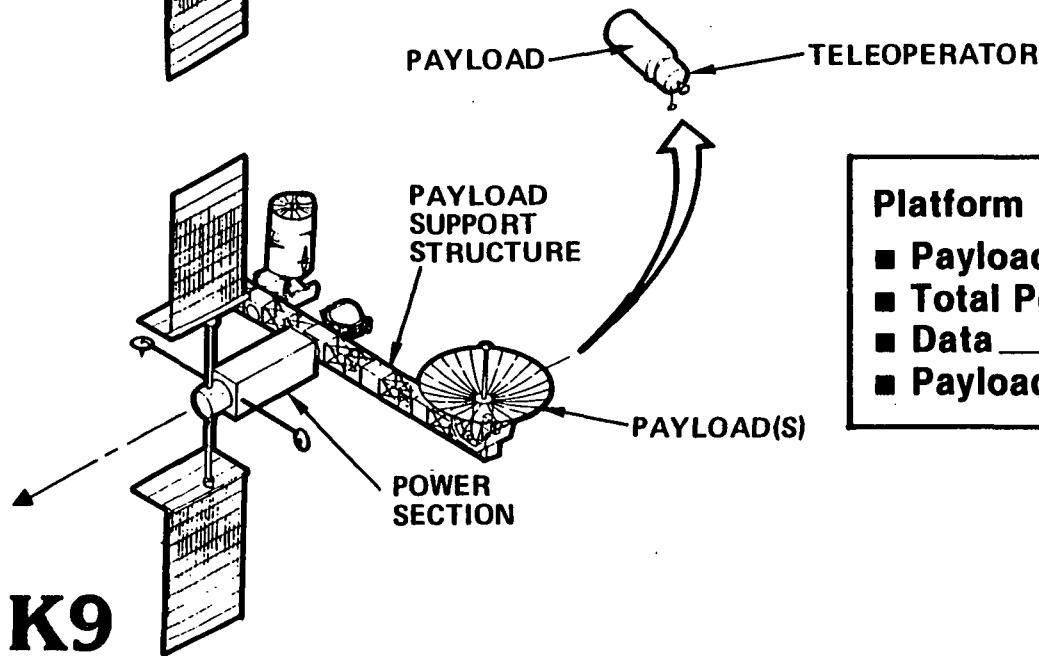
INITIAL CAPABILITY SPACE STATION FACILITY

VFY065



Manned Space Station

| | |
|----------------------|----------|
| ■ Crew | 4 |
| ■ Payload Power | 24 kW |
| ■ Total Power | 37 kW |
| ■ Data | 120 Mbps |
| ■ Payload Complement | 10 |

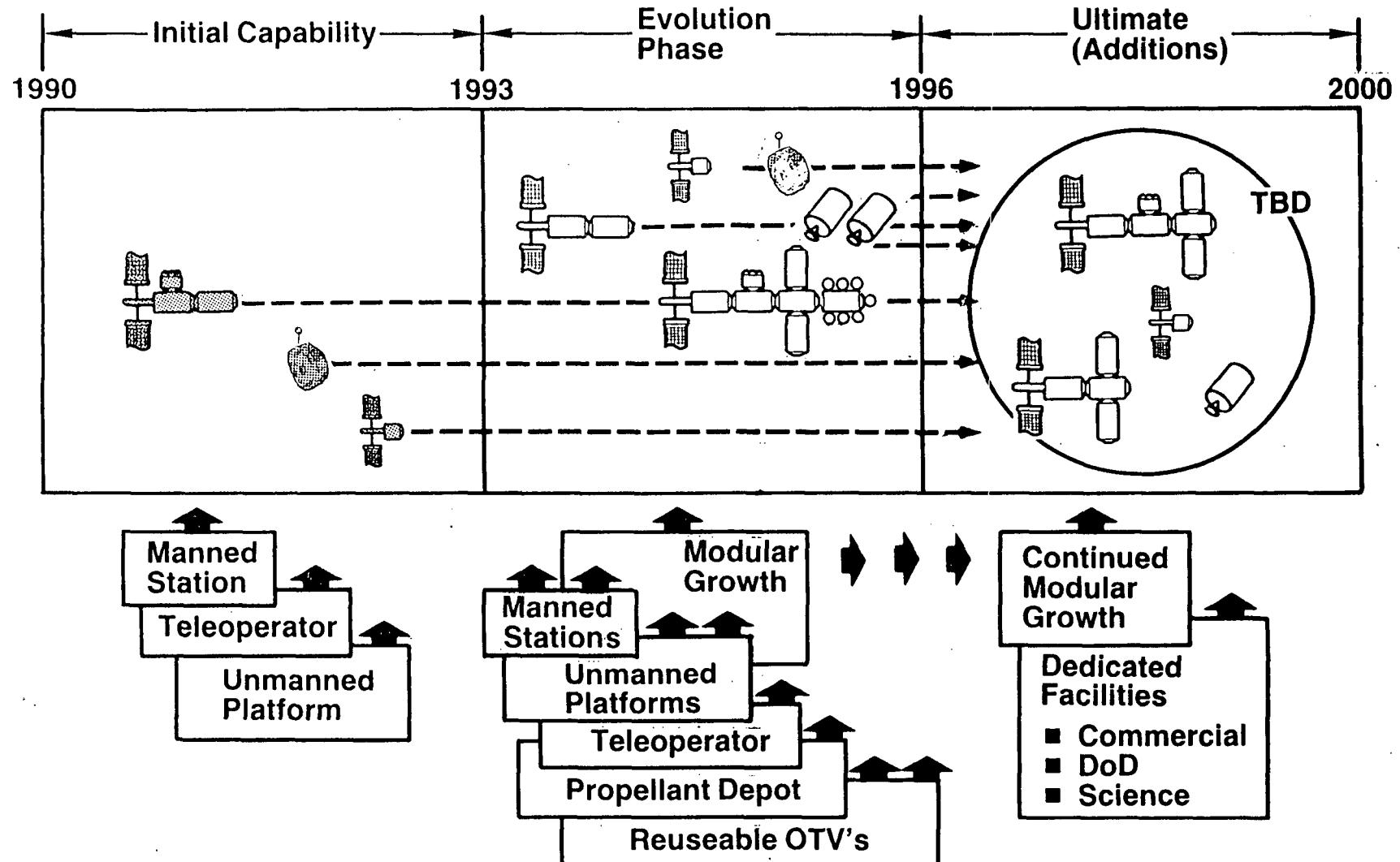


Platform

| | |
|----------------------|----------|
| ■ Payload Power | 35 kW |
| ■ Total Power | 38 kW |
| ■ Data | 120 Mbps |
| ■ Payload Complement | 10 |

K9

CAPABILITY GROWTH OPTIONS

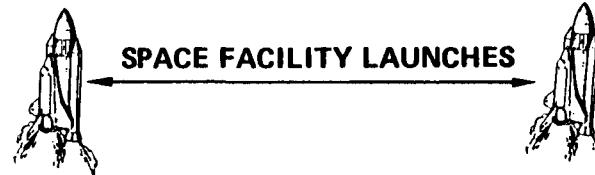
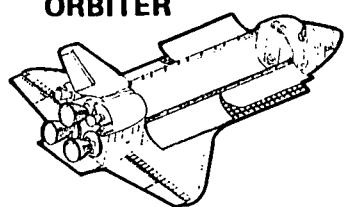


STS ELEMENTS FOR SPACE STATION PROGRAM

VFY064

| 1990 | 1991 | 1992 | 1993 | 1994 | 1995 | 1996 | 1997 | 1998 | 1999 |
|------|------|------|------|------|------|------|------|------|------|
|------|------|------|------|------|------|------|------|------|------|

ORBITER



LOGISTICS/PAYOUT

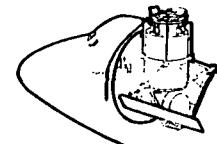
ORBITER RMS



SPACE FACILITY ASSEMBLY

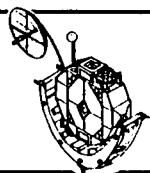
BERTHING OPERATIONS

DOCKING MODULE



BERTHING OPERATIONS (CREW AND LOGISTICS TRANSFER)

TMS



SPACE FACILITY ASSEMBLY/PLACEMENT

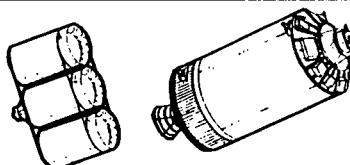
PAYOUT DEPLOYMENT/RETRIEVAL

MMU



PAYOUT SERVICING

REUSABLE OTB



PAYOUT DEPLOYMENT/RETRIEVAL

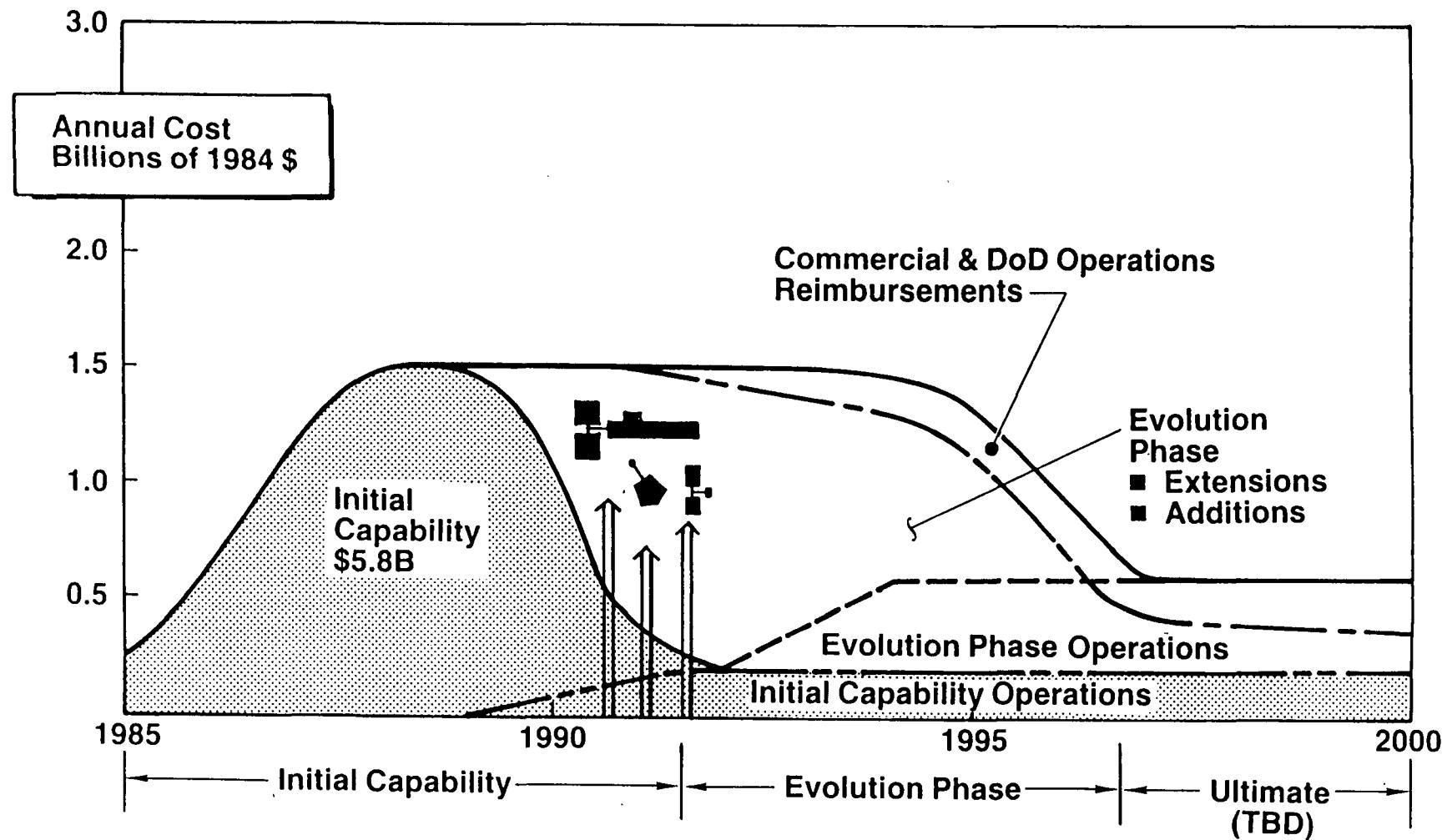
DEPOT PROVISIONS

K11

SAMPLE PROGRAM COSTS

100% MISSION CAPTURE

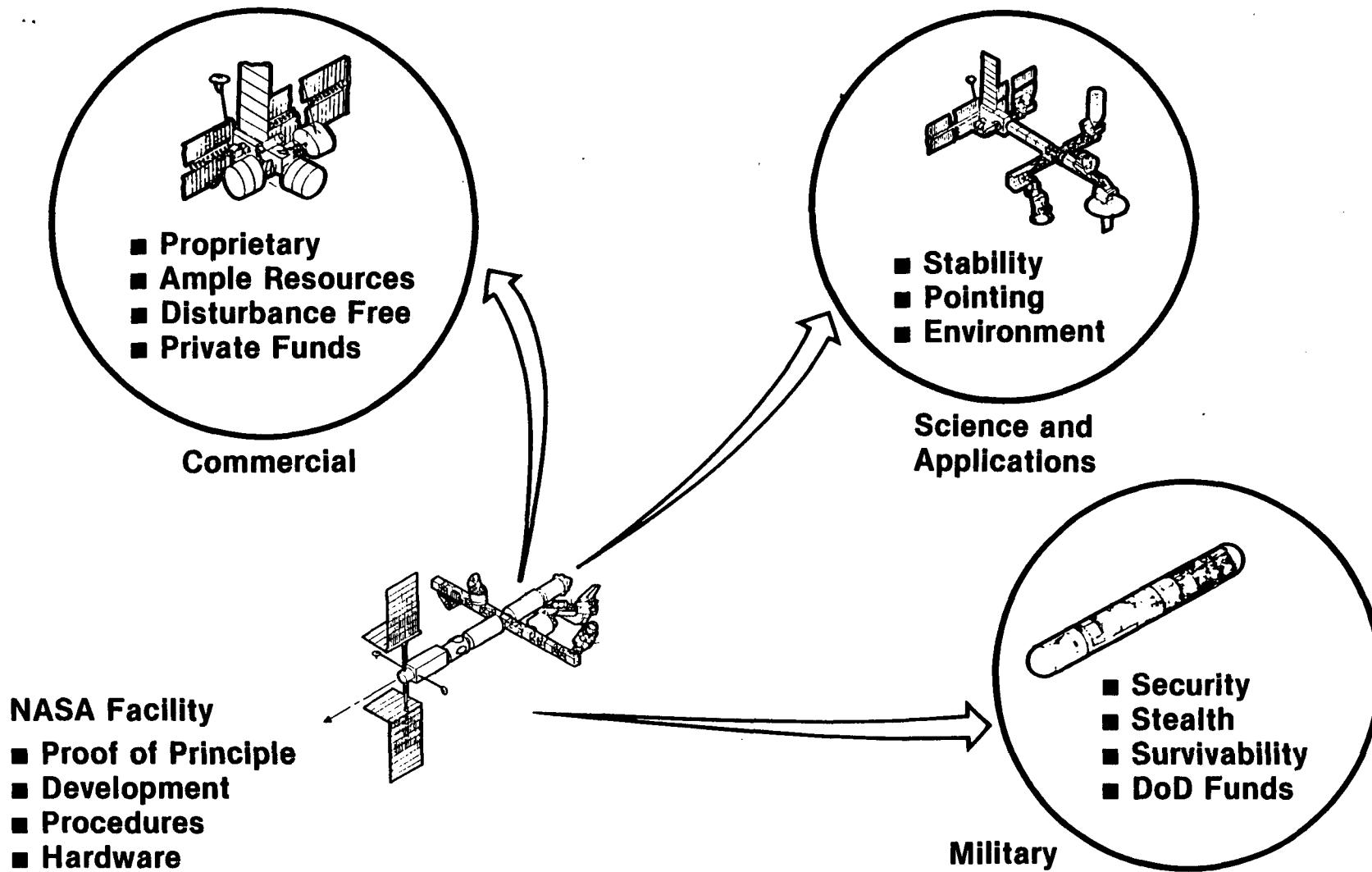
VFY269



K11

SPACE STATION SYSTEM GROWTH BASED ON SPECIALTY NEEDS

VFY066



K13

CONCLUSIONS

MISSION IMPLEMENTATION CONCEPTS

- 100 Percent Mission Capture Possible
 - Within Total Budget Limits
 - Space Stations/Platforms/Transportation
- Buildup Constraints
 - Rate of Budget Availability
 - Production Rates
- Initial Capability in 1990 - 1991

K13

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